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A COMPUTER CODE FOR THREE-DIMENSIONAL INCOMPRESSIBLE
FLOWS USING NONORTHOGONAL BODY-FITTED COORDINATE SYSTEMS

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16. ABSTRACT In this report, a numerical method for solving the equations of motion of three-dimensional incompressible flows in nonorthogonal body-fitted coordinate (BFC) systems has been developed. The equations of motion are transformed to a generalized curvilinear coordinate system from which the transformed equations are discretized using finite difference approximations in the transformed domain. The hybrid scheme is used to approximate the convection terms in the governing equations. Solutions of the finite difference equations are obtained iteratively by using a pressure-velocity correction algorithm (SIMPLE-C). Numerical examples of two- and three-dimensional, laminar and turbulent flow problems are employed to evaluate the accuracy and efficiency of the present computer code. The user's guide and computer program listing of the present code are also included.			
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NOMENCLATURE

<u>Symbol</u>	<u>Definition</u>
A	link coefficient of finite difference equation
A_p	link coefficient of the node at the center of a control volume
A^u	link coefficient for the u-equation
A^v	link coefficient for the v-equation
A^w	link coefficient for the w-equation
A^o	link coefficient for time marching scheme
C_v	specific heat constant at constant volume
C_1	turbulence model constant, = 1.44
C_2	turbulence model constant, = 1.92
C_μ	turbulence model constant, = 0.09
D	diffusion coefficient for the pressure correction equation
e	internal energy per unit mass (jour/kg)
J	Jacobian of the metric transformation
k	turbulence kinetic energy (m^2/s^2)
k'	thermal conductivity of the fluid
P	pressure in the fluid (N/m^2)
P_r	production term for the turbulent kinetic energy
Q	energy added per unit volume (jour/ m^3)
S	source term
S_u	source term of the u-equation
S_v	source term of the v-equation
S_w	source term of the w-equation
T	temperature ($^{\circ}K$)
t	time (sec)
u	velocity in x direction

v	velocity in y direction
w	velocity in z direction
X	X-coordinate (m)
Y	Y-coordinate (m)
Z	Z-coordinate (m)

Greek

Γ	diffusion coefficient
ϵ	turbulent kinetic energy dissipation rate (m^2/s^3)
Δ	difference operator
ϕ	variable of general transport equation
ϕ^0	solution at the previous time level
σ_k	turbulence model constant, = 1.0
σ_ϵ	turbulence model constant, = 1.3
ξ	curvilinear coordinate
ζ	curvilinear coordinate
μ	molecular viscosity ($\text{N}\cdot\text{s}/\text{m}^2$)
μ_t	turbulent eddy viscosity ($\text{N}\cdot\text{s}/\text{m}^2$)
μ_{eff}	effective viscosity ($\text{N}\cdot\text{s}/\text{m}^2$)
ρ	density (kg/m^3)
Σ	summation over all values around a grid node P
i	
η	curvilinear coordinate

Subscript

i	index of all possible values
ref	reference value
max	maximum quantity

Superscript

- o previous time level solution
- * current solution
- ' correction quantity

CONTRACTOR REPORT

A COMPUTER CODE FOR THREE-DIMENSIONAL INCOMPRESSIBLE FLOWS USING NONORTHOGONAL BODY-FITTED COORDINATE SYSTEMS

INTRODUCTION

With the currently increasing computer capability and various flow solvers developed, numerical simulations of three-dimensional incompressible flow problems using Reynolds-average Navier-Stokes equations are now becoming more feasible in many engineering design and analysis applications. In many real world flow problems, the boundary geometries are complex such that it is more accurate to describe the geometries using body-fitted coordinate (BFC) systems. Especially for internal flow problems with complex geometries such as those of the hot gas manifold (HGM) of the Space Shuttle Main Engine (SSME), the use of nonorthogonal BFC systems for numerical solutions can be beneficial in many aspects. It is not only the boundary geometries that can be represented more closely using BFC systems, but also grid-refined solutions can be obtained without increasing an excessive amount in computer memory. In addition, once a particular flow problem has been set up, the redesign or optimization process of the boundary shapes can be performed very easily using BFC systems.

Several numerical methods [1, 2, 3, 4, 5, 6] have been developed for solving the incompressible Navier-Stokes equations in 3-D BFC systems. The main difference between these methods lies in the way of finding a pressure field such that the flowfield can be as close to divergence-free as possible (i.e. to satisfy the mass conservation equation). This is the main feature and difficulty of solving the incompressible flow problems. Numerical methods of References 1, 2 and 3, for instance, have employed the pseudocompressibility approach and time-iterative scheme to generate the pressure field so that the continuity equation is satisfied when a steady state solution is reached. In these methods, artificial smoothing techniques must be used to obtain a strong coupling between the velocity and pressure fields. Methods of References 4, 5 and 6, on the other hand, have utilized a successive pressure-velocity correction scheme by using a Poisson's equation for pressure correction derived approximately from the continuity and momentum equations. For these latter methods, grid staggering between the velocity vectors and the pressure nodes must be used to ensure stability of the numerical solutions.

There are several possible methods of grid staggering associated with different features in solving the pressure correction equation. These grid staggering methods were discussed in Reference 6, from which one of the methods was shown to be the most promising arrangement (i.e. with the velocity vectors located at the faces of a volume which contains the pressure and other scalars at its center). But, this method has one drawback, that the velocity components are solved using different control volumes. It is for this reason that a grid staggering system similar to the one used by Vanka et al. [4] is developed in the present study. The present method of grid staggering and pressure correction equation that was described by Vanka [4] and Maliska [6]. Also, using the present method, the same control volume is used for the velocity components and scalar quantities.

In the following sections, basic elements for establishing the present computer code for solving the curvilinear Navier-Stokes equations in three-dimensional space (CNS3D) will be described. These are followed by a series of standard numerical examples used to evaluate the accuracy and efficiency of the present numerical method. The numerical examples include laminar flow driven-cavity problem, cases of laminar and turbulent flows over backward-facing steps, and 3-D laminar flows inside a 90-deg-bend square duct. Applications of the present code to the internal flow problems of SSME will be included in future publications.

A user's guide to the present CNS3D code is provided in Appendix A. Appendix B contains a list and definitions of all the major fortran symbols used in the computer program which is listed in Appendix C.

TRANSFORMATION OF THE EQUATIONS OF MOTION

For incompressible Newtonian fluid, the continuity, momentum and energy equations can be written as:

$$U_t + E_x + F_y + G_z = S \quad (1)$$

where (x, y, z) represent the Cartesian coordinates, and

$$U = \begin{Bmatrix} \rho \\ \rho u \\ \rho v \\ \rho w \\ \rho e - Q \end{Bmatrix} \quad E = \begin{Bmatrix} \rho u \\ \rho uu - \mu u_x \\ \rho uv - \mu v_x \\ \rho uw - \mu w_x \\ \rho ue - k' T_x \end{Bmatrix}$$

$$F = \begin{Bmatrix} \rho v \\ \rho vu - \mu u_y \\ \rho vu - \mu v_y \\ \rho vw - \mu w_y \\ \rho ve - k' T_y \end{Bmatrix} \quad G = \begin{Bmatrix} \rho w \\ \rho wu - \mu u_z \\ \rho wv - \mu v_z \\ \rho ww - \mu w_z \\ \rho we - k' T_z \end{Bmatrix}$$

$$S = \left\{ \begin{array}{l} 0 \\ (\mu u_x)_x + (\mu v_x)_y + (\mu w_x)_z - p_x \\ (\mu u_y)_x + (\mu v_y)_y + (\mu w_y)_z - p_y \\ (\mu u_z)_x + (\mu v_z)_y + (\mu w_z)_z - p_z \\ \mu [2(u_x^2 + v_y^2 + w_z^2) + (v_x + u_y)^2 + (w_y + v_z)^2 + (u_z + w_x)^2 \\ \quad - \frac{2}{3} (u_x + v_y + w_z)^2] \end{array} \right\}$$

e = the internal energy per unit mass = $C_v T$ for perfect gas

Q = energy added per unit volume

k' = thermal conductivity of the fluid .

Equation (1) is transformed to a general curvilinear coordinate system (ξ, η, ζ) , which results in equation (2).

$$\begin{aligned} U_t + E_\xi \xi_x + E_\eta \eta_x + E_\zeta \zeta_x + F_\xi \xi_y + F_\eta \eta_y + F_\zeta \zeta_y \\ + G_\xi \xi_z + G_\eta \eta_z + G_\zeta \zeta_z = S \end{aligned} \quad (2)$$

where

$$\xi_x = J(y_\eta z_\zeta - y_\zeta z_\eta)$$

$$\xi_y = -J(x_\eta z_\zeta - x_\zeta z_\eta)$$

$$\xi_z = J(x_\eta y_\zeta - x_\zeta y_\eta)$$

$$\eta_x = -J(y_\xi z_\zeta - y_\zeta z_\xi)$$

$$\eta_y = J(x_\xi z_\zeta - x_\zeta z_\xi)$$

$$\eta_z = -J(x_\xi y_\zeta - x_\zeta y_\xi)$$

$$\zeta_x = J(y_\xi z_\eta - y_\eta z_\xi)$$

$$\zeta_y = -J(x_\xi z_\eta - x_\eta z_\xi)$$

$$\zeta_z = J(x_\xi y_\eta - x_\eta y_\xi)$$

$$J = 1/[x_\xi(y_\eta z_\xi - y_\xi z_\eta) - x_\eta(y_\xi z_\xi - y_\xi z_\eta) + x_\xi(y_\eta z_\eta - y_\eta z_\xi)] .$$

The transformation coefficients, ξ_x , ξ_y , ξ_z , η_x , η_y , η_z , ζ_x , ζ_y , and ζ_z , are computed numerically using second order central differencing. In the transformed domain, the grid sizes (i.e., $\Delta\xi$, $\Delta\eta$, and $\Delta\zeta$) are set to be unity. This simplifies the calculation of the transformation coefficients.

For turbulent flow computations, the present code has employed the standard $k-\epsilon$ turbulence model [7] to provide the turbulent eddy viscosity μ_t . The standard $k-\epsilon$ turbulence model (which consists of a turbulent kinetic energy equation, k -equation, and a turbulent kinetic energy dissipation rate equation, ϵ -equation) is given as:

$$(\rho k)_t + \left(\rho u_i k - \frac{\mu_{\text{eff}}}{\sigma_k} k_{x_i} \right) x_i = \rho (P_r - \epsilon) \quad (3)$$

$$(\rho \epsilon)_t + \left(\rho u_i \epsilon - \frac{\mu_{\text{eff}}}{\sigma_\epsilon} \epsilon_{x_i} \right) x_i = \rho \frac{\epsilon}{k} (C_1 P_r - C_2 \epsilon) \quad (4)$$

where the effective viscosity μ_{eff} is calculated from:

$$\mu_{\text{eff}} = \mu + \mu_t = \mu + \rho C_\mu k^2 / \epsilon$$

and the turbulent kinetic energy production term, P_r , is defined as:

$$P_r = C_\mu \frac{k^2}{\epsilon} [(u_y + v_x)^2 + (v_z + w_y)^2 + (w_x + u_z)^2 + 2(u_x^2 + v_y^2 + w_z^2)]$$

The turbulence model constants are:

$$C_\mu = 0.09 , \quad \sigma_k = 1.0 , \quad \sigma_\epsilon = 1.3$$

$$C_1 = 1.44 , \quad C_2 = 1.92 .$$

Also, the molecular viscosity μ in equation (1) is replaced by the effective viscosity μ_{eff} for turbulent flow cases.

In order to save the computational efforts, the widely used wall function approach [8] is employed to provide the near wall boundary conditions for the momentum and energy equations and the $k-\epsilon$ turbulence model. This approach avoids the requirement of integrating the governing equations up to the wall which requires a large number of additional grid points near the wall.

Equations (2), (3), and (4) form a closed set of nonlinear partial differential equations governing the fluid motion. This set of equations are to be solved by means of finite difference approximations which are performed in the transformed domain. For treating the convection terms, the hybrid scheme [9] is employed for simplicity (although other more elaborate schemes such as central differencing plus artificial dissipation scheme, QUICK scheme, or skew upwind differencing scheme, etc. can be implemented [10]). These are described in the following sections.

DISCRETIZATION OF THE EQUATIONS OF MOTION

In this section, finite difference approximations are used to discretize the governing equations, equations (2), (3), and (4). Second-order central differencing is used for the diffusion terms and the source terms. The hybrid differencing scheme [9] is employed to approximate the convection terms in the governing equations. The finite difference discretizations are performed in the transformed domain. The solution procedure for the discretized equations using a velocity-pressure correction algorithm (SIMPLE-C) of References 11 and 12 will be described in the next section.

The governing equations of motion can be represented by the following model transport equation in which ϕ denotes all the dependent variables respectively and Γ is the diffusion coefficient..

$$\begin{aligned}
 & (\rho \phi)_t + [\rho u \phi - \Gamma(\phi_\xi \xi_x + \phi_\eta \eta_x + \phi_\zeta \zeta_x)]_\xi \xi_x \\
 & + [\rho u \phi - \Gamma(\phi_\xi \xi_x + \phi_\eta \eta_x + \phi_\zeta \zeta_x)]_\eta \eta_x \\
 & + [\rho u \phi - \Gamma(\phi_\xi \xi_x + \phi_\eta \eta_x + \phi_\zeta \zeta_x)]_\zeta \zeta_x \\
 & + [\rho v \phi - \Gamma(\phi_\xi \xi_y + \phi_\eta \eta_y + \phi_\zeta \zeta_y)]_\xi \xi_y \\
 & + [\rho v \phi - \Gamma(\phi_\xi \xi_y + \phi_\eta \eta_y + \phi_\zeta \zeta_y)]_\eta \eta_y \\
 & + [\rho v \phi - \Gamma(\phi_\xi \xi_y + \phi_\eta \eta_y + \phi_\zeta \zeta_y)]_\zeta \zeta_y \\
 & + [\rho w \phi - \Gamma(\phi_\xi \xi_z + \phi_\eta \eta_z + \phi_\zeta \zeta_z)]_\xi \xi_z \\
 & + [\rho w \phi - \Gamma(\phi_\xi \xi_z + \phi_\eta \eta_z + \phi_\zeta \zeta_z)]_\eta \eta_z \\
 & + [\rho w \phi - \Gamma(\phi_\xi \xi_z + \phi_\eta \eta_z + \phi_\zeta \zeta_z)]_\zeta \zeta_z = S
 \end{aligned} \quad (5)$$

Discretization of equation (5) is performed using finite difference approximations in the transformed domain. The second order central differencing is used for approximating the diffusion terms. For the convection terms, the hybrid differencing scheme [9] is employed (i.e., using central differencing for cell Peclet number less than or equal to 2 and switching to upwind differencing when the cell Peclet number is greater than 2). The finite difference equation is arranged by collecting terms according to the grid nodes around a control volume as shown in Figure 1. The final expression is given by equation (6) in which A represents the link coefficients between grid nodes P, E, W, N, S, T, B, NE, NW, NT, NB, SE, SW, ST, SB, ET, EB, WT, and WB as shown in Figure 1.

$$A_P \phi_P = A_E \phi_E + A_W \phi_W + A_N \phi_N + A_S \phi_S + A_T \phi_T + A_B \phi_B + S_1 \quad (6)$$

where

$$\begin{aligned} S_1 = & S + A_P^\circ \phi_P^\circ + A_{NE} \phi_{NE} + A_{NW} \phi_{NW} + A_{NT} \phi_{NT} + A_{NB} \phi_{NB} \\ & + A_{SE} \phi_{SE} + A_{SW} \phi_{SW} + A_{ST} \phi_{ST} + A_{SB} \phi_{SB} \\ & + A_{ET} \phi_{ET} + A_{EB} \phi_{EB} + A_{WT} \phi_{WT} + A_{WB} \phi_{WB} \end{aligned}$$

$$A_P = A_E + A_W + A_N + A_S + A_T + A_B + A_P^\circ$$

$$A_P^\circ = \rho_P^\circ / \Delta t \quad .$$

The subscript $^\circ$ denotes the solution at the previous time level. A fully implicit formulation is employed for solving the time dependent transient problems.

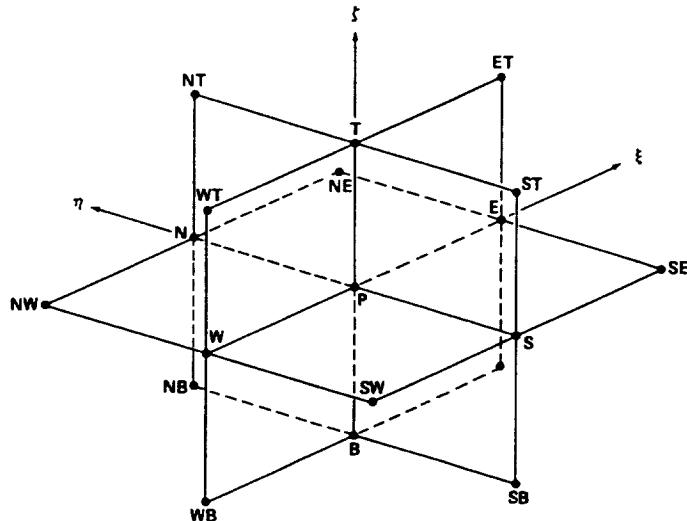


Figure 1. Three-dimensional grid structure and labeling around a grid node P.

Thus, the nonlinear equations of motion are approximated by a system of linear algebraic equations which have the form of equation (6). Only one program subroutine is designed to calculate the link coefficients and the source terms. The number of algebraic equations depends on the number of interior grid points. For a grid size of $10 \times 10 \times 10$ the number of algebraic equations to be solved would be around 512. This large system of equations are preferred to be solved by some iterative methods, such as Gauss-Seidel iteration, line-underrelaxation method [13] or Stone's method [14], etc., rather than using direct methods such as Gaussian elimination method. Only a few (6 to 10) iterations through the whole computational domain are needed and a complete convergence of the system of algebraic equations is not required. Since equation (6) is only a linearized version of the governing equations which are nonlinear and coupled in nature, solutions of the equations of motion must be obtained through global iterations among the equations. A tentative solution to equation (6) will not affect the final results significantly. On the other hand, if too many iterations are used to get a better solution of equation (6), then a great deal of computing time would be virtually wasted. However, the above argument can not be applied when the pressure correction equation (which will be derived in the next section) is solved. Since during each global iteration it is desirable to retain a divergence-free velocity field, better solution of the pressure correction equation would in effect promote the convergence of the whole numerical scheme. Therefore, more iterations are usually used to solve the pressure correction equation.

SOLUTION PROCEDURES

The governing equations used in the present analysis are nonlinear and strongly coupled. Iterative procedures are employed to drive the equations to a converged solution. It is particularly important for incompressible flow to make the flow field satisfy the continuity equation and the momentum equations at the same time. This requires a correct pressure field associated with a divergence-free velocity field. A velocity-pressure correction procedure is developed in the present study to drive the pressure field and the velocity field to be divergence free. This kind of procedure requires grid staggering between the velocity components and the locations where the pressure is estimated and stored such that the velocity field and the pressure field will not be uncoupled.

In the present study, staggering grid systems as shown in Figure 2 (for 2-D case) are used. The velocity components, u and v , are solved and stored at the grid nodes and the pressure, p , is located at the corners of the control volume of u and v . In this way, solutions of u and v can be solved using the same control volume and coupling between u , v and p can also be enforced. To estimate the pressure field, a pressure correction equation is derived approximately from the discretized momentum and continuity equations. The velocity and pressure fields are then corrected using the solutions of the pressure correction equation.

First, the finite difference momentum equations (for u , v and w) can be written as:

$$A_p^u u_p^* = \sum_i A_i^u u_i^* = p_x^* + S_u \quad (8a)$$

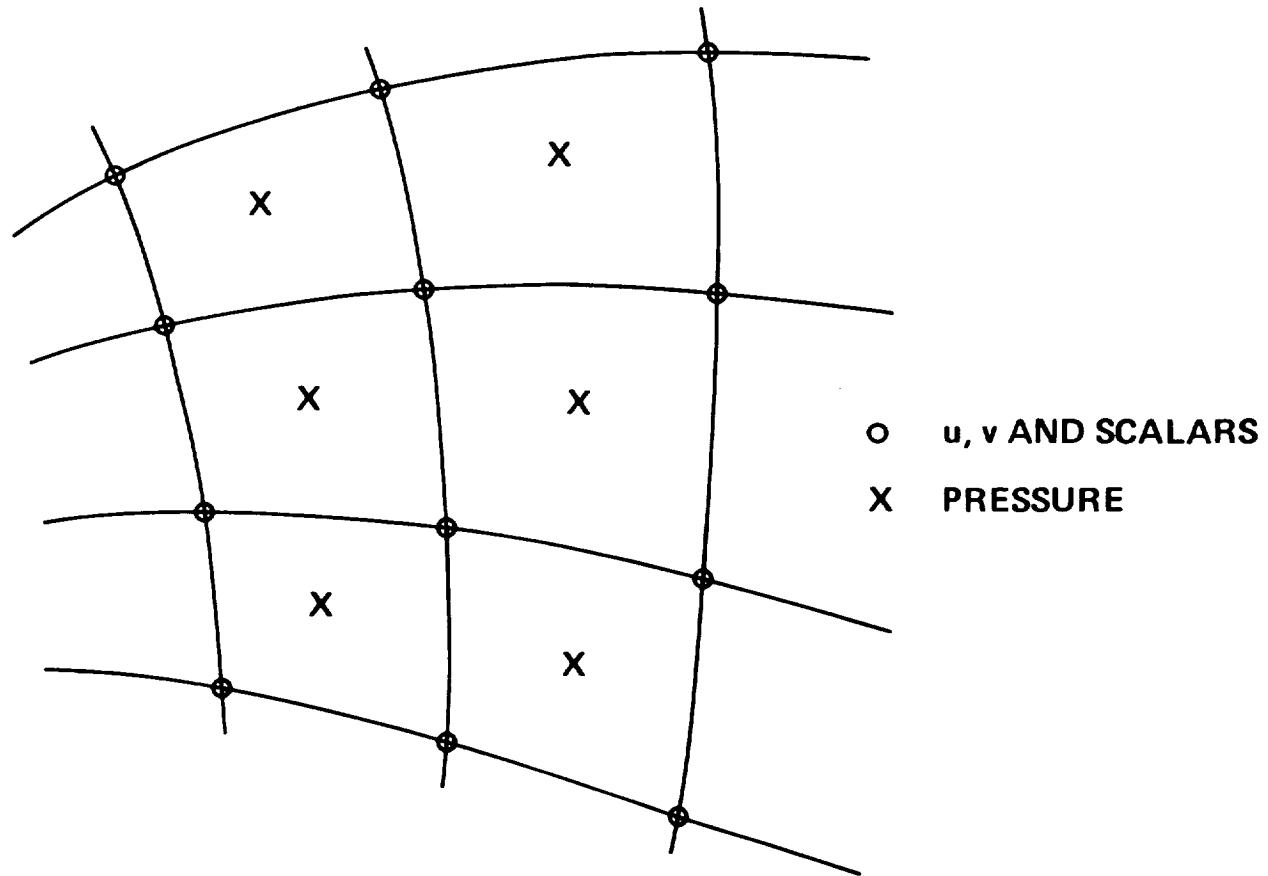


Figure 2. Locations where the variables are stored (staggering grids are used).

$$A_P^V v_P^* = \sum_i A_i^V v_i^* = P_y^* + S_v \quad (8b)$$

$$A_P^W w_P^* = \sum_i A_i^W w_i^* - P_z^* + S_w \quad (8c)$$

where u^* , v^* , w^* , and p^* represent the solutions of equations (8a) and (8b). To satisfy the continuity equation the velocities and pressure are corrected according to the following relations:

$$u = u^* + u' \quad (9a)$$

$$v = v^* + v' \quad (9b)$$

$$w = w^* + w' \quad (9c)$$

$$P = P^* + P' \quad (9d)$$

A new set of momentum equations can be constructed approximately using the divergence-free flow field, u , v , w , and p :

$$A_P^u u_P = \sum_i A_i^u u_i - p_x + S_u \quad (10a)$$

$$A_P^v v_P = \sum_i A_i^v v_i - p_y + S_v \quad (10b)$$

$$A_P^w w_P = \sum_i A_i^w w_i - p_z + S_w \quad (10c)$$

By subtracting equations (8a) through (8c) from equations (10a) through (10c), respectively, the following equations result:

$$A_P^u u_P' = \sum_i A_i^u u_i' - p_x' \quad (11a)$$

$$A_P^v v_P' = \sum_i A_i^v v_i' - p_y' \quad (11b)$$

$$A_P^w w_P' = \sum_i A_i^w w_i' - p_z' \quad (11c)$$

According to SIMPLE-C algorithm [11], equations (11a) through (11c) are rearranged to be:

$$(A_P^u - \sum_i A_i^u) u_P' = \sum_i A_i^u (u_i' - u_P') - p_x' \quad (12a)$$

$$(A_P^v - \sum_i A_i^v) v_P' = \sum_i A_i^v (v_i' - v_P') - p_y' \quad (12b)$$

$$(A_P^w - \sum_i A_i^w) w_P' = \sum_i A_i^w (w_i' - w_P') - p_z' \quad (12c)$$

The first terms on the right-hand side of equations (12a) through (12c) are neglected to simplify the formulation. Thus,

$$u_p' = - \left(\frac{1}{A_p u - \sum_i A_i u} \right) p_x' = - D_u p_x' \quad (13a)$$

$$v_p' = - \left(\frac{1}{A_p v - \sum_i A_i v} \right) p_y' = - D_v p_y' \quad (13b)$$

$$w_p' = - \left(\frac{1}{A_p w - \sum_i A_i w} \right) p_z' = - D_w p_z' \quad . \quad (13c)$$

Using the decompositions of equations (9a) through (9c), the continuity equation can be written as:

$$u_x + v_y + w_z = (u_x^* + v_y^* + w_z^*) + (u_x' + v_y' + w_z') = 0 \quad . \quad (14)$$

Substituting equations (13a) through (13c) into equation (14), the following pressure correction equation can be obtained:

$$-[(D_u p_x')_x + (D_v p_y')_y + (D_w p_z')_z] = -(u_x^* + v_y^* + w_z^*) \quad . \quad (15)$$

Equation (15) is a Poisson's equation with the source term equal to the local divergence of the flow field. To enforce the coupling between the velocity and pressure fields, the source term of equation (15) is first evaluated at the control volumes centered between the velocity nodes as shown in Figure 3. An averaged source term is then calculated at the cell center of p node for solving equation (15). In this way, the difficulties in solving the pressure correction equation, as described by Vanka [4] and Maliska [6], are eliminated. Coupling between the velocity and the pressure field is also assured.

According to the above analyses, the present numerical method contains the following solution steps:

- 1) Guess initial velocity and pressure field.
- 2) Solve for the velocity field using equations (8a) through (8c).
- 3) Solve for other scalar transport equations.
- 4) Solve the pressure correction equation, equation (15).

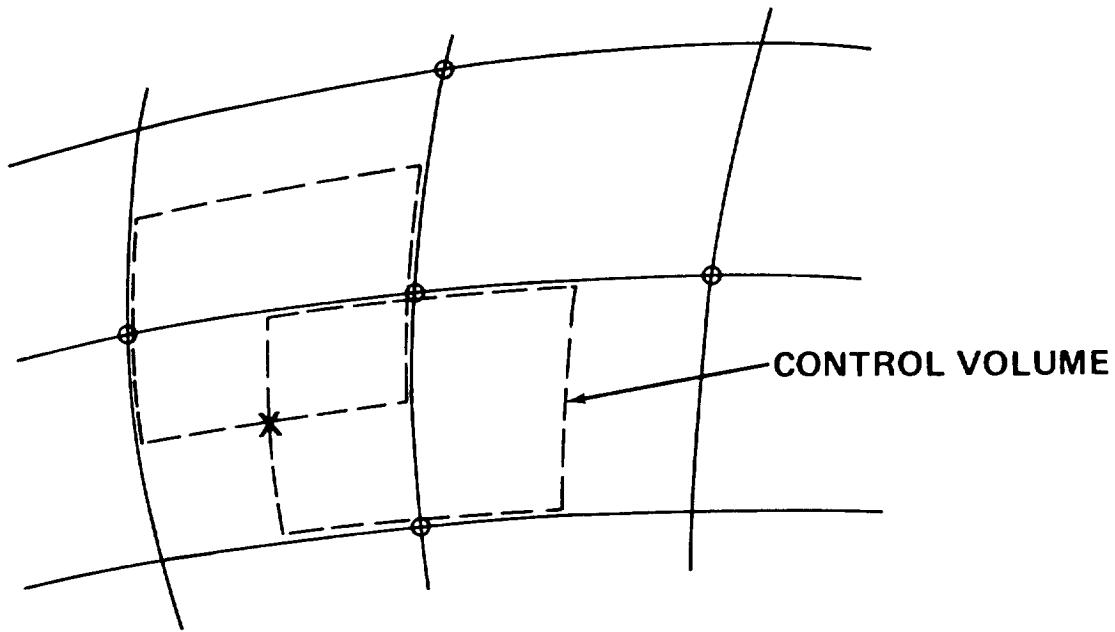


Figure 3. Control volumes where the mass conservation is evaluated for solving the pressure correction equation.

- 5) Correct the velocity and pressure fields using equations (13a) through (13c) and equation (9d).
- 6) Go back to step (2) until solution converges.

A converged solution is obtained when the following criterion is met:

$$\text{Error} \equiv (|\Delta u|_{\max} + |\Delta v|_{\max} + |\Delta w|_{\max}) / U_{\text{ref}} + |P'|_{\max} / \rho U_{\text{ref}}^2 \leq 3 \times 10^{-4},$$

where Δu , Δv , and Δw represent velocity changes during each iteration due to the solutions of the momentum equations.

In solving the momentum equations in step (b) above, underrelaxation factor of about 0.6 is recommended. With this, A_p 's in equations (8a) through (8c) are modified according to the underrelaxation factor. For the correction of velocity field, no underrelaxation is required. But the correction of pressure field should be under-relaxed slightly (around 0.9) when the grid nonorthogonality is strong. This is different from that suggested by References 11 and 12 (which recommend no under-relaxation for pressure correction).

NUMERICAL EXAMPLES

In this section, several numerical examples are employed to demonstrate the efficiency and accuracy of the present numerical method. To serve this purpose, 2-D and 3-D, laminar and turbulent flow cases are included. These cases are: (a) 2-D laminar driven square-cavity flows; (b) 2-D laminar flows over a backward-facing step; (c) 2-D turbulent flows over a backward-facing step; (d) 3-D developing laminar flow inside a 90-deg-bend square duct. Detailed descriptions and results of the computation of the above cases are included as follows.

A. 2-D Laminar Driven Square-Cavity Flows

The first test case is concerning laminar recirculating flows inside a square cavity. Only one side of the walls is moving at a constant speed tangent to that wall. This case has been studied extensively by Burggraf [15] and has often been used as one of the standard testing cases for numerical methods in solving the incompressible Navier-Stokes equations. Physical geometry and wall boundary conditions are illustrated in Figure 4. Reynolds number of the flow (based on the cavity size and the moving wall velocity) studied in the present analysis is 400. Two different mesh systems, as shown in Figure 5, are used to study the effect of grid non-orthogonality on the accuracy of the present method. The grid system of Figure 5(a) is uniform and orthogonal while the grid system of Figure 5(b) is non-uniform and non-orthogonal.

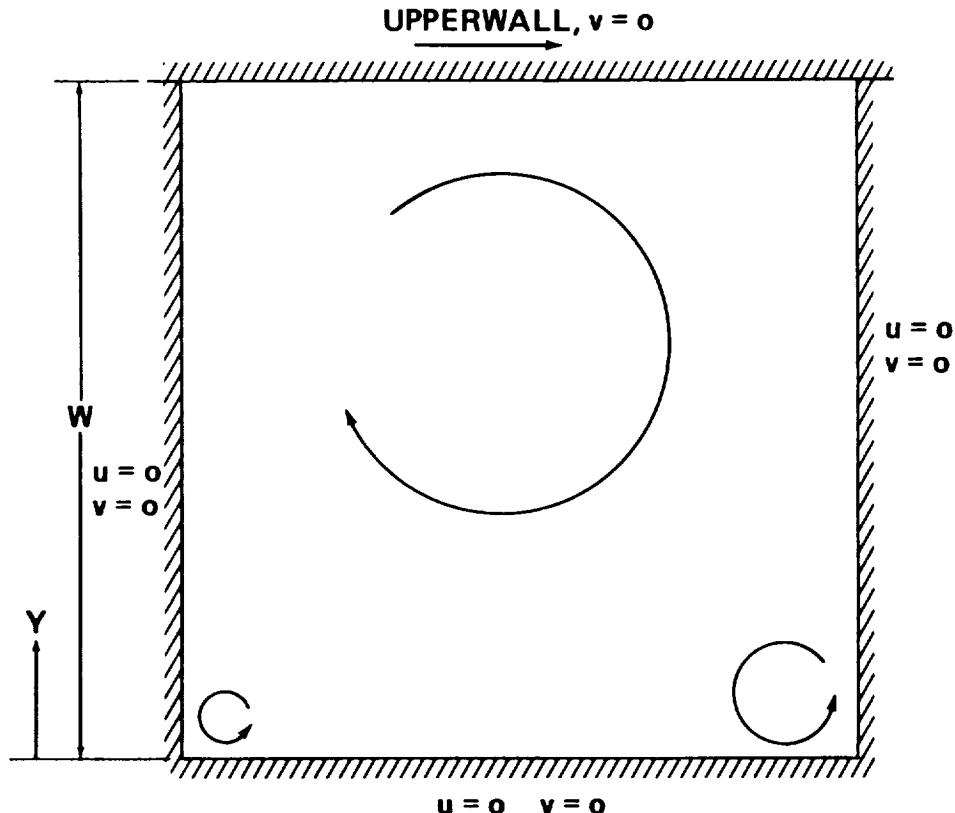
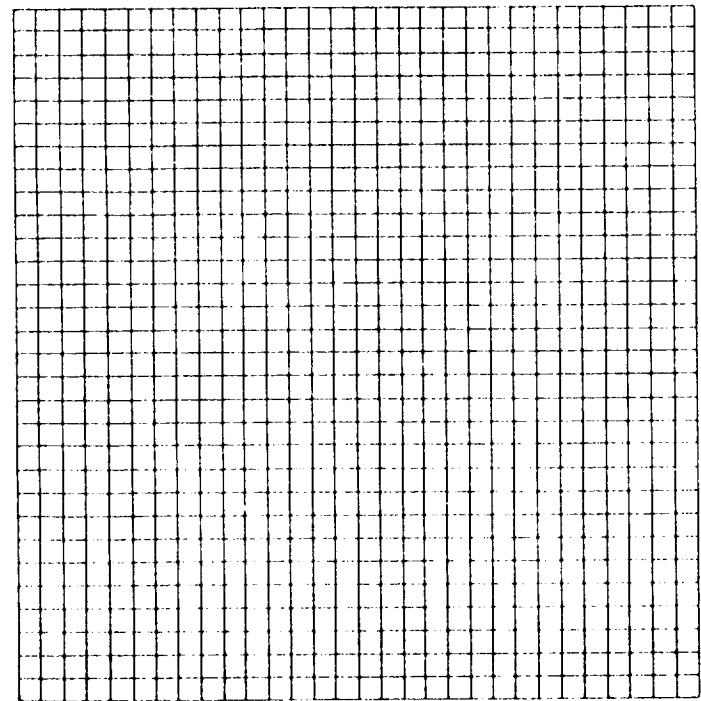
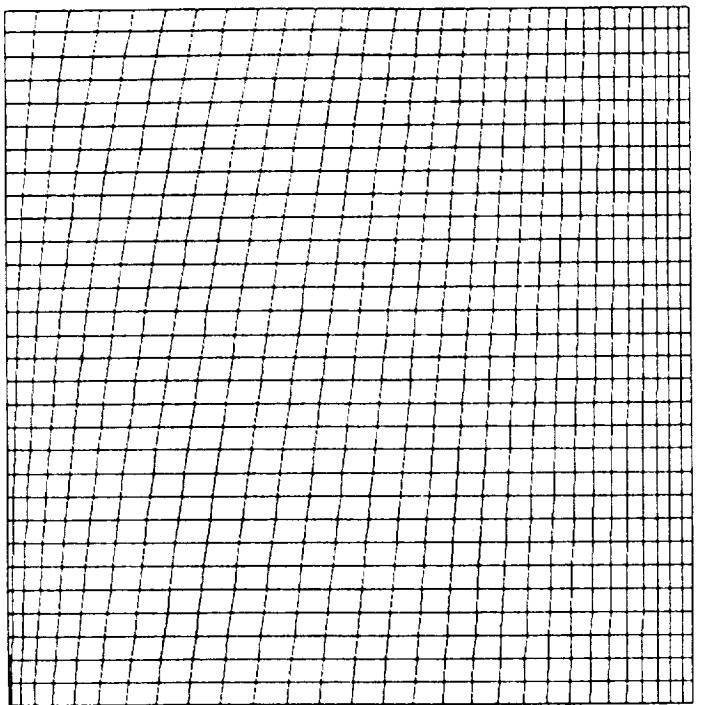


Figure 4. Physical geometry and wall boundary conditions for laminar flows inside a wall-driven square cavity.



(a)



(b)

Figure 5. Mesh systems used for driven cavity problem. (a) Uniform and orthogonal grid. (b) Nonuniform and nonorthogonal grid.

Results of the computations are shown in Figures 6 and 7. Velocity vector plots of the predicted flow fields are compared in Figure 6 for the mesh systems shown in Figure 5. Detailed comparisons of the predicted velocity profiles along the mid-section of the cavity are illustrated in Figure 7. Predicted results of Burggraf [15] are also included. Good agreements between the present calculations and those

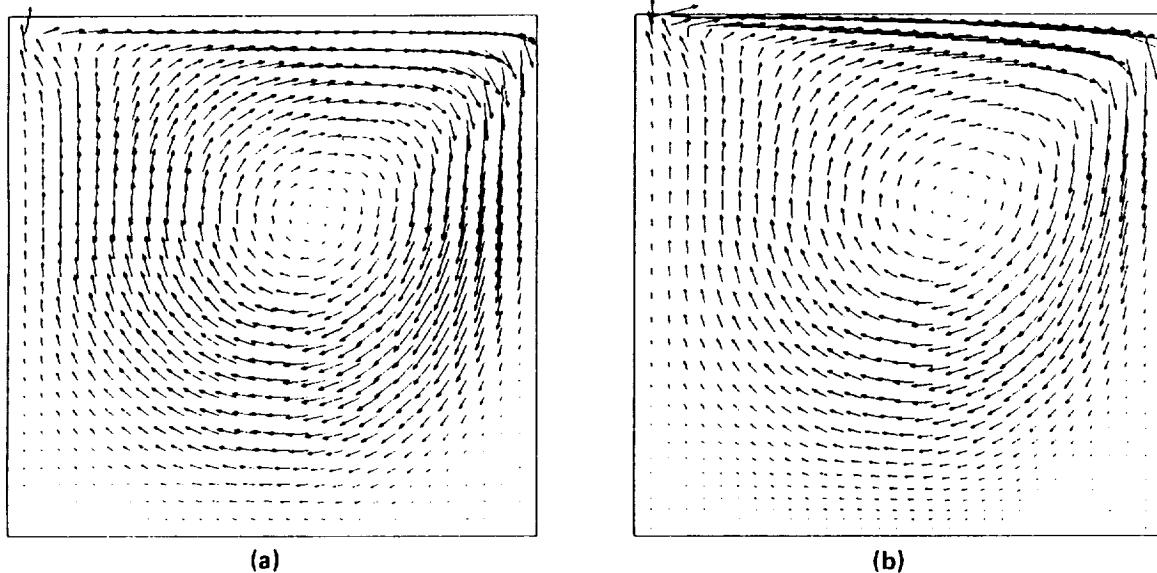


Figure 6. Velocity vector plots. (a) Orthogonal grid.
(b) Nonorthogonal grid.

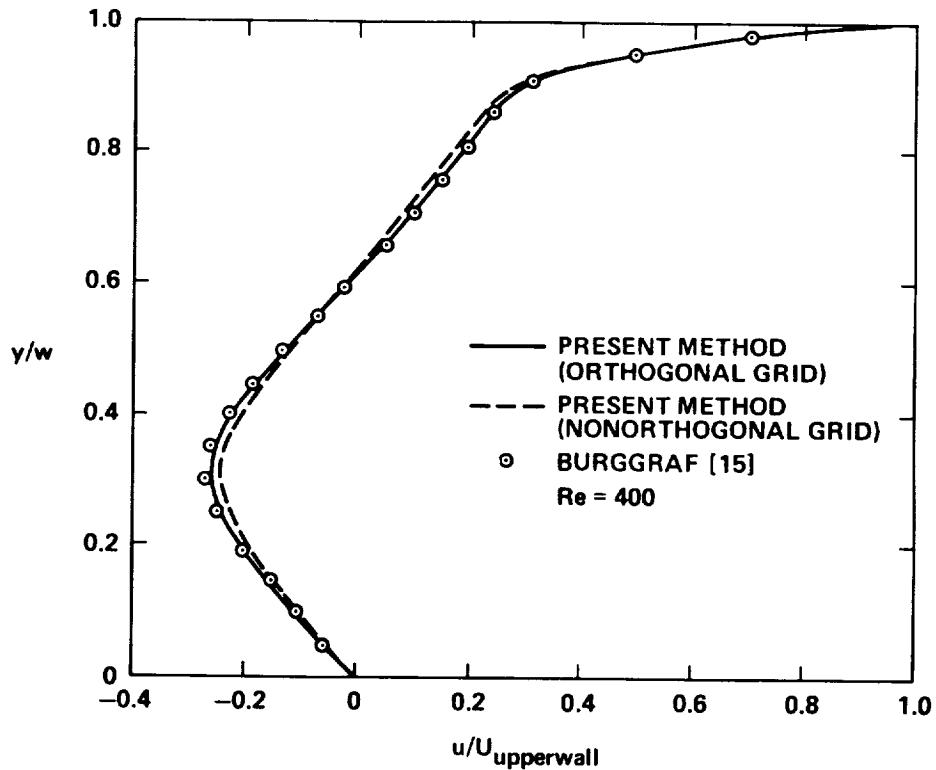


Figure 7. Comparisons of velocity profiles along the mid-section of the square cavity.

of Burggraf [15] are also included. Good agreements between the present calculations and those of Burggraf are shown in Figure 7. Discrepancies between the present predictions and Burggraf's results are mainly due to the hybrid differencing scheme used in the present method. The upwind part of the hybrid scheme produces large numerical diffusion which tends to reduce the strength of the vorticity inside the cavity. Effects of differencing schemes in approximating the convection terms on the predicted results will be studied in the next test case.

Convergence history of the computation of the present case using uniform grids is given in Figure 8 which shows that the present numerical method is quite different. Almost identical convergence rates were found for the non-orthogonal case.

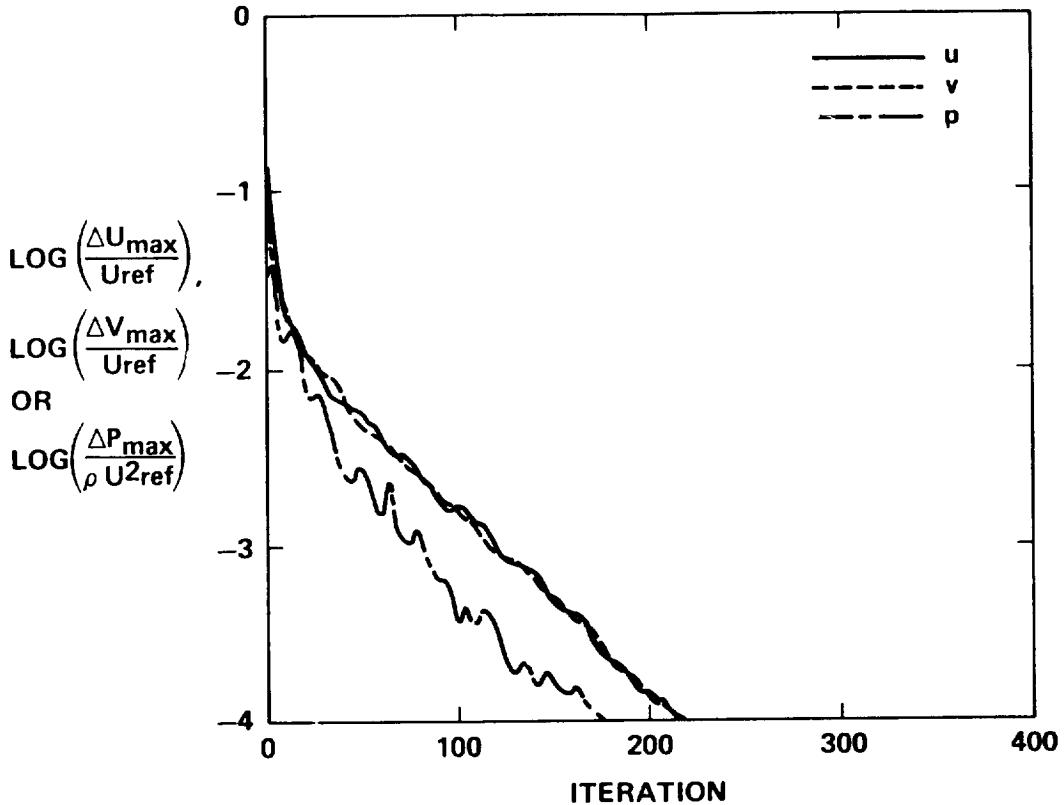


Figure 8. Convergence history for the driven cavity problem, $Re = 400$.

B. 2-D Laminar Flows Over a Backward-facing Step

This test case concerns 2-D laminar recirculating flows over a backward-facing step with 1:2 expansion ratio. The dependence of the size of the recirculation region (characterized by the reattachment length) on the Reynolds number (based on the inlet bulk velocity and twice of the inlet channel width) of the flow is of major concern. The physical domain and boundary conditions are illustrated in Figure 9 in which a fully developed laminar flow velocity profile is imposed at the flow entrance. A non-uniform grid of 45×45 was used for numerical computations. Several cases with different Reynolds numbers from 100 to 800 have been studied. An experimental and theoretical study about this problem, which results will be used as the basis of data comparisons, has been provided by Amaly et al. [16].

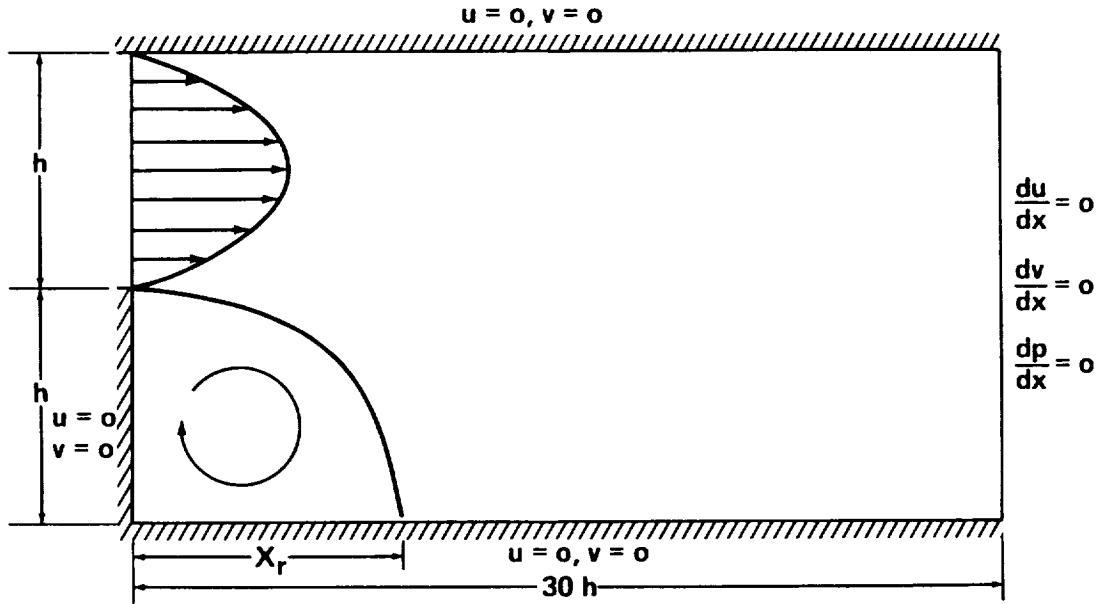


Figure 9. Physical geometry and boundary conditions of laminar flows over a backward-facing step (1:2 expansion).

To save computational efforts, the solution of one case with Reynolds number 100 is obtained in the first run. Then, a series of cases with increasing Reynolds numbers (i.e., 100, 200, 300, 400, 600, and 800) are calculated using the preceding results of lower Reynolds number as the initial guesses of the flow field. In this way, an average of 500 iterations for each case were needed to obtain converged solutions.

Two different differencing schemes in approximating the convection terms are employed to demonstrate the effects of the differencing schemes on the predictions. One of the schemes is the widely used hybrid scheme [9]. The other scheme employs the central differencing scheme plus an artificial dissipation term used to stabilize the solution which is similar to the one used by Rhie [17]. The artificial dissipation term becomes effective only when the cell Peclet number (or cell Reynolds number) exceeds 10.

Results of the present predictions using two different differencing schemes are compared with the experimental measurements [16] and other predictions as shown in Figure 10. It can be seen clearly from Figure 10 that the present method with hybrid scheme gives results similar to those predicted by TEACH code [16] while the present method with central differencing and artificial dissipation reveals predictions close to those predicted by INS3D [18] and the method of Kim and Moin [19]. This is reasonable since the TEACH code and the present method (with the first scheme) use the hybrid scheme which introduces large numerical dissipation by its upwind part (for cell Peclet number greater than 2). This tends to reduce the reattachment length for Reynolds number greater than 400. The second scheme, which is similar to the ones used in INS3D and the method of Kim and Moin, has the numerical accuracy close to second order by setting the artificial dissipation to be as small as the solution stability permits such that better accuracy of the predictions is expected.

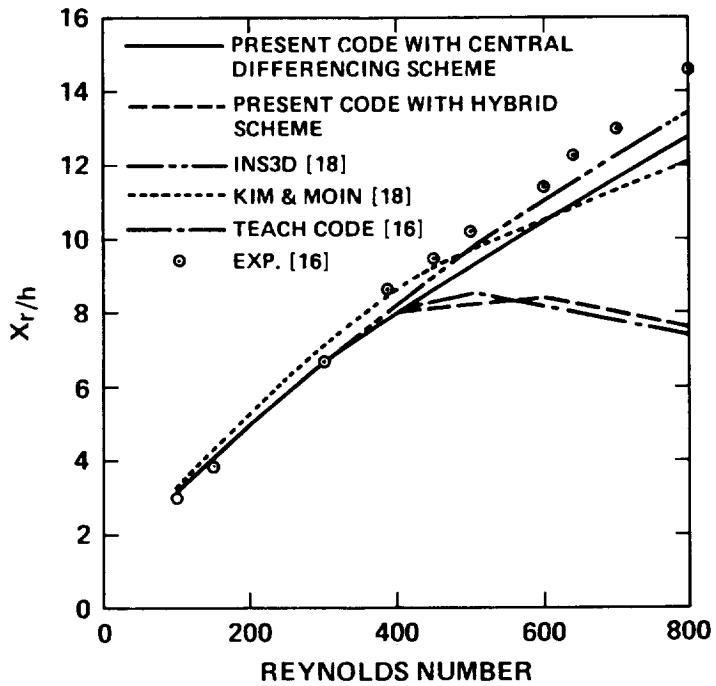


Figure 10. Reattachment length versus Reynolds number for laminar flows over a backward-facing step (1:2 expansion).

Stream function plots of the predictions using the two differencing schemes for Reynolds number 600 are compared in Figure 11. It is shown in Figure 11 that the second scheme gives a smooth shape of the recirculation zone while the hybrid scheme gives a sudden change in the shape of the recirculation region upstream of the reattachment point. Also, larger sizes of the separation regions on the step side wall and along the upper wall are predicted using the second scheme.

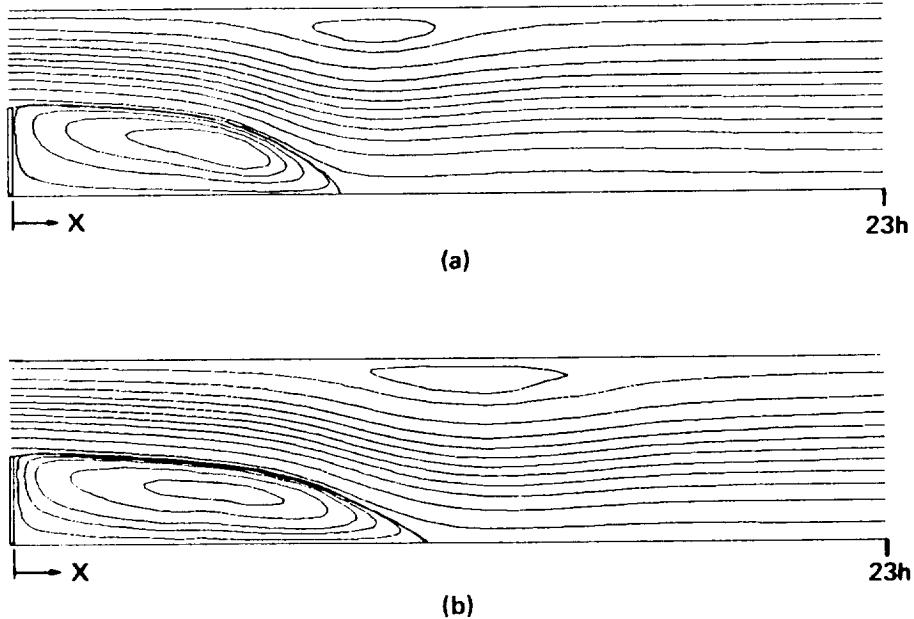


Figure 11. Streamline plots for laminar flow over a backward-facing step (1:2 expansion). (a) Hybrid Scheme. (b) Central differencing plus artificial dissipation scheme.

C. 2-D Turbulent Flows Over a Backward-Facing Step

In order to demonstrate the applicability of the present method to turbulent flow case, one of the standard test cases presented in the Stanford Conference [20] is selected here (i.e., turbulent flow over a 2:3 expansion backward-facing step). The standard $k-\epsilon$ turbulence model was used to provide the eddy viscosity for the transport equations. The physical geometry and boundary conditions imposed are shown in Figure 12. The calculation domain extends upstream of the expansion plane by 4 step heights and downstream of the expansion plane by 30 step heights to assure a fully developed velocity profile at the exit. A uniform velocity profile is located at the inlet plane. A 45×42 grid was used in the computation. 300 iterations were required to obtain converged solutions. Only hybrid differencing schemes were used in this case.

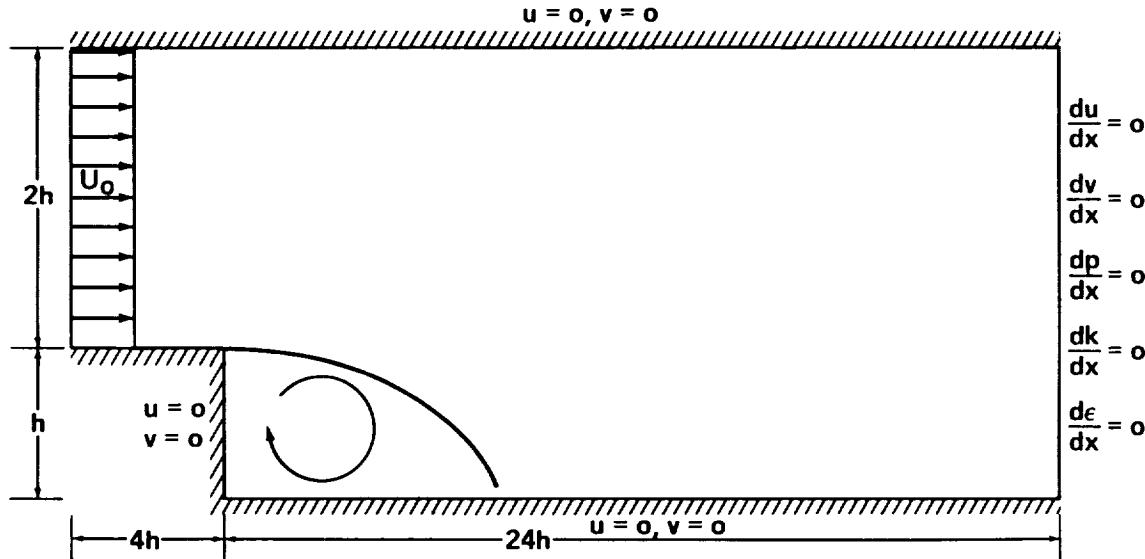


Figure 12. Physical geometry and boundary conditions of turbulent flows over a backward-facing step (2:3 expansion).

Results of the computation are shown in Figures 13, 14, and 15. These results are compared with the experimental measurements [20]. The under-prediction of the reattachment length is mainly due to the fast development of the mixing layer downstream of the expansion plane which is the characteristics of the standard $k-\epsilon$ turbulence model. Numerical diffusion provided by the hybrid scheme also contributes some part to the discrepancies between the predictions and measurements.

D. Developing Laminar Flow Inside a 90-Deg-Bend Square Duct

This test case simulates a three-dimensional developing laminar flow inside a 90-deg-bend square duct as illustrated in Figure 16(a). The symmetry plane is located at $z = 0$ where the symmetric boundary conditions are imposed. A fully developed velocity profile of laminar flow inside a straight square duct is prescribed at the entrance which is 2.8 duct widths upstream of the bend. A zero pressure

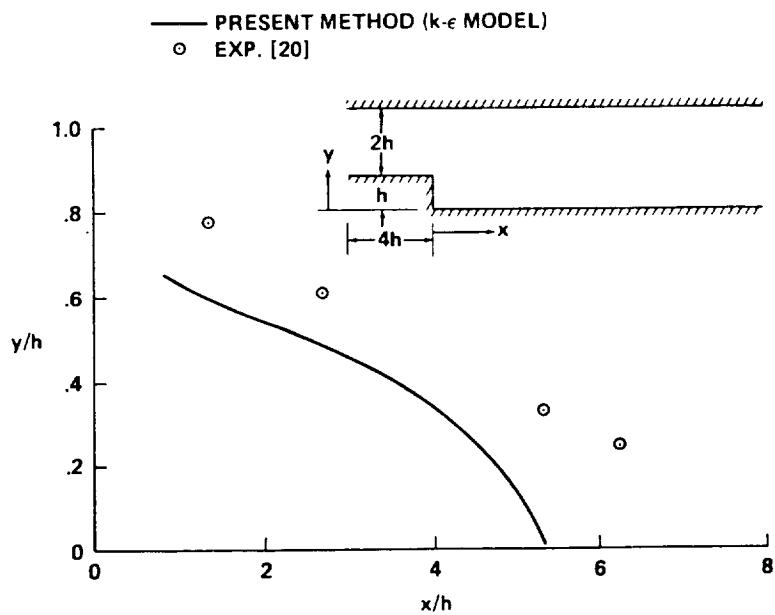


Figure 13. Locus of flow reversal inside the recirculation region for turbulent flow over a backward-facing step (2:3 expansion).

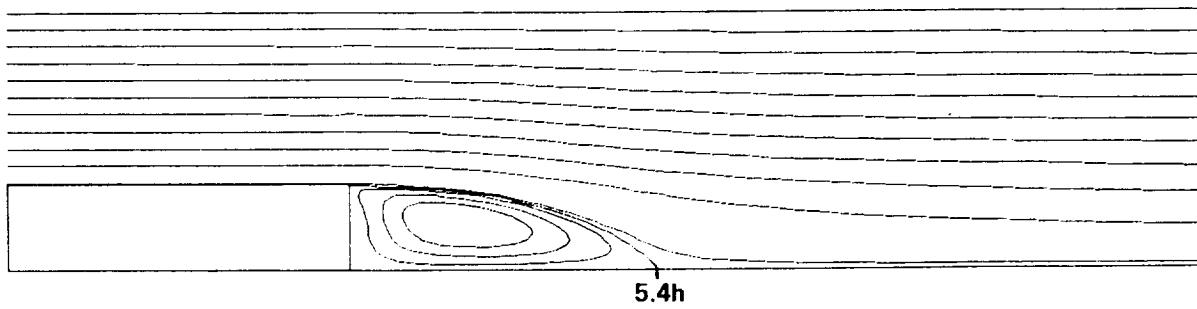


Figure 14. Stream line pattern of turbulent flow over a backward-facing step with 2:3 expansion ratio.

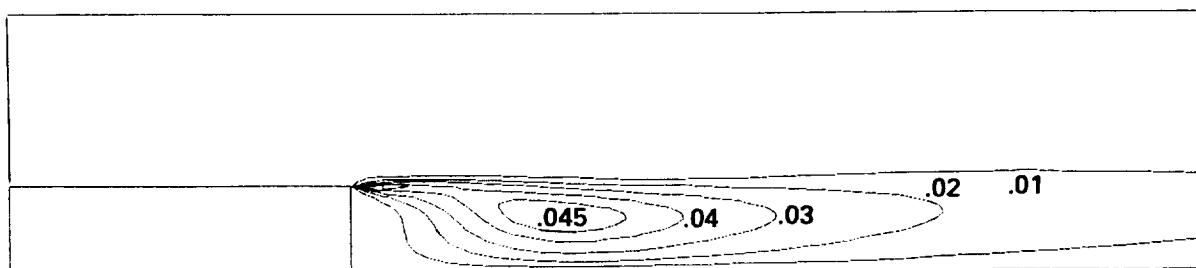


Figure 15. Contours of turbulent kinetic energy (k/U_0^2) of turbulent flow over a backward-facing step with 2:3 expansion ratio.

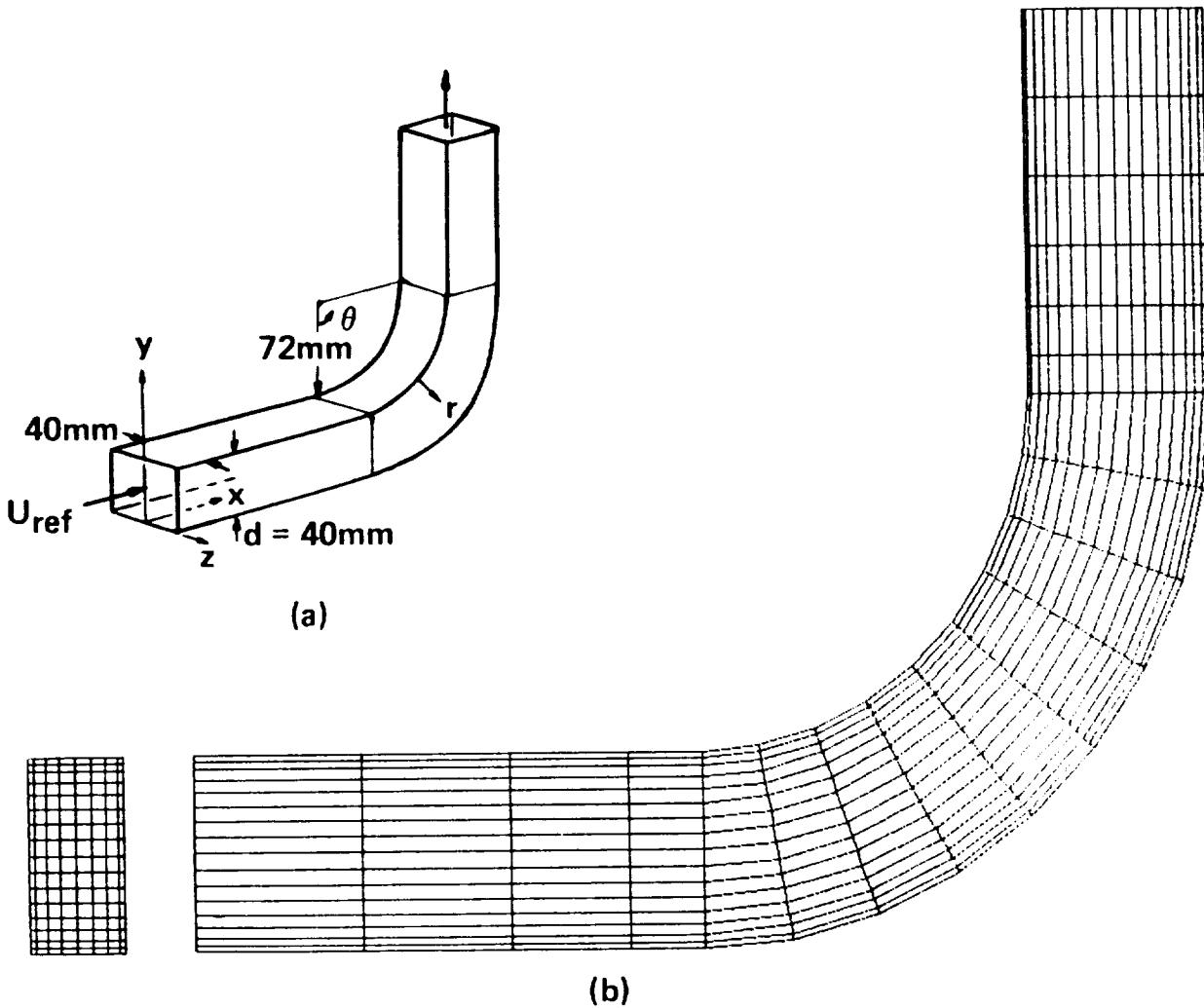


Figure 16. Geometry and mesh system of a 90-deg-bend square duct developing laminar flow problem.

gradient exit (which is 4.5 duct widths downstream of the bend) boundary condition is imposed. The Reynolds number of the flow (based on the duct hydrolic diameter and the inlet bulk velocity) is 790. A $21 \times 18 \times 10$ grid was used for numerical computations. The front view and side view of the mesh system are illustrated in Figure 16(b). Experimental measurements of Humphrey et al. [21] are used for data comparisons.

Velocity vector plots on three sections along the main flow directions (i.e., on $x-y$ plane) are shown in Figure 17. Secondary flow patterns at several stations across the bend are illustrated in Figure 18. These results are very similar to those obtained by Vanka [22] and Rhie [23]. Grid sizes of $50 \times 22 \times 15$ and $58 \times 15 \times 11$ were used by Rhie and Vanka, respectively. The present investigation, using only less than half of their grid numbers, gives highly encouraging results. Detailed comparisons between the measured and the predicted main velocity profiles are given in Figure 19.

With the above successful numerical simulations, it is believed that the present numerical method can be applied to general fluid dynamics problems with good numerical accuracy and efficiency.

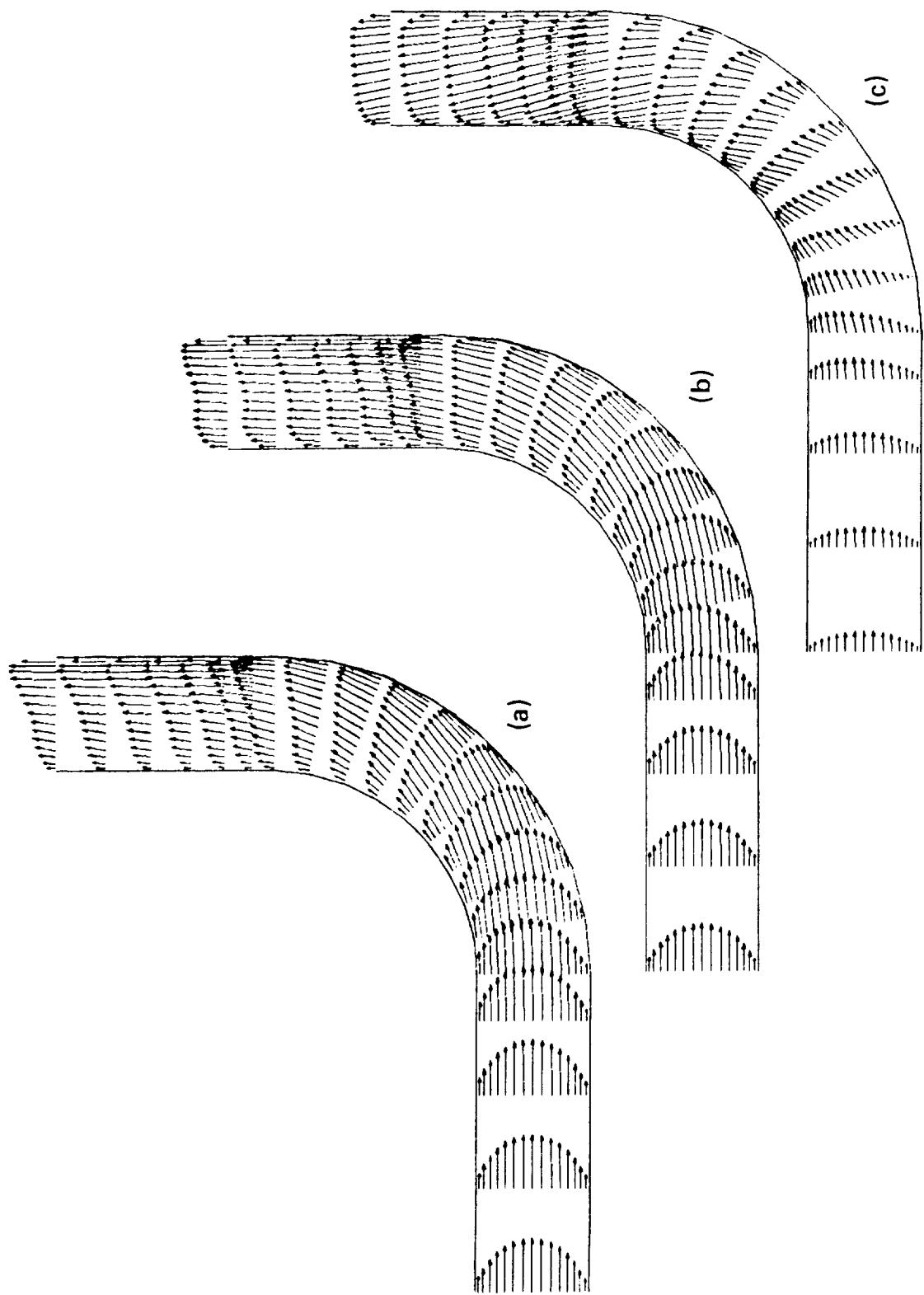


Figure 17. Primary velocity patterns of laminar flow inside a 90-deg-bend square duct.
(a) $z/d = 0.0$. (b) $z/d = 0.25$. (c) $z/d = 0.48$.

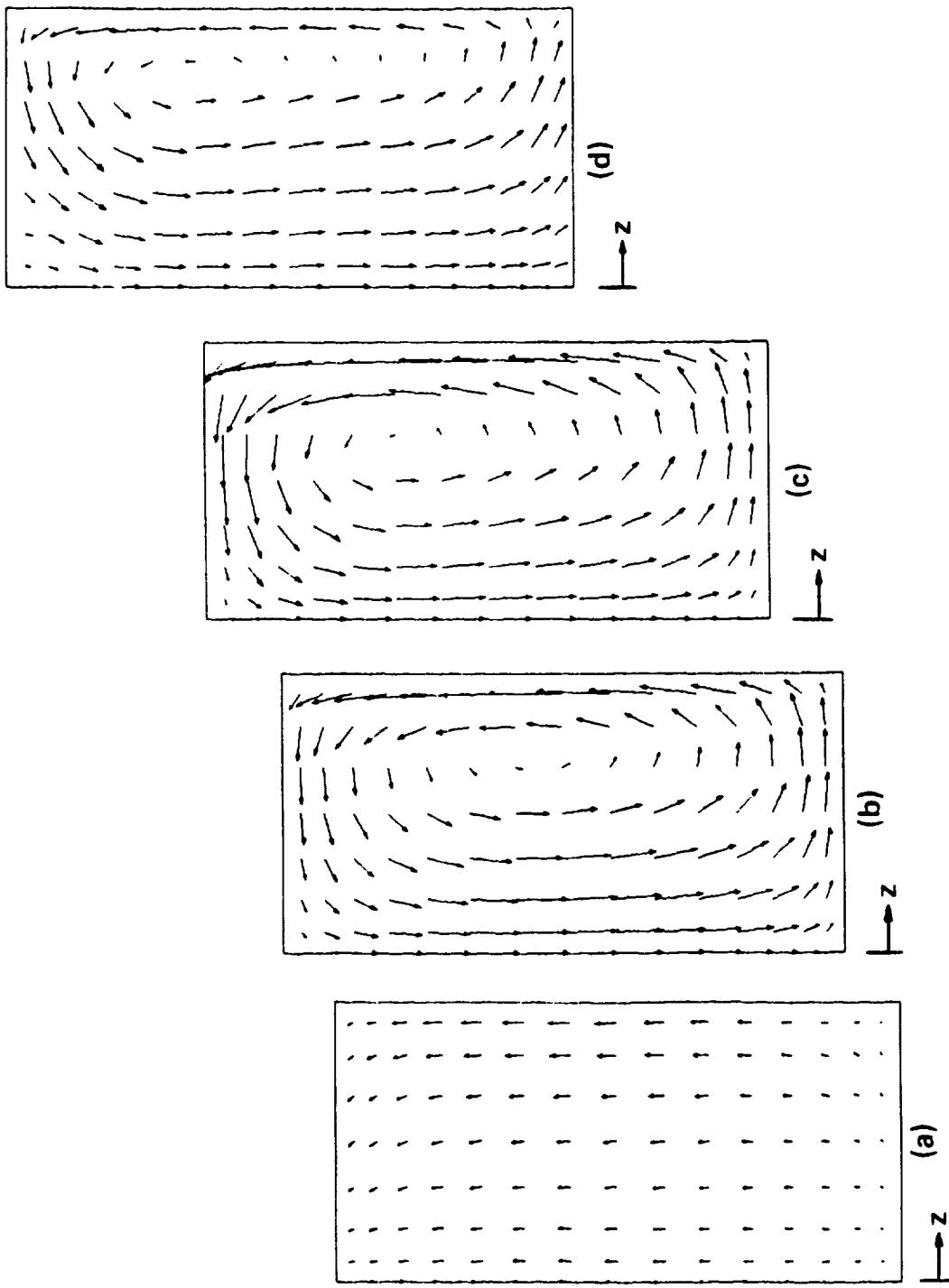


Figure 18. Secondary velocity patterns of laminar flow inside a 90-deg bend.
 (a) $\theta = 0$ deg. (b) $\theta = 30$ deg. (c) $\theta = 60$ deg. (d) $\theta = 90$ deg.

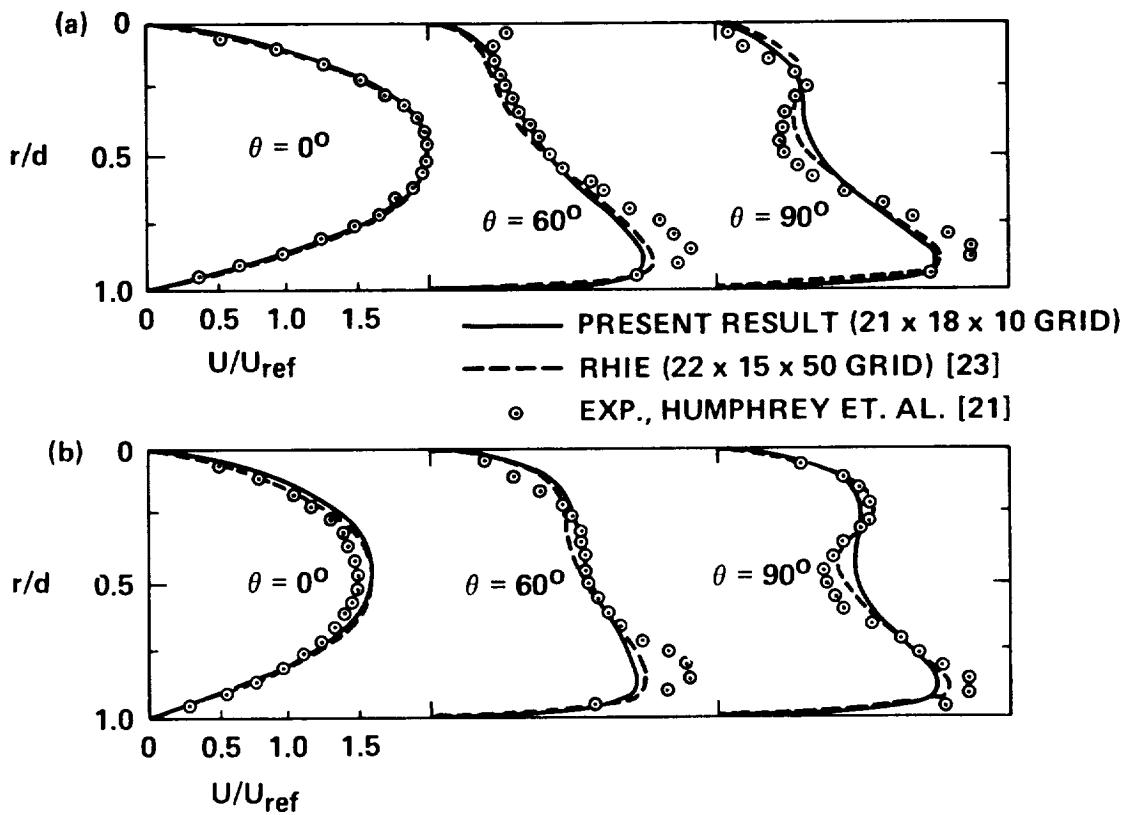


Figure 19. Primary velocity profiles for a 3-D 90-deg-bend square duct. (a) $z/d = 0.0$. (b) $z/d = 0.25$.

CONCLUSIONS

A numerical method for solving the steady or transient incompressible Navier-Stokes equations in three-dimensional body-fitted coordinate systems has been developed. In the present paper, the basic numerical algorithms and grid arrangements have been described in detail. A brief user's guide to the present computer code (CNS3D) has been included in Appendix A. A program listing has also been attached in Appendix C.

Several numerical testing examples of 2-D and 3-D, laminar and turbulent flow problems included in the present work have demonstrated that the present computer code is efficient and robust, and can be used as a reliable tool for engineering design and analysis applications. Applications of the present code to the internal turbulent flow problems of the SSME will be presented in the future publications.

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APPENDIX A
COMPUTER CODE STRUCTURES AND USER'S GUIDE

The global structure of the present computer code (CNS3D) can be represented by a flow chart, shown in Figure A-1. The user is referred to Appendix C for detailed information. First, the program requests inputs, from logic unit 5 (LU = 5), of program control parameters that specify the maximum number of iterations, the type of flow (i.e., laminar or turbulent), number of iterations for solving the pressure correction equation (typically 10), and underrelaxation factors for solving the transport equations, etc. This is followed by the definitions of all the program constants including turbulence model constants (these constants are subject to change according to the user's specific flow problem). Next, the program asks for inputs of the initial flow field guess from a restart file (LU = 8) which contains the grid system coordinates and flow field data that may be created by the user (including grid generation) or obtained from the previous solutions. Format of this data file is also subject to change according to the user's preference. Next, wall boundary control parameters, boundary grid normal distance to the wall, and wall boundary direction cosine are calculated in subroutine DIRCOS. Subroutine TRANF is then invoked to obtain the grid transformation coefficients. Before the solution procedure starts, the inlet mass flow rate is calculated which will be used to control the outlet mass flow rate to enhance mass conservation. The solution procedures consist of a series of subroutine calls to SOLVEQ starting from the solutions of the velocity vectors, u , v , and w , and then the solutions of scalar quantities (including the energy equation and the turbulence model equations) and finally the solution of the pressure correction equation to update the velocity and pressure field such that a divergence-free flow field can be retained.

After each global iteration of the solution procedures, the numerical of iterations and the maximum flow field corrections are checked with the initial settings. If the convergence criterion is satisfied or the number of iterations reaches the prescribed value then the solution procedures stop and the flow field solutions will be written on the pre-assigned disc file (LU = 7).

For instance, if a steady-state laminar flow problem (Reynolds number of 600) is of interest and a converged solution is expected within 300 iterations and the number of iterations for solving the pressure correction equation is 10 and the underrelaxation factors are 0.5 and 0.95 for transport equations and pressure correction equation, respectively, the first inputs from LU = 5 would be:

<u>Line</u>
1. 300 1 10 1
2. 0.5 0.5 0.5 0.95 0.5 0.5 0.5 0.5
3. 600. 0.0

In the second input sequence (i.e., from restart file), the program reads in $L \times M \times N$ lines of data records. See Figure A-2 for grid structures. Notice that the program requires variable dimensions of $(L+1, M+1, N+1)$ for solving the pressure correction equation. It is important to check the COMMON table for proper variable dimensions.

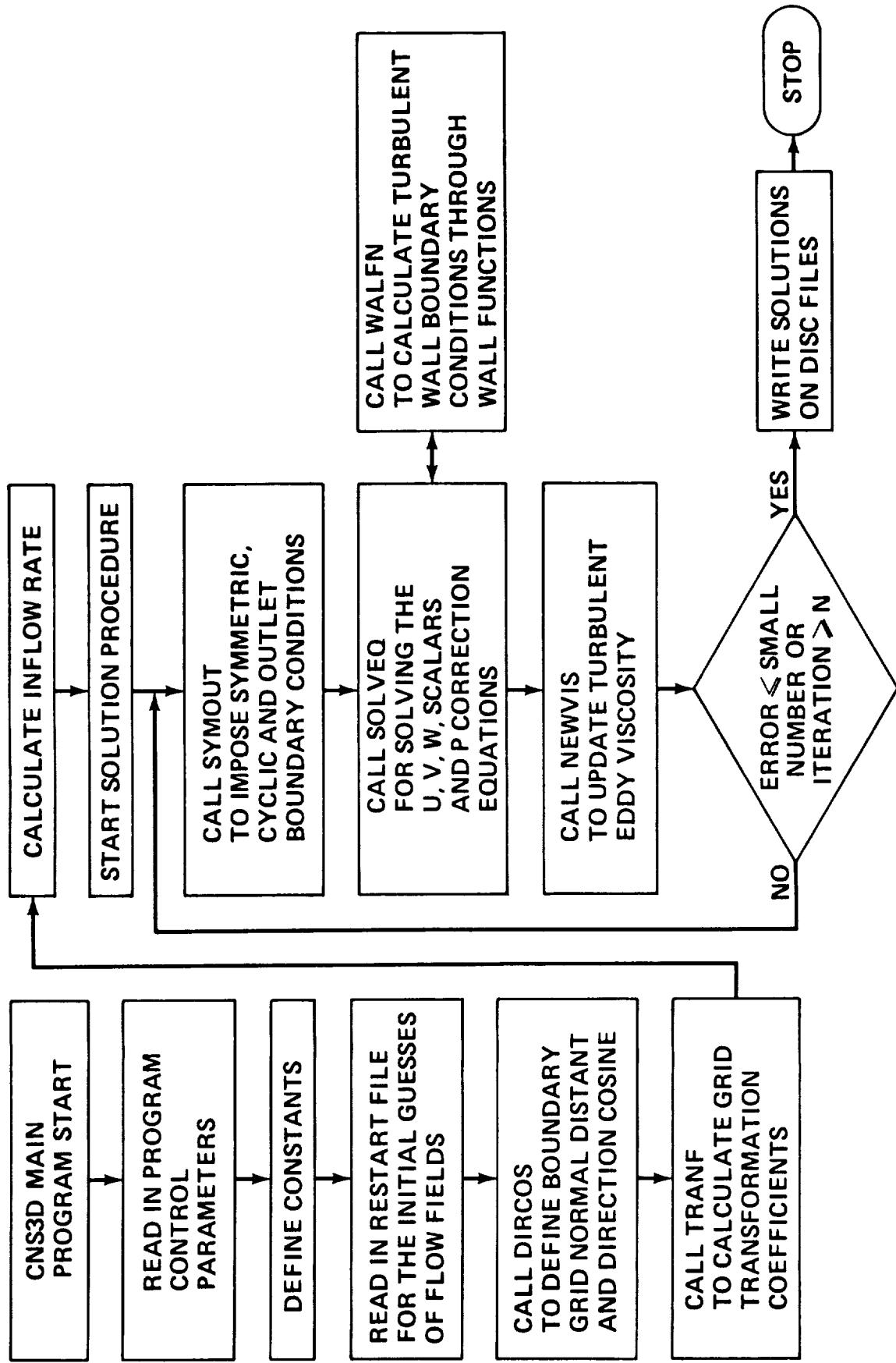


Figure A-1. Global structure of the present computer program CNS3D.

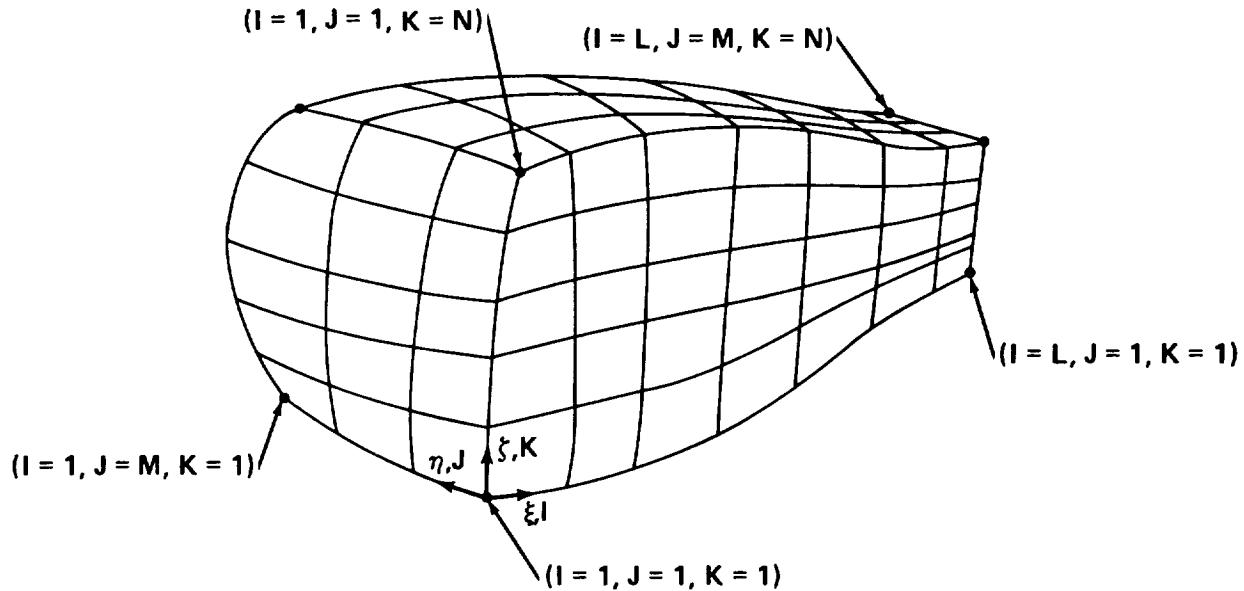


Figure A-2. Grid mesh structures for 3-D calculations.

If the flow problem involves symmetric or cyclic boundary conditions, then the user can look into the subroutine SYMOUT to specify the appropriate boundary conditions (the conditions shown in the program listing of Appendix C are for symmetric boundary conditions at $K = 1$). For cyclic boundary conditions at $K = 1$ and $K = N$, data at $K = 2$ and $K = N-1$ can be used to obtain boundary conditions at $K = 1$ and $K = N$ by requiring same gradients across $K = 1$ and $K = N$. This method is simple but will lag the boundary conditions by one iteration. A direct method without lagging the boundary conditions can also be employed by modifying the subroutine of linear algebra solver LINERX such that the boundary conditions can be part of the solution of the TDMA (tridiagonal matrix) solver.

In case of incorporating different wall functions for turbulent flow problems (e.g., References 24, 25, and 26), subroutine BOUNC and WALFN can be modified according to the user's method of wall treatments. The set of wall functions given in the program listing of Appendix C are derived from the conventional wall law and the equilibrium turbulent kinetic energy relations [8].

When additional source terms are to be added to the transport equations due to flow problem requirements, modifications to the source term calculation section in the subroutine SOLVEQ can be carried out. Notice that in the subroutine SOLVEQ source terms for the velocities v and w are included in the u -source section. Purpose of this is to save some computing time since these source terms use similar calculation routines.

Some times it is required to solve more transport equations other than the basic ones included in Appendix C. To modify the program to incorporate more equations,

several changes are necessary. First, new variables must be added to the COMMON table (this can be easily done through the computer editor session). Then, new source term sections are added in the SOLVEQ subroutine. Finally, subroutines WALFN and SYMOUT are modified to incorporate the new variables into the boundary condition setting routines.

APPENDIX B
LIST OF FORTRAN SYMBOLS

A(K)	= Matrix elements of a tridiagonal matrix
AB(I,J,K)	= Link coefficients through the bottom face of a control volume
AE(I,J,K)	= Link coefficients through the east face of a control volume
ALC	= Underrelaxation factor for symmetry or cyclic boundary conditions
ALE	= Underrelaxation factor for the ϵ -equation
ALK	= Underrelaxation factor for the k-equation
ALP	= Underrelaxation factor for the pressure correction equation
ALU	= Underrelaxation factor for the u-equation
ALV	= Underrelaxation factor for the v-equation
ALVIS	= Underrelaxation factor for the effective viscosity
ALW	= Underrelaxation factor for the w-equation
AN(I,J,K)	= Link coefficient through the north face of a control volume
ANAB	= Sum of the link coefficients at all faces
ANV1(I)	= Modified wall boundary link coefficient for v-equation
ANW1(I)	= Modified wall boundary link coefficient for w-equation
AP(I,J,K)	= Sum of the link coefficients around a control volume
APO(I,J,K)	= Link coefficients in time marching direction
ARDEN	= Area times density across a section in physical domain
AREA	= Area of a section in physical domain
AS(I,J,K)	= Link coefficients through the south face of a control volume
AT(I,J,K)	= Link coefficients through the top face of a control volume
AW(I,J,K)	= Link coefficients through the west face of a control volume
B(K)	= Matrix elements of a tridiagonal matrix
BB(I,J,K)	= Coefficients in Stone's partial factorization technique
BOUNC	= Subroutine for getting turbulent wall boundary conditions through wall functions

C(K)	= Matrix elements of a tridiagonal matrix
C1	= Turbulence model constant, = 1.44
C2	= Turbulence model constant, = 1.92
CB	= Convective flux through the bottom face of a control volume
CE	= Convective flux through the east face of a control volume
CK	= Von Karman constant, = 0.4
CMU	= Turbulence model constant, = 0.09
CMU1	= CMU**0.25
CMU2	= CMU**0.75
CN	= Convective flux through the north face of a control volume
CS	= Convective flux through the south face of a control volume
CT	= Convective flux through the top face of a control volume
CW	= Convective flux through the west face of a control volume
CX(I,J,K)	= Grid transformation coefficient, ξ_x
CY(I,J,K)	= Grid transformation coefficient, ξ_y
CZ(I,J,K)	= Grid transformation coefficient, ξ_z
D(K)	= Matrix elements of a tridiagonal matrix
DDB	= Diffusive flux through the bottom face of a control volume
DDE	= Diffusive flux through the east face of a control volume
DDN	= Diffusive flux through the north face of a control volume
DDS	= Diffusive flux through the south face of a control volume
DDT	= Diffusive flux through the top face of a control volume
DDW	= Diffusive flux through the west face of a control volume
DE(I,J,K)	= Turbulent kinetic energy dissipation rate, ϵ
DEO(I,J,K)	= DE at the previous time level
DEN(I,J,K)	= Density of the fluid
DENO(I,J,K)	= DEN at the previous time level
DENC	= Density at the center of a surface

DENIN	= Initial value of density of the fluid
DIRCOS	= Subroutine for calculating the boundary grid sizes and direction cosines
DITM	= Wall boundary average value of dissipation rate
DK(I,J,K)	= Turbulent kinetic energy, k
DKO(I,J,K)	= DK at the previous time level
DTT	= Time step size, Δt
DU(I,J,K)	= Diffusive coefficient for the p' -equation
DV(I,J,K)	= Diffusive coefficient for the p' -equation
DW(I,J,K)	= Diffusive coefficient for the p' -equation
E	= Wall law constant, = 9.01069
EREEXT	= Convergence criterion tolerance
ERRE	= Maximum correction in ϵ
ERRF	= Maximum correction of a variable
ERRK	= Maximum correction in k
ERRM	= Maximum correction in p
ERRU	= Maximum correction in u
ERRV	= Maximum correction in v
ERRW	= Maximum correction in w
EX(I,J,K)	= Grid transformation coefficient, η_x
EY(I,J,K)	= Grid transformation coefficient, η_y
EZ(I,J,K)	= Grid transformation coefficient, η_z
F(I,J,K)	= Tentative variable of the transport equations
FO(I,J,K)	= F at the previous time level
F1(I,J,K)	= Variable quantity at the previous iteration step
FLOW	= Outlet mass flow rate
FLOWIN	= Inlet mass flow rate
GEN(I,J,K)	= Turbulent kinetic energy production rate

HINUM = Large number, = 1.E30
 I = Index along the ξ grid lines
 IBC(I) = Boundary grid index
 IE = Index assigned for the transport equations
 IG = Problem control parameter, =1 for laminar flow and =2 for turbulent flow
 IITO = Total number of wall boundary grids
 IITY = Boundary grid face type
 IJLO(I,J,K) = Boundary grid sequential order
 INIT = Subroutine for initializing variables
 INPRO = Logical parameter for updating the effective viscosity
 INSOE = Logical parameter for solving the ϵ -equation
 INSOK = Logical parameter for solving the k-equation
 INSOP = Logical parameter for solving the p' -equation
 INSOT = Logical parameter for solving the T-equation
 INSOU = Logical parameter for solving the u-equation
 INSOV = Logical parameter for solving the v-equation
 INSOW = Logical parameter for solving the w-equation
 IS = Starting value of I of the solution domain
 ISWE = Number of sweeps for solving the ϵ -equation
 ISWK = Number of sweeps for solving the k-equation
 ISWP = Number of sweeps for solving the p' -equation
 ISWU = Number of sweeps for solving the u-equation
 ISWV = Number of sweeps for solving the v-equation
 ISWW = Number of sweeps for solving the w-equation
 IT = Last value of I of the solution domain
 ITT = Number of time steps
 J = Index along the η grid lines

JBC(I) = Boundary grid index
 JS = Starting value of J of the solution domain
 JT = Last value of J of the solution domain
 K = Index along the ζ grid lines
 KBC(I) = Boundary grid index
 KS = Starting value of K of the solution domain
 KT = Last value of K of the solution domain
 L = Maximum dimension of grid system in I direction
 LO = L + 1
 L1 = Starting point of blockage region in I direction
 L2 = Last point of blockage region in I direction
 LINERX = Subroutine for solving algebraic equations
 LT = L - 1
 M = Maximum dimension of grid system in J direction
 MO = M + 1
 M1 = Starting point of blockage region in J direction
 M2 = Last point of blockage region in J direction
 MC(I,J,K) = Wall blockage region control parameter
 MT = M - 1
 N = Maximum dimension of grid system in K direction
 NO = N + 1
 N1 = Starting point of blockage region in K direction
 N2 = Last point of blockage region in K direction
 NEWVIS = Subroutine for updating the effective viscosity
 NLIMT = Limit of maximum number of iterations
 NT = N - 1
 P = Static pressure (relative)
 PCXI = Pressure gradient, P_ξ

PDUV	= Blockage control parameter for link coefficients
PEDA	= Pressure gradient, p_η
PP	= Pressure correction, p'
PPBLK	= Global pressure correction
PSCI	= Pressure gradient, P_ζ
PTA	= Wall boundary source term for the momentum equations
PW	= Wall value control parameter
RENL	= Reynolds number of the fluid
SIGE	= Turbulence model constant, = 1.3
SIGK	= Turbulence model constant, = 1.0
SINX(I)	= Wall boundary direction cosine
SINY(I)	= Wall boundary direction cosine
SINZ(I)	= Wall boundary direction cosine
SMNUM	= Small number, 1.E-30
SOC1	= Source term due to shear stress
SOC2	= Source term due to shear stress
SOC3	= Source term due to shear stress
SOLVEQ	= Subroutine for solving general transport equation
SP(I,J,K)	= Linear part of the source term
SPK(I,J,K)	= Secondary linear part of the source term
SU(I,J,K)	= Constant part of the source term
SUK(I,J,K)	= Secondary constant part of the source term
SX(I,J,K)	= Grid transformation coefficient, ζ_x
SY(I,J,K)	= Grid transformation coefficient, ζ_y
SYMOUT	= Subroutine for setting flow boundary conditions
SZ(I,J,K)	= Grid transformation coefficient, ζ_z
TAUN(I)	= Wall shear stress

TIMT	= Total time
TJO(I,J,K)	= Jacobian of metric transformation
TM(I,J,K)	= Temperature
TMO(I,J,K)	= TM at the previous time level
TMULT	= Wall shear stress
TRANF	= Subroutine for calculating the grid transformation coefficients
TXXE(I,J,K)	= Metric coefficient for east face diffusive flux
TXXW(I,J,K)	= Metric coefficient for west face diffusive flux
TXYN(I,J,K)	= Metric coefficient for north face diffusive flux
TXYS(I,J,K)	= Metric coefficient for south face diffusive flux
TXZT(I,J,K)	= Metric coefficient for top face diffusive flux
TXZB(I,J,K)	= Metric coefficient for bottom face diffusive flux
TYYN(I,J,K)	= Metric coefficient for north face diffusive flux
TYYS(I,J,K)	= Metric coefficient for south face diffusive flux
TYXE(I,J,K)	= Metric coefficient for east face diffusive flux
TYXW(I,J,K)	= Metric coefficient for west face diffusive flux
TYZT(I,J,K)	= Metric coefficient for top face diffusive flux
TYZB(I,J,K)	= Metric coefficient for bottom face diffusive flux
TZZT(I,J,K)	= Metric coefficient for top face diffusive flux
TZZB(I,J,K)	= Metric coefficient for bottom face diffusive flux
TZXE(I,J,K)	= Metric coefficient for east face diffusive flux
TZXW(I,J,K)	= Metric coefficient for west face diffusive flux
TZYN(I,J,K)	= Metric coefficient for north face diffusive flux
TZYS(I,J,K)	= Metric coefficient for south face diffusive flux
U(I,J,K)	= U-velocity
UO(I,J,K)	= U at the previous time level
UC	= Velocity at the center of a surface
UCXI	= U-velocity gradient, u_ξ

UEDA	= U-velocity gradient, u_η
UINC	= Velocity correction at outlet plane
USCI	= U-velocity gradient, u_ζ
UX	= U-velocity gradient, u_x
UY	= U-velocity gradient, u_y
UZ	= U-velocity gradient, u_z
V(I,J,K)	= V-velocity
VO(I,J,K)	= V at the previous time level
VISC	= Molecular viscosity, μ
VISE(I,J,K)	= Effective viscosity, μ_{eff}
VCXI	= V-velocity gradient, v_ξ
VEDA	= V-velocity gradient, v_η
VSCI	= V-velocity gradient, v_ζ
VX	= V-velocity gradient, v_x
VY	= V-velocity gradient, v_y
VZ	= V-velocity gradient, v_z
W(I,J,K)	= W-velocity
WO(I,J,K)	= W at the previous time level
WALLFN	= Subroutine for calculating the wall functions
WALVAL	= Subroutine for assigning wall values
WCXI	= W-velocity gradient, w_ξ
WEDA	= W-velocity gradient, w_η
WSCI	= W-velocity gradient, w_ζ
WX	= W-velocity gradient, w_x
WY	= W-velocity gradient, w_y
WZ	= W-velocity gradient, w_z
X(I,J,K)	= X-coordinate

Y(I,J,K)	= Y-coordinate
YN(I)	= Wall normal distance from the last grid
YN1(I)	= Wall grid volume size
YPLN(I)	= Nondimensionalized YN, $y^+ = u_\tau y / v$
Z(I,J,K)	= Z-coordinate

**APPENDIX C
PROGRAM LISTING**

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FORTRAN VII: LICENSED RESTRICTED RIGHTS AS STATED IN LICENSE *****, SEE DOCUMENTATION PACKAGE, 04-101499.
1 00000001 C PROGRAM CNS30
2 BY: Y. S. CHEN CN: 8/13/1985
3
4 C*****C CRLVLINEAR N-S CODE FOR 3-D INCOMPRESSIBLE FLOWS *****
5
6 0000061 C
7 COMMON
8 1/VAR/U(21,18,10),V(21,18,10),P(21,18,10),DK(21,18,10),
9 2/DE(21,18,10),ERRU,ERRM,ERRR,ERRW,
10 3/PF(21,18,10),W(21,18,10),TM(21,18,10)
11 1/PRCP/ VISE(21,18,10),CEN(21,18,10),VISC,DENIN,FLOWIN
12 1/PCCR/ CUC(21,18,10),DV(21,18,10),DW(21,18,10)
13 1/TUR/ SIGK,SIGE,CMU,C1,C2,CMU2,CMU3,MINUM,SMNUM,ANV1(800),
14 2/YN(800),SINY(800),SINX(800),SINZ(800),ANW1(300),
15 3/YPLN(800),TAUN(800),ZEC(800),JAC(300),KBC(800),ITY(800),
16 4/TALW(800),GEN(21,18,10),MC((21,18,10),IJL0(21,18,10),IITO
17 1/COEF/ AP(21,18,10),SU((21,18,10),SP(21,18,10),SU((21,18,10),
18 2/SPK((21,18,10),AE((21,18,10),AW((21,18,10),AN((21,18,10),
19 3/ASC(21,18,10),AT(21,18,10),A3((21,18,10),APC(21,18,10)
20 COMMON
21 1/TTRAN/ X((21,18,10),Y((21,18,10),Z((21,18,10),TJO(21,18,10),
22 2/CX((21,18,10),CY((21,18,10),CZ((21,18,10),
23 3/EX((21,18,10),EY((21,18,10),EZ((21,18,10),
24 3/SX((21,18,10),SY((21,18,10),SZ((21,18,10),
25 1/LIMT/ L,MLT,MT,L1,L2,M1,M2,L0,M0,ISWU,ISWP,ISWK,ISWE,
26 2/ALU,ALV,ALP,ALK,ALV,ALVIS,ALW,N1,N2,NC,ISWW,IG,NT,ALC,DTT
27 COMMON
28 1/TTRAN/TXX((21,18,10),TXXW((21,18,10),TYY((21,18,10),
29 2/TYY((21,18,10),TZZT((21,18,10),TZZB((21,18,10),
30 3/TYX((21,18,10),TYXW((21,18,10),TYZT((21,18,10),
31 4/TYZS((21,18,10),TYYS((21,18,10),TYYS((21,18,10),
32 5/TXZT((21,18,10),TXZB((21,18,10),TXZE((21,18,10),
33 6/TZWX((21,18,10),TZYN((21,18,10),TZWN((21,18,10),
34 1/UNSTOY/UC((21,18,10),VC((21,18,10),WQ((21,18,10),OKO((21,18,10),
35 2/CEO((21,18,10),DENO((21,18,10),TMO((21,18,10),
36 3/LOGICAL INSCU,INSOV,INSCP,INSCOK,INSCOE,INPRO,INSOV,INSOV
37 C*****C INPUT DATA GUIDE *****
38 C NLIMT : MAXIMUM NO. OF ITERATIONS LIMIT
39 C
40 C IG = 1 : LAMINAR
41 C 2 : TURBULENT (K-E MODEL)
42 C
43 C ISWP : NO. OF SWEEPS FOR SOLVING THE P' EQUATION (PP).
44 C
45 C ITT : TOTAL NO. OF TIME STEPS.
46 C
47 C ALU,ALV,ALW,ALP,ALK,ALE,ALVIS,ALC : UNDER-RELAXATION FACTORS
48 C
49 C RENL : REFERENCE REYNOLDS NUMBER.
50 C
51 C DTT : TIME STEP FOR UNSTEADY PROBLEMS.
52 C
53 C*****C INPUT DATA (PROBLEM CONTROL SETTING)
54 55 0000061 REAC(5,100) NLIMT,IG,ISWP,ITT
55 56 0000301 REAC(5,200) ALU,ALV,ALW,ALP,ALK,ALE,ALVIS,ALC
57 57 00006C1 REAC(5,200) RENL,DTT

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ORIGINAL PAGE IS
OF POOR QUALITY

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115 C006E81      DK0(I,J,K)=CK(I,J,K)
116 0007341      DE0(I,J,K)=DE(I,J,K)
117 0007501      DENC(I,J,K)=DEN(I,J,K)
118 0007CC1      50 CONTINUE
119 C*****          ****
120          C-----GET BOUNDARY CONTROL PARAMETERS
121          CALL DIRCOS
122          C----SET BOUNDARY TURBULENCE QUANTITIES TO ZERO
123          DO 121 I=1,L
124          DO 121 J=1,M
125          DO 121 K=1,N
126          IF(WCT(I,J,K).NE.0) GC TO 122
127          30 TO 121
128          122 DCK(I,J,K)=0.0
129          DCK(I,J,K)=0.0
130          U(I,J,K)=0.0
131          V(I,J,K)=0.0
132          W(I,J,K)=0.0
133          121 CONTINUE
134          C-----CALCULATE GRID TRANSFORMATION COEFFICIENTS
135          CALL TRANF
136          C-----TURBULENT VISCOSITY
137          0009C01      IF(INPRC) CALL NEWVIS
138          C-----CALCULATE INLET MASS FLOW RATE
139          FLOW=FC0
140          0009E01      I=1
141          0009E31      DO 45 J=2,M
142          0009FC1      DO 45 K=2,N
143          000A101      UC=(UC(I,J,K)+U(I,J-1,K)+U(I,J,K-1)+U(I,J-1,K-1))*0.25
144          000AC01      DENC=(DENC(I,J,K)+DENC(I,J-1,K)+DENC(I,J,K-1)+DENC(I,J-1,K-1))*0.25
145          000B701      P1=(X(I,J,K)+X(I,J-1,K)-X(I,J-1,K)-X(I,J-1,K-1))*0.5
146          000C201      P2=(Y(I,J,K)+Y(I,J-1,K)-Y(I,J-1,K)-Y(I,J-1,K-1))*0.5
147          000CD01      P3=(Z(I,J,K)+Z(I,J-1,K)-Z(I,J-1,K)-Z(I,J-1,K-1))*0.5
148          000D801      Q1=(X(I,J,K)*X(I,J-1,K)-X(I,J,K-1)*X(I,J-1,K-1))*0.5
149          000E301      Q2=(Y(I,J,K)*Y(I,J-1,K)-Y(I,J,K-1)*Y(I,J-1,K-1))*0.5
150          000EE01      Q3=(Z(I,J,K)*Z(I,J-1,K)-Z(I,J,K-1)*Z(I,J-1,K-1))*0.5
151          000F901      AREA=SQRT(P1+P2+P3*P3)*SQRT(Q1*Q1+Q2*Q2+Q3*Q3)
152          00102C1      FLOWIN=FLOWIN+DENC*AREA*UC
153          45 CONTINUE
154          0010741      153
155          154 ITG=1
156          0010321      C-----TRANSIENT PROCESS
157          0010821      2 CONTINUE
158          00100C1      CALL SYMCUT(3,1,2,L,2,M,2,N)
159          158
160          0010E41      C-----PROCEDURES START
161          0010E41      1 CONTINUE
162          0011401      CALL SYMOUT(1,1,2,LT,2,MT,2,NT)
163          0011841      IF(INSCU) CALL SOLVEC(1,ISW,ALU,SIGU,ERRU,U,UC)
164          0011C81      IF(INSCV) CALL SOLVEC(2,ISW,ALV,SIGU,ERRV,V,VC)
165          00120C1      IF(INSCW) CALL SOLVEC(3,ISW,ALW,SIGU,ERRW,W,WC)
166          0012501      IF(INSCD) CALL SOLVEC(4,ISW,ALW,SIGU,ERRW,TH,TH0)
167          0012941      IF(INSCK) CALL SOLVEC(5,ISW,ALK,SIGK,ERRK,CK,CK0)
168          0012D81      IF(INSCO) CALL SOLVEC(6,ISW,ALE,SIGE,ERE,DE,DE0)
169          00131C1      IF(INSPC) CALL SOLVEC(7,ISW,ALP,SIGU,ERRM,PP,PP)
170          170
171          00133C1      C-----CONVERGENCE CHECK
172          WRITE(6,30C) ITG,ERRU,ERRV,ERRW,ERRK,ERRM,ERRP
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SUBROUTINE DIRCCS
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1/ TUR/ SIGK, SIGE, CMU, C1, C2, CMU1, CMU2, E, CK, MINUM, SMNUM, ANV1(800),
2 YN(800)/ YN1(800)/ SINX(800), SINZ(800), ANW1(800),
3 YPLN(800)/ TAUN(800), IBC(800), JBC(800), KBC(800), ITTY(800),
4 TAUW(800), GEN(21,18,10), MC(21,18,10), IJL0(21,18,10), IT0,
5 1/ TRAN/ XC(21,18,10), Y(21,18,10), Z(21,18,10), TJO(21,18,10),
6 2 CX(21,18,10), CY(21,18,10), CZ(21,18,10),
7 3 EX(21,18,10), EY(21,18,10), EZ(21,18,10),
8 3 SX(21,18,10), SY(21,18,10), SZ(21,18,10),
9 1/LINT/ LM, LNT, L1, L2, M1, M2, LC, MD, ISWU, ISWP, ISWV, ISW,
10 2 ALU, ALV, ALP, ALK, ALE, ALVIS, ALWN, N1, N2, NO, ISHW, IG, NT, ALC, DTT
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C-----SET DOMAIN BLOCKAGE : MC(I,J,K)=2
C-----SCALAR EBLOCKAGE : MC(I,J,K)=1
C-----PRESSURE BLOCKAGE : MC(I,J,K)=0
10 10 I=1,LO
10 10 J=1,MO
10 10 K=1,NC
MC(I,J,K)=0
IF(J.EQ.1.OR.J.EQ.N) MC(I,J,K)=1
IF(I.GE.L1.AND.I.LE.L2.AND.J.GE.M1.AND.J.LE.M2.AND.
1 K.GE.N1.AND.K.LE.N2) MC(I,J,K)=1
IF(I.GT.L1.AND.I.LE.L2.AND.J.GT.M1.AND.J.LE.M2.AND.
1 K.GT.N1.AND.K.LE.N2) MC(I,J,K)=2
C-----ADD BULKAGES AS NEEDED HERE
10 CONTINUE
C-----CALCULATE BOUNDARY GRID SIZES AND ORIENTATIONS
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115 0014E01 Q2=Y(I,J,K)-Y(I,J1,K)
116 0015321 Q3=Z(I,J,K)-Z(I,J1,K)
117 0015841 AA=SQRT((Q1-P1)**2+(Q2-P2)**2+(Q3-P3)**2)
118 0016021 CC=1.0
119 0016051 B3=SQRT((Q1+Q2+Q2+Q3+Q3)
120 0016541 C0TP=(C83*83+CC*CC-AA*AA)/(2*BB*CC)
121 00169C1 YN(III)=BB*495(CC0TH)
122 0016C81 Q1=X(I,J,K)-X(I,J2,K)
123 00171A1 Q2=Y(I,J,K)-Y(I,J2,K)
124 00176C1 Q3=Z(I,J,K)-Z(I,J2,K)
125 00179E1 B3=SQRT((Q1+Q2+Q2+Q3+Q3)
126 0018041 AA=SQRT((Q1-P1)**2+(Q2-P2)**2+(Q3-P3)**2)
127 0018661 C0TP=(C82*59+CC*CC-AA*AA)/(2*BB*CC)
128 0018C81 YN1(III)=B3*AS(CC0TH)+YN(III)*0.5
129 0019031 IJL0((I,J,K)=21:
130 0019341 III=III+1
131 0019421 CONTINUE
132 0019421 I=F(WCC+1,J,K) .EQ. 0) GO TO 4
133 0019421
C-----EAST
134 00197C1 IBC(III)=I
135 0019901 JBC(III)=J
136 0019A41 KBC(III)=K
137 0019981 ITTY(III)=3
138 0019C31 J1=J+1
139 0019D61 J2=J-1
140 0019E41 K1=K+1
141 0019F21 K2=K-1
142 001A001 I=F(J+EC, M1) J2=J
143 001A1E1 I=F(J+EC, M2) J1=J
144 001A3C1 I=K .EC. N1) K2=K
145 001A5A1 I=F(K .EC. N2) K1=K
146 001A781 I1=I-1
147 001A851 I2=I-2
148 001A941 P1=(Y(I,J1,K1)-Y(I,J1,K2))*((Z(I,J1,K2)-Z(I,J2,K))-1
149 001B001 ((Z(I,J1,K1)-Z(I,J1,K2))*((Y(I,J1,K2)-Y(I,J2,K))-P2=((Z(I,J1,K1)-Z(I,J1,K2))*((X(I,J1,K2)-X(I,J2,K))-1
150 001B091 P2=((Z(I,J1,K1)-Z(I,J1,K2))*((X(I,J1,K2)-X(I,J2,K))-1
151 001B0C1 P3=((X(I,J1,K1)-X(I,J1,K2))*((Z(I,J1,K2)-Z(I,J2,K))-1
152 001B0C1 1 ((Y(I,J1,K1)-Y(I,J1,K2))*((X(I,J1,K2)-Y(I,J2,K))-PQ=SQRT((P1*P1+P2*P2+P3*P3).
153 001B481 P1=F1/AC
154 001B591 P2=F2/PC
155 001B921 P3=F3/PC
156 001EA41 R1=(C1-F1**2)
157 001E361 R2=(C1-P2**2)
158 001EC81 R3=(C1-P3**2)
159 001EE41 Q1=X(I,J,K)-X(I,J,K)
160 001F001 Q2=Y(I,J,K)-Y(I,J,K)
161 001F1C1 Q3=Z(I,J,K)-Z(I,J,K)
162 001F421 AA=SQRT((Q1-P1)**2+(Q2-P2)**2+(Q3-P3)**2)
163 001F661 CC=1.0
164 001F8A1 B3=SQRT((Q1+Q2+Q3+Q3)
165 001FDC1 C0TP=(C82*32+CC*CC-AA*AA)/(2*BB*CC)
166 00202E1 YN(III)=BB*495(CC0TH)
167 0020301
168 0020FE1
169 00210A1
170 0021561
171 0021981

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229 002F00I IJLC(I,J,K)=III
230 002F2CI III=III+1
231 002F3AI 5 CONTINUE
232 002F3AI IF(WC(I,J,K+1) .EQ. 0) GO TO 6
233 C-----TOP
234 002F72I I3C(III)=I
235 002F86I IBC(III)=J
236 002F9AI KBC(III)=K
237 002FAEI IITY(III)=5
238 002FBEI I1=I+1
239 002FCCI I2=I-1
240 002FDAT J1=J+1
241 002FE8I J2=J-1
242 002FF6I IF(I .EQ. L1) I2=I
243 003014I IF(I .EQ. L2) I1=I
244 003032I IF(J .EQ. M1) J2=J
245 003050I IF(J .EQ. M2) J1=J
246 00306EI K1=K-1
247 00307CI K2=K-2
248 00308AI P1=(Y(I1,J1,K)-Y(I1,J2,K))*(Z(I1,J2,K)-Z(I2,J,K))
249 0031C6I 1 (Z(I1,J1,K)-Z(I1,J2,K))*(Y(I1,J2,K)-Y(I2,J,K))
250 0031C6I P2=(Z(I1,J1,K)-Z(I1,J2,K))*(X(I1,J2,K)-X(I2,J,K))
251 003302I 1 (X(I1,J1,K)-X(I1,J2,K))*(Z(I1,J2,K)-Z(I2,J,K))
252 003302I P3=(X(I1,J1,K)-X(I1,J2,K))*(Y(I1,J2,K)-Y(I2,J,K))
253 00343EI 1 (Y(I1,J1,K)-Y(I1,J2,K))*(X(I1,J2,K)-X(I2,J,K))
254 00348AI PQ=SQRT(P1*P1+P2*P2+P3*P3)
255 00349CI P1=F1/PQ
256 0034AEI P2=P2/PQ
257 0034CDI P3=P3/PQ
258 0034DCI R1=1.-P1**2
259 0034F8I R2=-.P2**2
260 0034F8I R3=1.-P3**2
261 003514I SIN(X(III))=SQRT(R1)
262 00353AI SIN(Y(III))=SQRT(R2)
263 00355EI SIN(Z(III))=SQRT(R3)
264 003582I Q1=X(I,J,K)-X(I,J,K1)
265 0035D4I Q2=Y(I,J,K)-Y(I,J,K1)
266 003626I Q3=Z(I,J,K)-Z(I,J,K1)
267 003678I AA=SQRT((Q1-P1)*2+(Q2-P2)**2+(Q3-P3)**2)
268 0036F6I CC=1.0
269 003702I BB=SQRT(Q1*Q1+C2*Q2+Q3*Q3)
270 00374EI COTH=(BB+BB+CC+CC-AA+AA)/(2*BB+CC)
271 003790I YN(III)=96*ABS(COTH)
272 00378CI Q1=x(I,J,K)-X(I,J,K2)
273 00380EI Q2=y(I,J,K)-Y(I,J,K2)
274 003860I Q3=z(I,J,K)-Z(I,J,K2)
275 003882I 9B=SQRT(Q1*Q1+Q2*Q2+Q3*Q3)
276 0038FEI AA=SQRT((Q1-P1)*2+(Q2-P2)**2+(Q3-P3)**2)
277 00397AI COTH=(BB+BB+CC+CC-AA+AA)/(2*BB+CC)
278 0039BCI YN1(III)=(BB+AB5(COTH)+YN(III))*0.5
279 0039FCI IJLC(I,J,K)=III
280 003A28I III=III+1
281 003A36I 6 CONTINUE
282 003A36I IF(WC(I,J,K-1) .EQ. 0) GO TO 30
283 C-----BOTTOM
284 003A66I IBC(III)=I
285 003A82I JAC(III)=J
286 484 485
287 487 488
288 489 490

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286 003A96I
287 003AAA1
288 003ABA1
289 003AC8I
290 003AD6I
291 003AE4I
292 003AF2I
293 003B10I
294 003B2E1
295 003B4C1
296 003B6A1
297 003B78I
298 003B86I
299 003FBCI
300 003CC2I
301 003DFF1
302 003DFF1
303 003F3AI
304 003F86I
305 003F98I
306 003FAA1
307 003FAA1
308 003FB8I
309 003FD28I
310 003FF4I
311 004010I
312 004036I
313 004054I
314 00407E1
315 004000I
316 004122I
317 004174I
318 0041F2I
319 0041FE1
320 00424A1
321 00428C1
322 0042B3I
323 00430A1
324 00435C1
325 0043A5I
326 0043FA1
327 004476I
328 004498I
329 0044F8I
330 004524I
331 004532I
332 004580I
333 00458E1
334 004598I
335 0045C4I
336 0045CC1

KBC(III)=K
IITY(III)=6
I1=I+1
I2=I-1
J1=J+1
J2=J-1
IF(I . EC, L1) I2=I
IF(I . EC, L2) I1=I
IF(J . EC, M1) J2=J
IF(J . EC, M2) J1=J
K1=K+1
K2=K+2
P1=(Y(I1,J1,K)-Y(I1,J2,K))*((Z(I1,J2,K)-Z(I2,J,K))-((Z(I1,J1,K)-Z(I1,J2,K))*((Y(I1,J2,K)-Y(I2,J,K))-((Z(I1,J1,K)-Z(I1,J2,K))*((X(I1,J2,K)-X(I2,J,K))-((X(I1,J1,K)-X(I1,J2,K))*((Z(I1,J2,K)-Z(I2,J,K))-((Z(I1,J1,K)-X(I1,J2,K))*((Y(I1,J2,K)-Y(I2,J,K))-((Y(I1,J1,K)-Y(I1,J2,K))*((X(I1,J2,K)-X(I2,J,K))-((X(I1,J1,K)-P1*(P1+P2+P3*p3),PQ=SQRT((P1+P2+P3*p3),
P1=P1/P
P2=P2/P
P3=P3/P
P4=P1/P
P5=P2/P
P6=P3/P
P7=P4/P
P8=P5/P
P9=P6/P
P10=P7/P
P11=P8/P
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P331=P328/P
P332=P329/P
P333=P330/P
P334=P331/P
P335=P332/P
P336=P333/P
CONTINUE
IITG=III-1
WRITE(6,100) L0,M0,NO,IIT0
FORMAT(415)
RETLRN
END

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NO ERRORS:F70 ROSS-01.OC SUBROUTINE CIRCCS 02/21/86 09:47:58 TABLE SPACE: 6 KB
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 203 WORDS
 SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

ORIGINAL PAGE IS
OF POOR QUALITY

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58 000EF4I   AB(I,J,K)= -PTR*(P1*Q3-P3*Q1)
59   DUC(I,J,K)= PTR*(C1*R2-Q2*R1)
60   CVC(I,J,K)= -PTR*(P1*R2-P2*R1)
61   DWC(I,J,K)= PTR*(P1*Q2-P2*Q1)
62 00101C1   CONTINUE
63   CALL WALVAL(1.0,2,L,2,M,2,N,AE)
64   CALL WALVAL(1.0,2,L,2,M,2,N,AW)
65   CALL WALVAL(1.0,2,L,2,M,2,N,AN)
66   CALL WALVAL(1.0,2,L,2,M,2,N,AS)
67   CALL WALVAL(1.0,2,L,2,M,2,N,AT)
68   CALL WALVAL(1.0,2,L,2,M,2,N,AT)
69   CALL WALVAL(1.0,2,L,2,M,2,N,AB)
70   CALL WALVAL(1.0,2,L,2,M,2,N,DU)
71   CALL WALVAL(1.0,2,L,2,M,2,N,DV)
72   CALL WALVAL(1.0,2,L,2,M,2,N,QW)
73   CALL WALVAL(1.0,2,L,2,M,2,N,SU)
74   DO 80 I=1,L
75   DO 80 J=1,M
76   DO 80 K=1,N
77   C(X(I,J,K)= (AE(I,J,K)+AE(I,J+1,K)+AE(I,J+1,K+1)+AE(I,J+1,K+1))+
78   1 AE(I+1,J,K)+AE(I+1,J+1,K)+AE(I+1,J,K+1)+AE(I+1,J+1,K+1))*0.125
79   CYC(I,J,K)= (AN(I,J,K)+AW(I,J+1,K)+AW(I,J+1,K+1)+AW(I,J+1,K+1))*0.125
80   1 AW(I+1,J,K)+AW(I+1,J+1,K)+AW(I+1,J,K+1)+AW(I+1,J+1,K+1))*0.125
81   CZC(I,J,K)= (ANC(I,J,K)+AN(I,J+1,K)+AN(I,J,K+1)+AN(I,J+1,K+1))*0.125
82   1 ANC(I,J,K)+ANC(I,J+1,K)+AN(I,J,K+1)+AN(I,J+1,K+1))*0.125
83   EXC(I,J,K)= (ASC(I,J,K)+AS(I,J+1,K)+AS(I,J,K+1)+ASC(I,J+1,K+1))*0.125
84   1 AS(I,I+1,J,K)+AS(I,I+1,J+1,K)+AS(I,I+1,J,K+1)+AS(I,I+1,J+1,K+1))*0.125
85   EYC(I,J,K)= (AT(I,J,K)+AT(I,J+1,K)+AT(I,J,K+1)+AT(I,J+1,K+1))*0.125
86   1 AT(I,I+1,J,K)+AT(I,I+1,J+1,K)+AT(I,I+1,J,K+1)+AT(I,I+1,J+1,K+1))*0.125
87   EZC(I,J,K)= (ABC(I,J,K)+ABC(I,J+1,K)+ABC(I,J,K+1)+ABC(I,J+1,K+1))*0.125
88   1 ABC(I,I+1,J,K)+ABC(I,I+1,J+1,K)+ABC(I,I+1,J,K+1)+ABC(I,I+1,J+1,K+1))*0.125
89   SXC(I,J,K)= (DCU(I,J,K)+DU(I,J+1,K)+DU(I,J,K+1)+DU(I,J+1,K+1))*0.125
90   1 DU(I,I+1,J,K)+DU(I,I+1,J+1,K)+DU(I,I+1,J,K+1)+DU(I,I+1,J+1,K+1))*0.125
91   SYC(I,J,K)= (DCV(I,J,K)+DV(I,J+1,K)+DV(I,J,K+1)+DV(I,J+1,K+1))*0.125
92   1 DCV(I,I+1,J,K)+DCV(I,I+1,J+1,K)+DCV(I,I+1,J,K+1)+DCV(I,I+1,J+1,K+1))*0.125
93   SZC(I,J,K)= (DW(I,J,K)+DW(I,J+1,K)+DW(I,J,K+1)+DW(I,J+1,K+1))*0.125
94   1 DW(I,I+1,J,K)+DW(I,I+1,J+1,K)+DW(I,I+1,J,K+1)+DW(I,I+1,J+1,K+1))*0.125
95   TJO(I,J,K)= (SU(I,J,K)+SU(I,J+1,K)+SU(I,J,K+1)+SU(I,J+1,K+1))*0.125
96   1 SU(I,I+1,J,K)+SU(I,I+1,J+1,K)+SU(I,I+1,J,K+1)+SU(I,I+1,J+1,K+1))*0.125
97   80  CONTINUE
98   9022380I   DO 160 I=2,LT
99   902394I   DO 200 J=1,M
100  9023A8I   DO 200 K=1,N
101  90239CI   DUC(I,J,K)=0.0
102  902388I   CONTINUE
103  902380I   DO 160 I=2,LT
104  902444I   DO 150 K=2,NT
105  902458I   DO 150 K=2,NT
106  90246C1   CXE= (CX(I+1,J,K)+CX(I,J,K))*0.5
107  902524I   CXW= (CX(I-1,J,K)+CX(I,J,K))*0.5
108  9025280I   CXN= (CX(I,J+1,K)+CX(I,J,K))*0.5
109  902529C1   CXS= (CX(I,J,K)+CX(I,J-1,K))*0.5
110  902636I   CXT= (CX(I,J,K+1)+CX(I,J,K))*0.5
111  902690I   CXB= (CX(I,J,K)+CX(I,J,K-1))*0.5
112  9026E5I   CYE= (CY(I+1,J,K)+CY(I,J,K))*0.5
113  902748I   CYW= (CY(I,J+1,K)+CY(I,J,K))*0.5
114  90272A4I   CYS= (CY(I,J,K)+CY(I,J-1,K))*0.5
115  90273A4I

```

115	002800I	CYT = (CY (I,J,K+1) + CY (I,J,K)) * 0.5	654
116	002854I	CYB = (CY (I,J,K) + CY (I,J,K-1)) * 0.5	655
117	002834I	CZE = (CZ (I,J+1,J,K) + CZ (I,J,K)) * 0.5	656
118	002910I	CZN = (CZ (I,J+1,J,K) + CZ (I,J,K)) * 0.5	657
119	002960I	CZLW = (CZ (I,J+1,J,K) + CZ (I,J,K)) * 0.5	658
120	0029C8I	CZS = (CZ (I,J,K) + CZ (I,J-1,K)) * 0.5	659
121	002A24I	CZT = (CZ (I,J,K+1) + CZ (I,J,K)) * 0.5	660
122	002A15I	CZB = (CZ (I,J,K) + CZ (I,J,K-1)) * 0.5	661
123	002A08I	EXE = (EX (I+1,J,K) * EX (I,J,K)) * 0.5	662
124	002834I	EXW = (EX (I,J,K) + EX (I,J,K-1)) * 0.5	663
125	002590I	EXN = (EX (I,J+1,K) + EX (I,J,K)) * 0.5	664
126	002835I	EXS = (EX (I,J,K) + EX (I,J-1,K)) * 0.5	665
127	002C48I	EXT = (EX (I,J,K+1) + EX (I,J,K)) * 0.5	666
128	002C44I	EXB = (EX (I,J,K) + EX (I,J,K-1)) * 0.5	667
129	0022FCI	EYE = (EY (I,J,K) + EY (I,J,K-1)) * 0.5	668
130	002D08I	EYW = (EY (I,J,K) + EY (I,J-1,K)) * 0.5	669
131	002084I	EYN = (EY (I,J+1,K) + EY (I,J,K)) * 0.5	670
132	002E10I	EYS = (EY (I,J,K) + EY (I,J-1,K)) * 0.5	671
133	002E66I	EYT = (EY (I,J,K+1) + EY (I,J,K)) * 0.5	672
134	002EC6I	EYB = (EY (I,J,K) + EY (I,J,K-1)) * 0.5	673
135	002F20I	EZE = (EZ (I+1,J,K) + EZ (I,J,K)) * 0.5	674
136	002F7CI	EZW = (EZ (I,J,K) + EZ (I-1,J,K)) * 0.5	675
137	002FD8I	EZN = (EZ (I,J+1,K) + EZ (I,J,K)) * 0.5	676
138	003034I	EZS = (EZ (I,J,K) + EZ (I,J-1,K)) * 0.5	677
139	003090I	EZT = (EZ (I,J,K+1) + EZ (I,J,K)) * 0.5	678
140	0030EAI	EZB = (EZ (I,J,K) + EZ (I,J,K-1)) * 0.5	679
141	003144I	SX = (SX (I,J,K) + SX (I,J,K)) * 0.5	680
142	0031A0I	SXW = (SX (I,J,K) + SX (I,J,K)) * 0.5	681
143	0031FCI	SXN = (SX (I,J+1,K) + SX (I,J,K)) * 0.5	682
144	003258I	SXZ = (SX (I,J,K) + SX (I,J,K+1)) * 0.5	683
145	003234I	SXT = (SX (I,J,K) + SX (I,J,K-1)) * 0.5	684
146	003305I	SX5 = (SX (I,J,K) + SX (I,J,K)) * 0.5	685
147	003368I	SYE = (SY (I+1,J,K) + SY (I,J,K)) * 0.5	686
148	00333C4I	SYW = (SY (I,J,K) + SY (I-1,J,K)) * 0.5	687
149	003420I	SYN = (SY (I,J+1,K) + SY (I,J,K)) * 0.5	688
150	00347C1	SYS = (SY (I,J,K) + SY (I,J-1,K)) * 0.5	689
151	003408I	SYT = (SY (I,J,K+1) + SY (I,J,K)) * 0.5	690
152	003532I	SYB = (SY (I,J,K) + SY (I,J,K-1)) * 0.5	691
153	00358C1	SZE = (SZ (I+1,J,K) + SZ (I,J,K)) * 0.5	692
154	003568I	SZT = (SZ (I,J,K) + SZ (I,J,K-1)) * 0.5	693
155	003644I	SZN = (SZ (I,J+1,K) + SZ (I,J,K)) * 0.5	694
156	0036A0I	SZS = (SZ (I,J,K) + SZ (I,J,K-1)) * 0.5	695
157	0036FCI	SZT = (SZ (I,J,K) + SZ (I,J,K-1)) * 0.5	696
158	003756I	TZT = (TZT (I,J,K) = SXT * SX (I,J,K) + SYT * SY (I,J,K)) * 0.5	697
159	003730I	TXE (I,J,K) = CX (I,J,K) + CY * CY (I,J,K) + CZE * CZ (I,J,K)	698
160	00385E1	TXW (I,J,K) = CX (I,J,K) + CY * CY (I,J,K) + CZN * CZ (I,J,K)	699
161	00390C1	TYN (I,J,K) = EXN * EX (I,J,K) + EYN * EY (I,J,K) + EZN * EZ (I,J,K)	700
162	00398AI	TYYS (I,J,K) = EXS * EX (I,J,K) + EYS * EY (I,J,K) + EZS * EZ (I,J,K)	701
163	003A68I	TZT (I,J,K) = TZT (I,J,K) + SXT * SX (I,J,K) + SYT * SY (I,J,K) + SZT * SZ (I,J,K)	702
164	003316I	TZB (I,J,K) = SXB * SX (I,J,K) + SYB * SY (I,J,K) + SZB * SZ (I,J,K)	703
165	0038C4I	TYXE (I,J,K) = EXE * CX (I,J,K) + EYE * CY (I,J,K) + EZE * CZ (I,J,K) * 0.25	704
166	003C78I	TYXW (I,J,K) = EXW * CX (I,J,K) + EYW * CY (I,J,K) + EZW * CZ (I,J,K) * 0.25	705
167	003D2C1	TYZT (I,J,K) = EXT * SX (I,J,K) + EYT * SY (I,J,K) + EZT * SZ (I,J,K) * 0.25	706
168	0030E0I	TYZB (I,J,K) = EXB * SX (I,J,K) + EYB * SY (I,J,K) + EZB * SZ (I,J,K) * 0.25	707
169	003E94I	TXYN (I,J,K) = CXN * EX (I,J,K) + CZN * EZ (I,J,K) * 0.25	708
170	003F48I	TXYS (I,J,K) = CXS * EX (I,J,K) + CY * EY (I,J,K) + CZS * EZ (I,J,K) * 0.25	709
171	003FFCI	TXZT (I,J,K) = CXT * SX (I,J,K) + CYT * SY (I,J,K) + CZT * SZ (I,J,K) * 0.25	710

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172 004080I TXZB(I,J,K)=(CXB+SX(I,J,K)+CYB*SY(I,J,K)+CZB*SZ(I,J,K))*0.25
173 004166I TZXE(I,J,K)=(SXE*CX(I,J,K)+SYE*CY(I,J,K)+SZE*CZ(I,J,K))*0.25
174 004218I TZXB(I,J,K)=(SXA*CX(I,J,K)+SYW*CY(I,J,K)+SZW*CZ(I,J,K))*0.25
175 0042CC1 TZYN(I,J,K)=(SXY*EX(I,J,K)+SYN*EY(I,J,K)+SZN*EZ(I,J,K))*0.25
176 004380I TZYS(I,J,K)=(SXS*EX(I,J,K)+SYS*EY(I,J,K)+Szs*EZ(I,J,K))*0.25
177 004434I CONTINUE
178 00447C1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TXXE)
179 0044D04I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TXXW)
180 00452C1 CALL WALVAL(1.C,2,LT,2,MT,2,NT,TYYN)
181 004584I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TYY)
182 00450C1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TZT)
183 004634I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TZB)
184 00468C1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TYXE)
185 0046E4I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TYXW)
186 00473C1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TYZT)
187 004794I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TYZB)
188 0047EC1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TXYN)
189 004844I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TXYS)
190 00489C1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TXZT)
191 0048F4I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TXZB)
192 00494C1 CALL WALVAL(1.C,2,LT,2,MT,2,NT,TZXE)
193 0049A4I CALL WALVAL(1.0,2,LT,2,MT,2,NT,TZWX)
194 0049FC1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TZYN)
195 004A5C1 CALL WALVAL(1.0,2,LT,2,MT,2,NT,TZYS)
196 004AAC1 RETURN
197 004A94I END

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NO ERRORS:F70  R05-01.0C  SUBROUTINE  TRANF  C2/21/86  09:50:44  TABLE SPACE: 10 KB
STATEMENT BUFFER: 20 LINES/1321 BYTES  STACK SPACE: 199 WORDS
SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

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1 0000001          SUBROUTINE INIT
2 0000041          COMMON
3          1/VAR/U(21,18,10),V(21,18,10),P(21,18,10),DX(21,18,10),
4          2/DE(21,18,10),ERRU,ERRV,ERRM,ERRK,ERRR,ERRW,
5          3/PF(21,18,10),W(21,18,10),TM(21,18,10)
6          1/PRCP/VISE(21,18,10),CEN(21,18,10),VISC,DENIN,FLWIN
7          1/PCCR/DU(21,18,10),DV(21,18,10),DW(21,18,10)
8          COMMON
9          1/LIMT/LM,LT,MT,L1,L2,M1,M2,LC,MD,ISWU,ISWV,ISWV,ISWU,
10         2/ALU,ALV,ALD,ALK,ALV,ALVIS,ALW,N1,N2,NC,ISWV,IG,NT,ALC,DTT
11         C-----INITIALIZE VARIABLES
12         20 10 I=1,LC
13         20 10 J=1,MC
14         20 10 K=1,ND
15         20 10 C(1,J,K)=0.0
16         20 10 V(1,J,K)=0.0
17         20 10 W(1,J,K)=0.0
18         20 10 P(1,J,K)=0.0
19         20 10 R(1,J,K)=0.0
20         20 10 C(1,J,K)=C(0
21         20 10 C(1,J,K)=C(0
22         20 10 CEN(1,J,K)=DENIN
23         20 10 VISE(1,J,K)=VISC
24         20 10 CONTINUE
25         20 10 RETURN
26         20 10 END
737          738
738          1/VAR/U(21,18,10),V(21,18,10),P(21,18,10),DX(21,18,10),
739          2/DE(21,18,10),ERRU,ERRV,ERRM,ERRK,ERRR,ERRW,
740          3/PF(21,18,10),W(21,18,10),TM(21,18,10)
741          1/PRCP/VISE(21,18,10),CEN(21,18,10),VISC,DENIN,FLWIN
742          1/PCCR/DU(21,18,10),DV(21,18,10),DW(21,18,10)
743          1/LIMT/LM,LT,MT,L1,L2,M1,M2,LC,MD,ISWU,ISWV,ISWV,ISWU,
744          2/ALU,ALV,ALD,ALK,ALV,ALVIS,ALW,N1,N2,NC,ISWV,IG,NT,ALC,DTT
745          746
746          747
747          1/VAR/U(21,18,10),V(21,18,10),P(21,18,10),DX(21,18,10),
748          2/DE(21,18,10),ERRU,ERRV,ERRM,ERRK,ERRR,ERRW,
749          3/PF(21,18,10),W(21,18,10),TM(21,18,10)
750          1/PRCP/VISE(21,18,10),CEN(21,18,10),VISC,DENIN,FLWIN
751          1/PCCR/DU(21,18,10),DV(21,18,10),DW(21,18,10)
752          1/LIMT/LM,LT,MT,L1,L2,M1,M2,LC,MD,ISWU,ISWV,ISWV,ISWU,
753          2/ALU,ALV,ALD,ALK,ALV,ALVIS,ALW,N1,N2,NC,ISWV,IG,NT,ALC,DTT
754          755
755          756
756          757
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762          15

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NO. ERRORS: 070 R25=01.0C SUBROUTINE INIT 02/21/36 09:51:07 TABLE SPACE: 3 KB
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 126 WORDS
 SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

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1 0000001
2 0000041
3 1/VAR/U(21,18,10),V(21,18,10),P(21,18,10),DK(21,18,10),,
4 2 DE(21,18,10),ERRU,ERRM,ERRK,ERRR,ERRW,
5 3 PP(21,18,10),W(21,18,10),TM(21,18,10)
6 1/PRCP/ VISE(21,18,10),CEN(21,18,10),VIS,C,DENIN,FLWIN
7 1/PCQR/ CU(21,18,10),DV(21,18,10),DW(21,18,10)
8 1/TUR/ SIGK,SIGE,CMU,C1,C2,CMU1,CMU2,E,CK,HINUM,SMNUM,ANV1(800),
9 2 YN(800),YN1(800),SINY(800),SINZ(800),ANW1(800),
10 3 YPLN(SC0),T4UN(800),IEC(600),JSC(800),KBC(800),ITY(800),
11 4 TALW(800),GEN(21,18,10),MC(21,18,10),IJLO(21,18,10),IITO
12 COMMON
13 1/LINT/ L,MLT,MT,L1,L2,M1,M2,LO,MO,ISWU,ISWV,ISWK,ISWE,
14 2 ALU,ALV,ALP,ALK,AL,E,ALVTS,ALW,N,N1,N2,NO,ISWW,IG,NT,ALC,DTT
15 C-----EVALUATE TURBULENT VISCOSITY
16 CC 10 I=1,AL
17 CC 10 J=1,N
18 CC 10 K=1,N
19 IFICK(I,J,K) .LE. SMNUM DK(I,J,K)=SMNUM
20 IFICE(I,J,K) .LE. SMNUM DE(I,J,K)=SMNUM
21 IFEC(I,J,K) .LE. SMNUM GO TO 12
22 000134I TURVISDEF(I,J,K)*MU*G(I,J,K)*2/DE(I,J,K)+VISC
23 0001C2I GO TO 14
24 0001C8I 12 TURVIS=VISC
25 0001D4I 14 CONTINUE
26 VISE(I,J,K)=VISE(I,J,K)+ALVIS*(TURVIS-VISE(I,J,K))
27 000252I 10 CONTINUE
28 00029AI RETURN
29 0002A0I END

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NO ERRORS: F7D ROS-01.GC SUBROUTINE NEWVIS 02/21/86 C9:51:23 TABLE SPACE: 5 KB
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 203 WORDS
 SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

ORIGINAL PAGE IS
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CONT=GAN*TZYN(I,J,K)+GAT*TYZ(I,J,K)
DON8=-GAN*TZYN(I,J,K)-GAB*TYB(I,J,K)
DOST=-GAS*TZYS(I,J,K)-GAT*TZT(I,J,K)
DOS8=GAS*TZYS(I,J,K)+GAB*TYS(I,J,K)
SUC(I,J,K)=CPO*(I,J,K)+GONE*(I,J+1,K)+GOS8*F(I+1,J-1,K)+*
C0N8*=F(I-1,J+1,K)+GOS8*F(I-1,J-1,K)+*
D0ET*=F(I+1,J,K+1)+G0E8*F(I+1,J,K-1)+*
D0WT*=F(I-1,J,K+1)+D0WB*F(I-1,J,K-1)+*
D0NT*=F(I,J+1,K-1)+D0NE*F(I,J+1,K-1)+*
D0ST*=F(I,J,K-1)+D0NS8*F(I,J-1,K-1)
D0V(I,J,K)=CPO*V(I,J,K)+CONE*V(I+1,J+1,K)+D0SE*V(I+1,J-1,K)+*
D0N8*V(I-1,J+1,K)+D0EW*V(I-1,J-1,K)+*
D0ET*V(I+1,J,K+1)+D0ES*V(I+1,J,K-1)+*
D0WT*V(I-1,J,K+1)+D0WB*V(I-1,J,K-1)+*
D0NT*V(I,J,K+1)+D0NE*V(I+1,J+1,K)+D0SE*V(I-1,J-1,K)+*
D0C(I,J,K)=CPO*W(I,J,K)+CONE*W(I+1,J+1,K)+D0SE*W(I+1,J-1,K)+*
D0N8*W(I-1,J+1,K)+D0SW*W(I-1,J-1,K)+*
D0ET*W(I+1,J,K+1)+D0ES*W(I+1,J,K-1)+*
D0WT*W(I-1,J,K+1)+D0WB*W(I-1,J,K-1)+*
D0NT*W(I,J,K+1)+D0NE*W(I+1,J+1,K)+D0SE*W(I-1,J-1,K)+*
D0C(I,J,K)=CPO
PP(I,J,K)=CPO*3K(I,J,K)+D0NE*OK(I+1,J+1,K)+D0SE*OK(I+1,J-1,K)+*
D0N8*OK(I-1,J+1,K)+D0SW*OK(I-1,J-1,K)+*
D0ET*OK(I,J+1,K+1)+D0EB*OK(I+1,J,K-1)+*
D0WT*OK(I-1,J,K+1)+D0WB*OK(I-1,J,K-1)+*
D0NT*OK(I,J+1,K+1)+D0NS*OK(I,J+1,K-1)+*
D0ST*OK(I,J-1,K+1)+D0SE*OK(I,J-1,K-1)
SUK(I,J,K)=CPO
SPK(I,J,K)=CPO-APC(I,J,K)
1 SPK(I,J,K)=UE-UP
2 UCX1=UE-UP
3 UEDA=UN-US
4 USC1=UT-US
5 VCX1=VE-VW
6 VEDA=VN-VS
7 VSCI=VT-VS
8 WCX1=WE-WW
9 WEDA=WN-WS
10 WSCI=WT-WB
11 UX=UX1*CX(I,J,K)*UEDA*EX(I,J,K)+USCI*SX(I,J,K)
12 UY=UCX1*CY(I,J,K)*UEDA*SY(I,J,K)+USCI*SY(I,J,K)
13 UZ=UCX1*CZ(I,J,K)*UEDA*EZ(I,J,K)+USCI*SZ(I,J,K)
14 VX=VCX1*CX(I,J,K)*VEDA*EX(I,J,K)+VSCI*SX(I,J,K)
15 VY=VCX1*CY(I,J,K)*VEDA*EX(I,J,K)+VSCI*SY(I,J,K)
16 UX=UX1*CZ(I,J,K)*VEDA*EZ(I,J,K)+VSCI*SZ(I,J,K)
17 UX=UCX1*CX(I,J,K)*WEDA*EX(I,J,K)+WSCI*SX(I,J,K)
18 MY=WCX1*CY(I,J,K)*WEDA*SY(I,J,K)+WSCI*SY(I,J,K)
19 WZ=WCX1*CZ(I,J,K)*WEDA*EZ(I,J,K)+WSCI*SZ(I,J,K)
20 GEN(I,J,K)=VISE(I,J,K)*((UX*VX)*2+(WZ*WY)*2+(VZ*WY)*2+(WZ*UZ)**2+
1 2*(UX*UX+VY*VY+(WZ*WZ)))
2 CONTINUE
3 -----CALCULATE SOURCE TERMS
4 GO TO 112/3/4/5/6/7), IE
5 -----U-, V-, W-SOURCES
6 CONTINUE

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ORIGINAL PAGE IS
OF POOR QUALITY

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00 15 I=IS,IT          110
00 15 J=JS,JT          111
00 15 K=KS,KT          964
00 15 GAE=0.5*(F1(I+1,J,K)+F1(I,J,K)) 965
00 15 GAW=0.5*(F1(I-1,J,K)+F1(I,J,K)) 966
00 15 GAN=0.5*(F1(I,J+1,K)+F1(I,J,K)) 967
00 15 GAS=0.5*(F1(I,J-1,K)+F1(I,J,K)) 968
00 15 GAT=0.5*(F1(I,J,K+1)+F1(I,J,K)) 969
00 15 GAB=0.5*(F1(I,J,K-1)+F1(I,J,K)) 970
00 15 GBCX1=(P(I+1,J+1,K)+P(I+1,J,K+1)+P(I+1,J+1,K+1))- 971
00 15 P(I,J,K)-P(I,J+1,K)-P(I,J,K+1)-P(I,J+1,K+1)*0.25 972
00 15 1 PECA=(P(I,J+1,K)+P(I+1,J+1,K)+P(I,J+1,K+1)+P(I,J+1,K+1))- 973
00 15 PSCI=(P(I,J,K+1)-P(I+1,J,K)-P(I,J+1,K+1)-P(I,J+1,K+1)*0.25 974
00 15 1 P(I,J,K)-P(I+1,J,K)-P(I,J+1,K)-P(I+1,J+1,K+1)- 975
00 15 1 P(I,J,K)-P(I+1,J,K)-P(I,J+1,K)-P(I+1,J+1,K+1)- 976
00 15 1 UCZ=U(I+1,J,K)-U(I,J,K)-U(I,J,K) 977
00 15 UCW=U(I,J,K)-U(I-1,J,K) 978
00 15 UCN=(U(I+1,J+1,K)+U(I+1,J,K)-U(I-1,J,K)-U(I-1,J,K))*0.25 979
00 15 UCT=(U(I+1,J,K)+U(I+1,J-1,K)-U(I-1,J,K)-U(I-1,J,K))*0.25 980
00 15 UCB=(U(I+1,J,K+1)+U(I+1,J,K)-U(I-1,J,K)-U(I-1,J,K))*0.25 981
00 15 UCB3=(U(I+1,J,K)+U(I+1,J-1,K)-U(I-1,J,K)-U(I-1,J,K))*0.25 982
00 15 UEE=(U(I+1,J+1,K)+U(I+1,J,K)-U(I-1,J,K)-U(I-1,J,K))*0.25 983
00 15 UEW=(U(I,J+1,K)+U(I,J,K)-U(I,J,K)-U(I,J-1,K))*0.25 984
00 15 UEN=U(I,J+1,K)-U(I,J,K)-U(I,J,K) 985
00 15 UET=(U(I,J+1,K+1)+U(I,J,K+1)-U(I,J-1,K)-U(I,J-1,K))*0.25 986
00 15 UEB=(U(I,J+1,K)+U(I,J+1,K-1)-U(I,J-1,K)-U(I,J-1,K))*0.25 987
00 15 USE=(U(I+1,J,K+1)+U(I+1,J,K)-U(I-1,J,K-1)-U(I-1,J,K-1))*0.25 988
00 15 USE3=(U(I+1,J,K+1)+U(I+1,J,K)-U(I-1,J,K-1)-U(I-1,J,K-1))*0.25 989
00 15 USE=(U(I+1,J,K+1)+U(I+1,J,K)-U(I-1,J,K-1)-U(I-1,J,K-1))*0.25 990
00 15 USW=(U(I+1,J,K+1)+U(I+1,J,K)-U(I-1,J,K-1)-U(I-1,J,K-1))*0.25 991
00 15 USW3=(U(I+1,J,K+1)+U(I+1,J,K)-U(I-1,J,K-1)-U(I-1,J,K-1))*0.25 992
00 15 USS=(U(I,J,K+1)+U(I,J,K)-U(I,J-1,K-1)-U(I,J-1,K-1))*0.25 993
00 15 UST=U(I,J,K+1)-U(I,J,K) 994
00 15 USB=U(I,J,K)-U(I,J,K-1) 995
00 15 VCE=V(I+1,J,K)-V(I,J,K) 996
00 15 VCN=(V(I+1,J+1,K)+V(I+1,J,K)+V(I+1,J-1,K)-V(I-1,J,K))*0.25 997
00 15 VCS=(V(I+1,J,K)+V(I+1,J-1,K)-V(I-1,J,K)-V(I-1,J-1,K))*0.25 998
00 15 VCT=(V(I+1,J,K+1)+V(I+1,J,K)-V(I-1,J,K+1)-V(I-1,J,K+1))*0.25 999
00 15 VCB=(V(I+1,J,K+1)+V(I+1,J,K)-V(I-1,J,K-1)-V(I-1,J,K-1))*0.25 1000
00 15 VEE=(V(I+1,J,K+1)+V(I+1,J,K)-V(I-1,J,K)-V(I-1,J,K))*0.25 1001
00 15 VEW=(V(I,J+1,K)+V(I,J,K)-V(I,J-1,K)-V(I,J-1,K))*0.25 1002
00 15 VEN=V(I,J+1,K)-V(I,J,K) 1003
00 15 VES=V(I,J,K)-V(I,J-1,K) 1004
00 15 VET=(V(I,J+1,K+1)+V(I,J+1,K)-V(I,J-1,K)-V(I,J-1,K))*0.25 1005
00 15 VEB=(V(I,J+1,K)+V(I,J+1,K-1)-V(I,J-1,K)-V(I,J-1,K-1))*0.25 1006
00 15 VSE=(V(I,J+1,K+1)+V(I,J+1,K)-V(I,J-1,K)-V(I,J-1,K))*0.25 1007
00 15 VSW=(V(I,J+1,K+1)+V(I,J+1,K)-V(I,J-1,K)-V(I,J-1,K))*0.25 1008
00 15 VSN=(V(I,J+1,K+1)+V(I,J,K)-V(I,J-1,K)-V(I,J-1,K))*0.25 1009
00 15 VSS=(V(I,J+1,K+1)+V(I,J,K)-V(I,J-1,K)-V(I,J-1,K))*0.25 1010
00 15 VST=(V(I,J,K+1)-V(I,J,K)) 1011
00 15 VSB=(V(I,J,K)-V(I,J-1,K)-V(I,J-1,K)) 1012
00 15 VSE=V(I,J+1,K)-V(I,J,K) 1013
00 15 VCE=V(I,J+1,K)-V(I,J,K) 1014
00 15 VCW=V(I,J,K)-V(I-1,J,K) 1015
00 15 VCN=(V(I+1,J+1,K)+V(I+1,J,K)-V(I-1,J+1,K)-V(I-1,J+1,K))*0.25 1016
00 15 VCS=(V(I+1,J,K)+V(I+1,J-1,K)-V(I-1,J,K)-V(I-1,J-1,K))*0.25 1017
00 15 VCT=(V(I+1,J,K+1)+V(I+1,J,K)-V(I-1,J,K+1)-V(I-1,J,K+1))*0.25 1018
00 15 VCB=(V(I+1,J,K+1)+V(I+1,J,K)-V(I-1,J,K-1)-V(I-1,J,K-1))*0.25 1019

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229 0086D2I   WEE=(W(C(I+1,J+1,K)+W(C(I,J+1,K)-W(I+1,J-1,K)-W(I,J-1,K))*0.25
230 008C8EI   WEE=(W(C(I,J+1,K)+W(C(I-1,J+1,K)-W(I,J-1,K))*0.25
231 008D4AI   WEN=W(C(I,J+1,K)-W(C(I,J,K)
232 008DA0I   WES=W(C(I,J,K)-W(C(I,J-1,K)
233 008DF6I   WET=(W(C(I,J+1,K+1)+W(C(I,J+1,K)-W(I,J-1,K-1)*0.25
234 008EAEI   WEB=(W(C(I,J+1,K)+W(C(I,J+1,K-1)-W(I,J-1,K)-W(I,J-1,K-1))*0.25
235 008F66I   WSE=(W(C(I+1,J,K+1)+W(C(I,J,K+1)+W(C(I,J,K+1)-W(I,J,K-1))*0.25
236 00901AI   WSW=(W(C(I,J,K+1)+W(C(I,J,K+1)-W(I,J,K-1)-W(I,J,K-1))*0.25
237 0090CEI   WSN=(W(C(I,J+1,K+1)+W(C(I,J,K+1)-W(I,J+1,K-1)-W(I,J,K-1))*0.25
238 009182I   WSS=(W(C(I,J,K+1)+W(C(I,J,K+1)-W(I,J,K-1)-W(I,J,K-1))*0.25
239 009236I   WST=W(C(I,J,K+1)-W(C(I,J,K-1)
240 00928AI   WS3=W(C(I,J,K)-W(C(I,J,K-1)
241 0092CEI   WSU(C(I,J,K)=SU(C(I,J,K)-PCXI*CX(C(I,J,K)-PEDA*EX(C(I,J,K)-PSCI*SX(C(I,J,K)
242 009384I   OVC(I,J,K)=OV(C(I,J,K)-PCXI*CY(C(I,J,K)-PEDA*EY(C(I,J,K)-PSCI*SX(C(I,J,K)
243 00948AI   OW(C(I,J,K)=DW(C(I,J,K)-PCXI*CZ(C(I,J,K)-PSCI*SY(C(I,J,K)
244 009560I   CXE=(CX(C(I+1,J,K)+CX(C(I,J,K))*0.5
245 00953C1   CXW=(CX(C(I-1,J,K)+CX(C(I,J,K))*0.5
246 009618I   CXN=(CX(C(I,J+1,K)+CX(C(I,J,K))*0.5
247 009674I   CXS=(CX(C(I,J,K)+CX(C(I,J-1,K))*0.5
248 009600I   CXT=(CX(C(I,J,K+1)+CX(C(I,J,K))*0.5
249 00972AI   CXB=(CX(C(I,J,K)+CX(C(I,J,K-1))*0.5
250 009784I   EXE=(EX(C(I+1,J,K)+EX(C(I,J,K))*0.5
251 0097E0I   EXW=(EX(C(I,J,K)+EX(C(I,J,K-1))*0.5
252 00983C1   EXN=(EX(C(I,J+1,K)+EX(C(I,J,K))*0.5
253 009898I   EXS=(EX(C(I,J,K)+EX(C(I,J-1,K))*0.5
254 0098F6I   EXT=(EX(C(I,J,K+1)+EX(C(I,J,K))*0.5
255 00994E1   EXB=(EX(C(I,J,K)+EX(C(I,J,K-1))*0.5
256 0099A8I   SXE=(SX(C(I+1,J,K)+SX(C(I,J,K))*0.5
257 009A04I   SXW=(SX(C(I,J,K)+SX(C(I,J,K-1))*0.5
258 009A60I   SXN=(SX(C(I,J+1,K)+SX(C(I,J,K))*0.5
259 009ABC1   SXS=(SX(C(I,J,K)+SX(C(I,J-1,K))*0.5
260 009B18I   SXT=(SX(C(I,J,K+1)+SX(C(I,J,K-1))*0.5
261 009B72I   SXB=(SX(C(I,J,K)+SX(C(I,J,K-1))*0.5
262 009BCC1   QE=GAE*(LUCE*CXE*UEE*EXE+USE*SXE)
263 009C00I   QW=GAW*(LUCW*CWX+UEE*EXW+UW*SXW)
264 009C34I   QN=GAN*(LUCN*CXN+UEN*EXN+UNN*SXN)
265 009C68I   QS=GAS*(LUCS*CXS+UEE*EXS+USS*SXS)
266 009C9C1   QT=GAT*(LUCT*CXT*UEE*EXT+UST*SXT)
267 009C90I   QB=GAB*(LUCB*CXB+UEB*EX9+UUB*SXB)
268 009D04I   SOC1=CX(C(I,J,K)*(QCE*QJW)+EX(C(I,J,K)*(QN-QS)+SX(C(I,J,K)*(QT-QB)
269 009D04I   QE=GAE*(WCE*CXE*UEE*EXE+WE*SXE)
270 009D08I   QW=GAW*(LUCW*CWX+UEE*EXW+UW*SXW)
271 009E40I   QN=GAN*(LUCN*CXN+UEN*EXN+UNN*SXN)
272 009E40I   QS=GAS*(LUCS*CXS+UEE*EXS+USS*SXS)
273 009E74I   QT=GAT*(LUCT*CXT*UEE*EXT+UST*SXT)
274 009E88I   QB=GAB*(LUCB*CXB+UEB*EX9+UUB*SXB)
275 009EDC1   SOC2=CY(C(I,J,K)*(QCE*QJW)+EX(C(I,J,K)*(QN-QS)+SY(C(I,J,K)*(QT-QB)
276 009F7C1   QE=GAE*(WCE*CXE*UEE*EXE+WE*SXE)
277 009FB0I   QW=GAW*(LUCW*CWX+UEE*EXW+UW*SXW)
278 009FE4I   QN=GAN*(LUCN*CXN+UEN*EXN+UNN*SXN)
279 00A018I   QS=GAS*(LUCS*CXS+UEE*EXS+USS*SXS)
280 00A04C1   QT=GAT*(LUCT*CXT*UEE*EXT+UST*SXT)
281 00A080I   QB=GAB*(LUCB*CXB+UEB*EX9+UUB*SXB)
282 00A084I   SOC3=CY(C(I,J,K)*(QCE*QJW)+EX(C(I,J,K)*(QN-QS)+SZ(C(I,J,K)*(QT-QB)
283 00A154I   SU(C(I,J,K)=SU(C(I,J,K)*(QCE*QJW)+SOC1+SOC2+SOC3+AP0(C(I,J,K)*F0(C(I,J,K)
284 00A226I   CY=(CY(C(I+1,J,K)+CY(C(I-1,J,K))*0.5
285 00A262I   CYW=(CY(C(I,J,K)+CY(C(I-1,J,K))*0.5

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286 00A2BEI
 287 00A31AI
 288 00A376I
 289 00A300I
 290 00A42AI
 291 00A486I
 292 00A4E2I
 293 00A53EI
 294 00A59AI
 295 00A5F4I
 296 00A64EI
 297 00A66AI
 298 00A706I
 299 00A762I
 300 00A7BEI
 301 00A818I
 302 00A872I
 303 00A8A6I
 304 00A8DAI
 305 00A905I
 306 00A942I
 307 00A976I
 308 00A99AI
 309 00AA4AI
 310 00AA7EI
 311 00AAE6I
 312 00AAE6I
 313 00AB1AI
 314 00AB45I
 315 00AB82I
 316 00AC22I
 317 00AC56I
 318 00AC8AI
 319 00ACBEI
 320 00ACF2I
 321 00AD26I
 322 00AD5AI
 323 00ADFAI
 324 00A58I
 325 00AE94I
 326 00AF10I
 327 00AF6CI
 328 00AFC8I
 329 00B022I
 330 00B07CI
 331 00B008I
 332 00B134I
 333 00B190I
 334 00B1ECI
 335 00B246I
 336 0032A0I
 337 00B2FCI
 338 00B358I
 339 00B394I
 340 00B410I
 341 00B46AI
 342 00B4C4I

CYN=(CY(I,J+1,K)+CY(I,J,K))+0.5
 CY5=(CY(I,J,K)+CY(I,J-1,K))+0.5
 CYT=(CY(I,J,K+1)+CY(I,J,K))+0.5
 CYB=(CY(I,J,K)+CY(I,J,K-1))+0.5
 EYE=(EY(I,J,K)+EY(I,J,K))+0.5
 EYW=(EY(I,J,K)+EY(I,J-1,K))+0.5
 EYN=(EY(I,J+1,K)+EY(I,J,K))+0.5
 EYS=(EY(I,J,K)+EY(I,J-1,K))+0.5
 EYT=(EY(I,J,K+1)+EY(I,J,K))+0.5
 EYB=(EY(I,J,K)+EY(I,J,K-1))+0.5
 SYE=(SY(I,J+1,K)+SY(I,J,K))+0.5
 SYW=(SY(I,J,K)+SY(I,J-1,K))+0.5
 SYN=(SY(I,J+1,K)+SY(I,J,K))+0.5
 SYS=(SY(I,J,K)+SY(I,J-1,K))+0.5
 SYB=(SY(I,J+1,K)+SY(I,J,K))+0.5
 SY3=(SY(I,J,K)+SY(I,J,K-1))+0.5
 QE=GAE*(CUCE*CYE+UEE*EYE+USE*SYE)
 QW=GAW*(CUCH*CYW+UEW*EYW+UW*SYW)
 QN=GAN*(UCUN*CYN+UEN*EYN+UN*SYN)
 QS=GAS*(UCS*SYS+UES*EYS+USS*SYS)
 QT=GAT*(UCT*CYT+UET*SYT+UST*SYT)
 QB=GAB*(UCB*CY3+UEB*EYB+UB*SYB)
 SOC1=C(X(I,J,K)*(2E-QW)+EX(I,J,K)*(QN-QS)+SX(I,J,K)*(QT-QB))
 QW=GAW*(UCVW*CYW+UEW*EYW+VSW*SYW)
 QN=GAN*(UCN*CYN+VEN*EYN+VSN*SYN)
 QS=GAS*(UCS*SYS+VES*EYS+VSS*SYS)
 QT=GAT*(UCVCT*CYT+VET*EYT+VST*SYT)
 QB=GAB*(UCVB*CY3+UEB*EYB+VB*SYB)
 SOC2=CY(I,J,K)*(QE-CW)+EY(I,J,K)*(QN-QS)+SY(I,J,K)*(QT-QB)
 QE=GAE*(WCE*CYE+HEE*EYE+WE*SYE)
 QW=GAW*(UCW*CYW+UEW*EYW+WSW*SYW)
 QN=GAN*(UCN*CYN+VEN*EYN+VSN*SYN)
 QS=GAS*(UCS*SYS+WES*EYS+VSS*SYS)
 QT=GAT*(UCWCT*CYT+WET*EYT+WST*SYT)
 QB=GAB*(UCB*CY2+WE*EYB+WSB*SYB)
 SOC3=C(Z(I,J,K)*(QE-QW)+EZ(I,J,K)*(QN-QS)+SZ(I,J,K)*(QT-QB))
 DV(C,I,J,K)=D(Y(I,J,K))+SOC1+SOC2+SOC3
 CZE=(CZ(I,J+1,K)+CZ(I,J,K))+0.5
 CZW=(CZ(I,J,K)+CZ(I-1,J,K))+0.5
 CZN=(CZ(I,J+1,K)+CZ(I,J,K))+0.5
 CZS=(CZ(I,J,K)+CZ(I,J-1,K))+0.5
 CZT=(CZ(I,J,K+1)+CZ(I,J,K))+0.5
 CZ9=(CZ(I,J,J)+CZ(I,J,K-1))+0.5
 EZE=(EZ(I,J+1,J,K)+EZ(I,J,J))+0.5
 EZW=(EZ(I,J,K)+EZ(I,J-1,J,K))+0.5
 CZS=(CZ(I,J,K)+CZ(I,J,K))+0.5
 EZN=(EZ(I,J,J)+EZ(I,J,K))+0.5
 EZS=(EZ(I,J,K)+EZ(I,J-1,J,K))+0.5
 EZT=(EZ(I,J,K+1)+EZ(I,J,K))+0.5
 EZB=(EZ(I,J,J)+EZ(I,J,K-1))+0.5
 SZS=(SZ(I,J,J)+SZ(I,J,K))+0.5
 EZW=(EZ(I,J,K)+EZ(I,J-1,J,K))+0.5
 SZN=(SZ(I,J,J)+SZ(I,J,K))+0.5
 SZS=(SZ(I,J,K)+SZ(I,J-1,J,K))+0.5
 SZT=(SZ(I,J,K+1)+SZ(I,J,K))+0.5
 SZB=(SZ(I,J,K)+SZ(I,J,K-1))+0.5
 QE=GAE*(CUCE*CYE+UEE*EYE+USE*SYE)

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343 0084F8I 2W=GAW*(UCW*CZW+UEW*EZW+UW*SZW)
344 00852CI QN=GAN*(UCN*CZN+UN*EN*EZN+UN*SZN)
345 008560I 2S=GAS*(UCS*CZS+UES*EZS+USS*SZS)
346 008594I 2T=GAT*(UCT*CZT+UET*EZT+UST*SZT)
347 0085C8I 2B=GAB*(UCB*CZB+UEB*EZB+USS*SZB)
348 0085FCI SOC1=C(X(I,J,K))*EX((I,J,K)*(QN-QS)+SX(I,J,K)*(QT-QB)
349 00869CI CE=GAE*(VCE*CZE+VEE*EZE+VSE*SZE)
350 0086D0I 3W=GAW*(VCW*CZW+VEM*EZM+VSM*SZW)
351 008704I QN=GAN*(VCN*CZN+VENEZN+VSN*SZN)
352 008738I 2S=GAS*(VC5*CZS+VEE*EZE+VSS*SZS)
353 00874CI 2T=GAT*(VC7*CZT+VET*EZT+VST*SZT)
354 008740I 2B=GAB*(VCB3*CZB+VEB*EZB+VSB*SZB)
355 0087D6I SOC2=C(Y(I,J,K)*(QE-CQ)+EY(I,J,K)*(QN-QS)+SY(I,J,K)*(CT-QB)
356 008874I 2E=GAE*(WC E*CZE+WE *EZE+WE *SZE)
357 008848I 2H=GAW*(WCW*CZW+WEW*EZW+WSW*SZW)
358 0088D2I 2N=GAN*(WCN*CZN+HEN*EZN+WSN*SZN)
359 008910I 2S=GAS*(WC5*CZS+WE5*EZE+WS5*SZS)
360 008944I 2T=GAT*(WC7*CZT+WE7*EZT+WT*SZT)
361 008973I 2B=GAB*(WC5*CZB+WE5*EZB+WS5*SZB)
362 00894CI SOC3=C(Z(I,J,K)*(QE-CQ)+EZ(I,J,K)*(QN-QS)+SZ(I,J,K)*(QT-QB)
363 008A4CI 2W(I,J,K)=DW(Z,J,K)+SOC1+SCC2+SOC3
364 008A4AI SU(I,J,K)=SU(Z,J,K)+TJC(I,J,K)
365 008A31CI SP(Z,J,K)=SPK(Z,J,K)*TJC(I,J,K)
366 008338I 15 CONTINUE
367 008382I 367 GO TO EC
368 0083E2I C-----V-SCURCE
369 0083E2I 2 CONTINUE
370 008382I DO 25 I=IS,IT
371 0083FAI DO 25 J=JS,JT
372 008C12I DO 25 K=KS,KT
373 008C2AI SU(I,J,K)=(DV(I,J,K)+APC(I,J,K)*FD(I,J,K))*TJO(I,J,K)
374 008C6EI 25 CONTINUE
375 008D36I GC TO EC
376 008382I C-----W-SCURCE
377 00803CI 3 CONTINUE
378 00803CI DO 35 I=IS,IT
379 008054I DO 35 J=JS,JT
380 00806CI DO 35 K=KS,KT
381 008C84I SU(I,J,K)=(CW(I,J,K)+AP0(I,J,K)*FD(I,J,K))*TJO(I,J,K)
382 008E43I 35 CONTINUE
383 008E90I 35 GO TO EC
384 008E90I C-----TM-SOURCE
385 008E90I 4 CONTINUE
386 008E90I 4 GO TO EC
387 008E90I C-----K-SCURCE
388 008E9CI 5 CONTINUE
389 008E9CI DO 55 I=IS,IT
390 008E34I DO 55 J=JS,JT
391 008ECC4I DO 55 K=KS,KT
392 008E44I SU(I,J,K)=GEN(I,J,K)+PP(I,J,K)+AP0(I,J,K)*FO(I,J,K)
393 008FAAI P1=GEN(I,J,K)*2
394 008FDEI SP(I,J,K)=SPK(I,J,K)-CMU*P1*F(I,J,K)/VISE(I,J,K)
395 00C08AI SU(I,J,K)=SU(I,J,K)*TJC(I,J,K)
396 00C0FCI SP(I,J,K)=SP(I,J,K)*TJO(I,J,K)
397 00C15EI 55 CONTINUE
398 00C15EI 55 GO TO EC
399 00C156I C-----E-SCURCE

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**ORIGINAL PAGE IS
OF POOR QUALITY**

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400 00C1ACI   6  CONTINUE
401 00C1ACI   70 65 I=IS,IT
402 00C1D4I   DO 65 J=JS,JT
403 00C1ECI   DO 65 K=KS,KT
404 00C204I   P1=CEN(I,J,K)**2
405 00C238I   SU(I,J,K)=SU(I,J,K)+C1*CMU*GEN(I,J,K)*P1*D(K(I,J,K) /
406 1          VISE(I,J,K)+AP0(I,J,K)*FO(I,J,K)
407 00C355CI   TMOK=OK(I,J,K)+SMNUM
408 00C38EI   SP(I,J,K)=SPK(I,J,K)-C2*DEN(I,J,K)*F(I,J,K)/TMOK
409 00C43AI   SU(I,J,K)=SU(I,J,K)+TJC(I,J,K)
410 00C4ACI   SP(I,J,K)=SP(I,J,K)*TJC(I,J,K)
411 00C51EI   CONTINUE
412 00C566I   65  CONTINUE
413 00C566I   GO TO 6C
414 00C566I   C-----P-SCURCE
415 00C566I   7  CONTINUE
416 00C584I   70 75 I=IS,IT
417 00C595CI   75 75 J=JS,JT
418 00C584I   SU(I,J,K)=SU(I,J,K)+0.25*GEN(I,J,K)+4PO(I,J,K)*FO(I,J,K)
419 00C680I   TMOK=OK(I,J,K)+SMNUM
420 00C682I   SP(I,J,K)=SPK(I,J,K)-DE(I,J,K)/TMOK
421 00C72CI   SU(I,J,K)=SU(I,J,K)*TJC(I,J,K)
422 00C795I   SP(I,J,K)=SP(I,J,K)*TJC(I,J,K)
423 00C810I   75  CONTINUE
424 00C853I   60  CONTINUE
425 00C858I   C-----MODIFY WALL BOUNDARY CONITIONS THRU WALL FUNCTIONS
426 00C858I   IF(IG .NE. 2) GC TO 41C
427 00C866I   CALL BCUNC(IE,F)
428 00C89CI   410  CONTINUE
429 00C89CI   C-----SET SYMMETRIC, CYCLIC AND EXIT LINK COEFF.
430 00C89CI   CALL SYMOUT(2,IE,IS,IT,JS,JT,KS,KT)
431 00C89CI   C-----LINK CCEFF. ASSEMBLY AND BLOCKAGES
432 00C8D8I   DO 506 I=IS,IT
433 00C8F0I   DO 500 J=JS,JT
434 00C908I   DO 500 K=KS,KT
435 00C920I   F1(I,J,K)=F(I,J,K)
436 00C972I   ANAE=AEC(I,J,K)+AW(I,J,K)+AN(I,J,K)+AS(I,J,K)+AT(I,J,K)+1
437 00CA82I   AB(I,J,K)+AP0(I,J,K)
438 00CA82I   AP(I,J,K)=ANAB-SP(I,J,K)
439 00CA04I   PDUV=1.0
440 00CA04I   IF(MCC(I,J,K) .LT. 1) GC TO 530
441 00CB16I   AP(I,J,K)=ALF
442 00CB42I   AN(I,J,K)=0.0
443 00CB66I   ASC(I,J,K)=0.0
444 00CB94I   AE(I,J,K)=0.0
445 00CB94I   AW(I,J,K)=0.0
446 00CB94I   AT(I,J,K)=0.0
447 00CC1EI   AB(I,J,K)=0.0
448 00CC44I   SU(I,J,K)=F(I,J,K)
449 00CC9CI   PDUV=0.0
450 00CC48I   530  CONTINUE
451 00CC48I   C-----UNDER RELAXATION
452 00CC48I   P1=1-2*AP(I,J,K)
453 00CCDAI   AP(I,J,K)=AP(I,J,K)/ALF
454 00CD2CI   SU(I,J,K)=SU(I,J,K)+PDUV*(1.0-ALF)*AP(I,J,K)*F(I,J,K)
455 00CD2CI   IF(IE .EQ. 1) DU(I,J,K)=TJO(I,J,K)*PDUV/(P1-ANAB)
456 00CE4CI   500  CONTINUE

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457 00CE94I C----LINEAR EQUATIONS SLOVER
458      DO 550 I=1 ISWF
459 00CEA8I 550 CALL LINERX(1,IS,J$,KS,IT,JT,KT,FF)
460 C----CALCULATE MAXIMUM CORRECTION OF CURRENT ITERATION
461      DO 555 IT=IS,IT
462      DO 555 J=JS,JT
463      DO 555 K=KS,KT
464      P1=ABSF(I,J,K)-F1(I,J,K)
465      ERRF=MAX1(ERRF,P1)
466 00CFCAI 555  CONTINUE
467 000032I      RETURN
468
469 00006AI C----P EQUATION
470 00003AI 10  CONTINUE
471 00003AI      IS=2
472 00C042I      IT=L
473 00004EI      JS=2
474 00D056I      JT=M
475 00D062I      KS=2
476 00D06AI      KT=N
477      C----SOURCE AT PX LOCATIONS
478      DO 310 I=IS,IT
479      DO 310 J=JS,JT
480      DO 310 K=KS,KT
481      F(I,J,K)=0.0
482 00D0F0I      SUK(I,J,K)=0.0
483 00D11CI      SPK(I,J,K)=0.0
484 00D148I      SPC(I,J,K)=0.0
485 00D174I      DENE=DEN(I,J,K)
486 00D1A0I      DENH=DEN(I-1,J,K)
487 00D100I      DENH=0.25*(DEN(I,J,K)+DEN(I-1,J,K)+DEN(I,J+1,K)+DEN(I-1,J+1,K))
488 00D284I      DENS=0.25*(CEN(I,J,K)+CEN(I-1,J,K)+DEN(I,J-1,K)+DEN(I-1,J-1,K))
489 00D338I      DENT=0.25*(DEN(I,J,K)+DEN(I-1,J,K)+DEN(I,J,K+1)+DEN(I-1,J,K+1))
490 00D3E8I      DENE=0.25*(DEN(I,J,K)+DEN(I-1,J,K)+DEN(I,J-1,K)+DEN(I-1,J-1,K))
491 00D498I      UE=U(I,J,K)
492 00D4C4I      UW=U(I-1,J,K)
493 00D4F4I      UN=C_25*(U(I,J,K)+U(I-1,J,K)+U(I,J+1,K)+U(I-1,J+1,K))
494 00D5A8I      US=C_25*(U(I,J,K)+U(I-1,J,K)+U(I,J-1,K)+U(I-1,J-1,K))
495 00D65CI      UT=C_25*(U(I,J,K)+U(I-1,J,K)+U(I,J,K+1)+U(I-1,J,K+1))
496 00D70CI      US=C_25*(U(I,J,K)+U(I-1,J,K)+U(I,J,K-1)+U(I-1,J,K-1))
497 00D76CI      VE=V(I,J,K)
498 00D7E8I      VW=V(I-1,J,K)
499 00D818I      VN=C_25*(V(I,J,K)+V(I-1,J,K)+V(I,J+1,K)+V(I-1,J+1,K))
500 00D8CC1      VS=C_25*(V(I,J,K)+V(I-1,J,K)+V(I,J-1,K)+V(I-1,J-1,K))
501 00D980I      VT=C_25*(V(I,J,K)+V(I-1,J,K)+V(I,J,K+1)+V(I-1,J,K+1))
502 00DA30I      V3=C_25*(V(I,J,K)+V(I-1,J,K)+V(I,J,K-1)+V(I-1,J,K-1))
503 00DAE0I      WE=W(I,J,K)
504 00B0C1      WW=W(I-1,J,K)
505 00D63CI      AN=G_25*(W(I,J,K)+W(I-1,J,K)+W(I,J+1,K)+W(I-1,J+1,K))
506 00D5F0I      WS=C_25*(W(I,J,K)+W(I-1,J,K)+W(I,J-1,K)+W(I-1,J-1,K))
507 00DCA4I      HT=C_25*(W(I,J,K)+W(I-1,J,K)+W(I,J,K+1)+W(I-1,J,K+1))
508 00D54I      WB=C_25*(W(I,J,K)+W(I-1,J,K)+W(I,J,K-1)+W(I-1,J,K-1))
509 00CE04I      CXQ=0.5*(CX(I,J,K)+CX(I-1,J,K))
510 00DE60I      EXQ=0.5*(EX(I,J,K)+EX(I-1,J,K))
511 00DEBCI      SXQ=0.5*(SX(I,J,K)+SX(I-1,J,K))
512 00DF18I      CYQ=0.5*(CY(I,J,K)+CY(I-1,J,K))
513 00CF74I      EYQ=0.5*(EY(I,J,K)+EY(I-1,J,K))

```

```

514 000F001
515 00E02CI
516 00E088I
517 00E0E4I
518 00E140I
519 00E174I
520 00E1A8I
521 00E1CC1
522 00E210I
523 00E244I
524 00E278I
525 00E322I
526 00E364I
527 00E382I
528 00E39AI
529 00E392I
530 00E393I
531 00E30EI
532 00E40EI
533 00E4C2I
534 00E576I
535 00E626I
536 00E636I
537 00E702I
538 00E732I
539 00E7E6I
540 00E89AI
541 00E94AI
542 00E9FAI
543 00E265I
544 00EA56I
545 00E60AI
546 00E938I
547 00EC66I
548 00ED15I
549 00ED4AI
550 00ED7AI
551 00EE2EI
552 00EEE2I
553 00EF92I
554 00F042I
555 00F09EI
556 00FOFAI
557 00F156I
558 00F192I
559 00F20EI
560 00F26AI
561 00F266I
562 00F322I
563 00F376I
564 00F382I
565 00F3E6I
566 00F41AI
567 00F44EI
568 00F482I
569 00F486I
570 00F560I
571 310 CONTINUE
572 C-----SOURCE AT PY LOCATIONS
573 DO 311 1=IS,IT
574 311 J=JS,JT
575 20 211 K=KS,KT
576 0ENN DEN(C,J,K)
577 DENS=DEN(C,J-1,K)
578 DENE=0.25*(DEN(C,J,K)+CEN(C,J-1,K)+DEN(I+1,J,K)+DEN(I+1,J-1,K))
579 DENW=0.25*(DEN(C,J,K)+DEN(C,J-1,K)+DEN(I-1,J,K)+DEN(I-1,J-1,K))
580 DENT=0.25*(DEN(C,J,K)+CEN(C,J-1,K)+DEN(I,J,K)+DEN(I,J-1,K))
581 DENG=0.25*(DEN(C,J,K)+CEN(C,J-1,K)+DEN(I,J,K-1)+DEN(I,J-1,K-1))
582 UN=L(C,J,K)
583 US=L(C,J-1,K)
584 US=0.25*(U(C,J,K)+U(C,J-1,K)+U(I,J,K)+U(I,J-1,K)+U(I+1,J,K)+U(I+1,J-1,K))
585 UW=C_25*(U(C,J,K)+U(C,J-1,K)+U(I,J,K)+U(I,J-1,K)+U(I-1,J,K)+U(I-1,J-1,K))
586 UT=C_25*(U(C,J,K)+U(I,J,K)+U(I,J-1,K)+U(C,I,J,K)+U(C,I,J-1,K)+U(I,J,K+1)+U(I,J-1,K+1))
587 US=C_25*(U(C,J,K)+U(I,J,K)+U(I,J-1,K)+U(I,J,K-1)+U(I,J-1,K-1))
588 VN=Y(C,J,K)
589 VS=Y(C,J-1,K)
590 VE=0.25*(V(C,J,K)+V(C,J-1,K)+V(I,J,K)+V(I,J-1,K)+V(I+1,J,K)+V(I+1,J-1,K))
591 VW=C_25*(V(C,J,K)+V(C,J-1,K)+V(I,J,K)+V(I,J-1,K)+V(I-1,J,K)+V(I-1,J-1,K))
592 VT=C_25*(V(C,J,K)+V(I,J,K)+V(I,J-1,K)+V(I,J,K+1)+V(I,J-1,K+1))
593 V3=C_25*(V(C,J,K)+V(C,J-1,K)+V(C,J,K-1)+V(C,J-1,K-1))
594 WN=W(C,J-1,K)
595 WS=W(C,J-1,K)
596 WE=C_25*(W(C,J,K)+W(C,J-1,K)+W(I,J,K)+W(I,J-1,K)+W(I+1,J,K)+W(I+1,J-1,K))
597 WW=C_25*(W(I,J,K)+W(I,J-1,K)+W(I,J,K)+W(I,J-1,K)+W(I-1,J,K)+W(I-1,J-1,K))
598 WT=C_25*(W(I,J,K)+W(I,J-1,K)+W(I,J,K)+W(I,J-1,K)+W(I,J,K+1)+W(I,J-1,K+1))
599 WB=C_25*(W(I,J,K)+W(I,J-1,K)+W(I,J,K)+W(I,J-1,K)+W(I,J,K-1)+W(I,J-1,K-1))
599 CXQ=C_5*(CX(I,J,K)+CX(I,J-1,K))
600 EXC=0.5*EX(C,J,K)+EX(C,J-1,K)
601 SXQ=0.5*(SX(I,J,K)+SX(I,J-1,K))
602 CYC=0.5*(CY(I,J,K)+CY(I,J-1,K))
603 EYQ=0.5*(EY(I,J,K)+EY(I,J-1,K))
604 SYQ=0.5*(SY(I,J,K)+SY(I,J-1,K))
605 CZQ=0.5*(CZ(I,J,K)+CZ(I,J-1,K))
606 EZQ=0.5*(EZ(I,J,K)+EZ(I,J-1,K))
607 SZQ=0.5*(SZ(I,J,K)+SZ(I,J-1,K))
608 CE=DENE*UE*CXQ+VE*CYC+WE*CZQ
609 CW=CENW*(UH*CXQ+VW*CYQ+WN*CZQ)
610 CN=CENN*(UN*EXQ+VN*EYQ+WN*EZQ)
611 CS=CENS*(US*EXQ+VS*EYQ+WS*EZQ)
612 CT=CENT*(UT*SXQ+VT*SYQ+WT*SZQ)
613 CB=CENB*(UB*5XQ+VB*SYQ+WB*SZQ)
614 SPK(C,J,K)=-CE-CW+CN-CS+CT-CB)*(TJO(I,J-1,K)+TJO(I,J,K))*0.5
615 311 CONTINUE

```

C-----SOURCE AT PZ LOCATIONS

```

571 00F5A8I
572 00F61CI
573 00F5COI
574 00F5D8I
575 00F5F0I
576 00F61CI
577 00F64AI
578 00F6FAI
579 00F7AAI
580 00F35AI
581 00F90AI
582 00F936I
583 00F964I
584 00FA14I
585 00FAC4I
586 00F374I
587 00FC24I
588 00FC50I
589 00FC78I
590 00F526I
591 00F6DCEI
592 00F588I
593 00FF36I
594 00FF66I
595 00FF98I
596 010048I
597 010CF8I
598 0101A8I
599 010258I
600 010252I
601 01030CI
602 010365I
603 0103COI
604 010414I
605 010474I
606 0104CEI
607 010528I
608 010582I
609 010536I
610 0105E4I
611 010615I
612 010652I
613 010686I
614 010634I
615 010762I
616 0107AAI
617 0107C2I
618 0107DAI
619 0107F2I
620 0108COI
621 0108COI
622 01080EI
623 01081CI
624 0108COI
625 010964I
626 010408I
627 010AACI

C-----LINK CCEFF. AND SOURCE TERM AT CELL CENTER
      KK=(DU(I,J,K)+DU(I,J,J,K)+DU(I,J,K,K)+DU(I,J,J,J,K))*0.25
      GAE=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAW=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAN=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAS=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAT=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25

      KK=K-1
      GAE=(DU(I,J,K)+DU(I,J,J,K)+DU(I,J,K,K)+DU(I,J,J,J,K))*0.25
      GAW=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAN=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAS=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25
      GAT=(DU(I,J,J,K)+DU(I,J,J,K)+DU(I,J,J,K,K)+DU(I,J,J,J,K))*0.25

```

ORIGINAL PAGE IS
OF POOR QUALITY

C-----PRESSURE AND VELOCITIES CORRECTIONS

```

685      011854I          PPREFF=F(2,2,2)
686          00 280 I=IS,IT
687          00 630 J=JS,JT
688          00 680 K=KS,KT
689          IF(NC(I,J,K) .GE. 2) GC TO 380
690          P1=F(I,J,K)-PPREF
691          P(I,J,K)=P(I,J,K)+ALP*P1
692          ERRF=AMAX1(ERRF,ABS(P1))
693          600 CONTINUE
694          00 600 I=2,LT
695          00 600 J=2,MT
696          00 600 K=2,NT
697          00 600 IF(NC(I,J,K) .GE. 1) GC TO 600
698          J1A32I
699          011A68I
700          011A76I
701          011A34I
702          011A92I
703          011B45I
704          011C0AI
705          011CC6I
706          011D32I
707          011E3EI
708          011EF4I
709          011FOCI
710          011F1EI
711          011F30I
712          011F5CI
713          011F88I
714          011F94I
715          011FE0I
716          01200C1
717          012038I
718          012064I
719          012090I
720          0120BCI
721          0120EAI
722          012118I
723          012146I
724          0121C0I
725          01223A1
726          0122B4I
727          0122FCI
728          012304I

          KK=K+1
          PE=(F(I,I,J,K)+F(I,I,J,K)+F(I,I,J,K)+F(I,I,J,K))*0.25
          PW=(F(I,J,J,K)+F(I,J,J,K)+F(I,J,J,K)+F(I,J,J,K))*0.25
          NE=(F(I,J,J,K)+F(I,J,J,K)+F(I,J,J,K)+F(I,J,J,K))*0.25
          PS=(F(I,J,K,K)+F(I,J,K,K)+F(I,J,K,K)+F(I,J,K,K))*0.25
          PT=(F(I,J,K,K)+F(I,J,K,K)+F(I,J,K,K)+F(I,J,K,K))*0.25
          P5=(F(I,J,K,K)+F(I,J,K,K)+F(I,J,K,K)+F(I,J,K,K))*0.25
          PCX1=PE-PW
          PE0A=PN-PS
          PSCI=PT-PB
          CXQ=CX(I,J,K)
          EXQ=EX(I,J,K)
          SXQ=SX(I,J,K)
          CYQ=CY(I,J,K)
          EYQ=EY(I,J,K)
          SYQ=SY(I,J,K)
          CZQ=CZ(I,J,K)
          EZQ=EZ(I,J,K)
          SZQ=SZ(I,J,K)
          PXX=PCXI*CXQ+PECA*EXC+PSCI*SXQ
          PYY=PCXI*CYQ+PEA*EYQ+PSCI*SYQ
          PZZ=PCXI*CZQ+PECA*EZQ+PSCI*SZQ
          U(I,J,K)=U(I,J,K)-CU(I,J,K)*PXX
          V(I,J,K)=V(I,J,K)-DU(I,J,K)*PYY
          W(I,J,K)=W(I,J,K)-CU(I,J,K)*PZZ
          CCNTINUE
          RETURN
          END

```

NO ERRORS:F70 ROS-01.0C SUBROUTINE SOLVED 02/21/86 09:58:31 TABLE SPACE: 18 KB
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 203 WORDS
 SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

```

1 00000011 1520
2 00000041 1521
3 00000042 1522
4 1/COEF 1/COEF(21,18,10),SU(21,18,10),SP(21,18,10),SU(21,18,10),
5 2 SPK(21,18,10),AE(21,18,10),AW(21,18,10),AN(21,18,10),
6 3 AS(21,18,10),AT(21,18,10),AB(21,18,10),AP(21,18,10),
7 C-----LINEAR EQUATIONS SOLVERS
8 00000042 1523
9 10 GO TO (1/2),ISOL
10 C-----LINE-RELAXATION USING T0MA
11 003C861 1524
12 003C861 1525
13 003C861 1526
14 003CCE1 1527
15 10 CONTINUE
16 003DC81 1528
17 003E101 1529
18 003E241 1530
19 003E3C1 1531
20 003E541 1532
21 003E901 1533
22 003EA81 1534
23 003ED41 1535
24 003F0C1 1536
25 10 CONTINUE
26 003FEC1 1537
27 00401E1 1538
28 0040541 1539
29 0040661 1540
30 0040881 1541
31 0040E01 1542
32 0040F81 1543
33 0041101 1544
34 00419E1 1545
35 0041CE1 1546
36 10 CONTINUE
37 0041D61 1547
38 0041D61 1548
39 0041EE1 1549
40 0042021 1550
41 0042161 1551
42 00422A1 1552
43 00423E1 1553
44 0042561 1554
45 00426E1 1555
46 0042801 1556
47 0042F21 1557
48 0043341 1558
49 00437C1 1559
50 0043941 1560
51 00439C1 1561
52 0043EE1 1562
53 00440A1 1563
54 00442E1 1564
55 00446E1 1565
56 0044AE1 1566
57 0044FE1 1567
1 10 CONTINUE
2 0041061 1568
3 0041EE1 1569
4 0042021 1570
5 0042161 1571
6 00422A1 1572
7 00423E1 1573
8 0042561 1574
9 00426E1 1575
10 0042801 1576
11 0042F21 1577
12 0043341 1578
13 00437C1 1579
14 0043941 1580
15 00439C1 1581
16 0043EE1 1582
17 00440A1 1583
18 00442E1 1584
19 00446E1 1585
20 0044AE1 1586
21 0044FE1 1587
22 10 CONTINUE
23 0041061 1588
24 0041EE1 1589
25 0042021 1590
26 0042161 1591
27 00422A1 1592
28 00423E1 1593
29 0042561 1594
30 00426E1 1595
31 0042801 1596
32 0042F21 1597
33 0043341 1598
34 00437C1 1599
35 0043941 1600
36 00439C1 1601
37 0043EE1 1602
38 00440A1 1603
39 00442E1 1604
40 00446E1 1605
41 0044AE1 1606
42 0044FE1 1607
43 10 CONTINUE
44 0041061 1608
45 0041EE1 1609
46 0042021 1610
47 0042161 1611
48 00422A1 1612
49 00423E1 1613
50 0042561 1614
51 00426E1 1615
52 0042801 1616
53 0042F21 1617
54 0043341 1618
55 00437C1 1619
56 0043941 1620
57 00439C1 1621
58 0043EE1 1622
59 00440A1 1623
60 00442E1 1624
61 00446E1 1625
62 0044AE1 1626
63 0044FE1 1627
64 10 CONTINUE
65 0041061 1628
66 0041EE1 1629
67 0042021 1630
68 0042161 1631
69 00422A1 1632
70 00423E1 1633
71 0042561 1634
72 00426E1 1635
73 0042801 1636
74 0042F21 1637
75 0043341 1638
76 00437C1 1639
77 0043941 1640
78 00439C1 1641
79 0043EE1 1642
80 00440A1 1643
81 00442E1 1644
82 00446E1 1645
83 0044AE1 1646
84 0044FE1 1647
85 10 CONTINUE
86 0041061 1648
87 0041EE1 1649
88 0042021 1650
89 0042161 1651
90 00422A1 1652
91 00423E1 1653
92 0042561 1654
93 00426E1 1655
94 0042801 1656
95 0042F21 1657
96 0043341 1658
97 00437C1 1659
98 0043941 1660
99 00439C1 1661
100 0043EE1 1662
101 00440A1 1663
102 00442E1 1664
103 00446E1 1665
104 0044AE1 1666
105 0044FE1 1667
106 10 CONTINUE
107 0041061 1668
108 0041EE1 1669
109 0042021 1670
110 0042161 1671
111 00422A1 1672
112 00423E1 1673
113 0042561 1674
114 00426E1 1675
115 0042801 1676
116 0042F21 1677
117 0043341 1678
118 00437C1 1679
119 0043941 1680
120 00439C1 1681
121 0043EE1 1682
122 00440A1 1683
123 00442E1 1684
124 00446E1 1685
125 0044AE1 1686
126 0044FE1 1687
127 10 CONTINUE
128 0041061 1688
129 0041EE1 1689
130 0042021 1690
131 0042161 1691
132 00422A1 1692
133 00423E1 1693
134 0042561 1694
135 00426E1 1695
136 0042801 1696
137 0042F21 1697
138 0043341 1698
139 00437C1 1699
140 0043941 1700
141 00439C1 1701
142 0043EE1 1702
143 00440A1 1703
144 00442E1 1704
145 00446E1 1705
146 0044AE1 1706
147 0044FE1 1707
148 10 CONTINUE
149 0041061 1708
150 0041EE1 1709
151 0042021 1710
152 0042161 1711
153 00422A1 1712
154 00423E1 1713
155 0042561 1714
156 00426E1 1715
157 0042801 1716
158 0042F21 1717
159 0043341 1718
160 00437C1 1719
161 0043941 1720
162 00439C1 1721
163 0043EE1 1722
164 00440A1 1723
165 00442E1 1724
166 00446E1 1725
167 0044AE1 1726
168 0044FE1 1727
169 10 CONTINUE
170 0041061 1728
171 0041EE1 1729
172 0042021 1730
173 0042161 1731
174 00422A1 1732
175 00423E1 1733
176 0042561 1734
177 00426E1 1735
178 0042801 1736
179 0042F21 1737
180 0043341 1738
181 00437C1 1739
182 0043941 1740
183 00439C1 1741
184 0043EE1 1742
185 00440A1 1743
186 00442E1 1744
187 00446E1 1745
188 0044AE1 1746
189 0044FE1 1747
190 10 CONTINUE
191 0041061 1748
192 0041EE1 1749
193 0042021 1750
194 0042161 1751
195 00422A1 1752
196 00423E1 1753
197 0042561 1754
198 00426E1 1755
199 0042801 1756
200 0042F21 1757
201 0043341 1758
202 00437C1 1759
203 0043941 1760
204 00439C1 1761
205 0043EE1 1762
206 00440A1 1763
207 00442E1 1764
208 00446E1 1765
209 0044AE1 1766
210 0044FE1 1767
211 10 CONTINUE
212 0041061 1768
213 0041EE1 1769
214 0042021 1770
215 0042161 1771
216 00422A1 1772
217 00423E1 1773
218 0042561 1774
219 00426E1 1775
220 0042801 1776
221 0042F21 1777
222 0043341 1778
223 00437C1 1779
224 0043941 1780
225 00439C1 1781
226 0043EE1 1782
227 00440A1 1783
228 00442E1 1784
229 00446E1 1785
230 0044AE1 1786
231 0044FE1 1787
232 10 CONTINUE
233 0041061 1788
234 0041EE1 1789
235 0042021 1790
236 0042161 1791
237 00422A1 1792
238 00423E1 1793
239 0042561 1794
240 00426E1 1795
241 0042801 1796
242 0042F21 1797
243 0043341 1798
244 00437C1 1799
245 0043941 1800
246 00439C1 1801
247 0043EE1 1802
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249 00442E1 1804
250 00446E1 1805
251 0044AE1 1806
252 0044FE1 1807
253 10 CONTINUE
254 0041061 1808
255 0041EE1 1809
256 0042021 1810
257 0042161 1811
258 00422A1 1812
259 00423E1 1813
260 0042561 1814
261 00426E1 1815
262 0042801 1816
263 0042F21 1817
264 0043341 1818
265 00437C1 1819
266 0043941 1820
267 00439C1 1821
268 0043EE1 1822
269 00440A1 1823
270 00442E1 1824
271 00446E1 1825
272 0044AE1 1826
273 0044FE1 1827
274 10 CONTINUE
275 0041061 1828
276 0041EE1 1829
277 0042021 1830
278 0042161 1831
279 00422A1 1832
280 00423E1 1833
281 0042561 1834
282 00426E1 1835
283 0042801 1836
284 0042F21 1837
285 0043341 1838
286 00437C1 1839
287 0043941 1840
288 00439C1 1841
289 0043EE1 1842
290 00440A1 1843
291 00442E1 1844
292 00446E1 1845
293 0044AE1 1846
294 0044FE1 1847
295 10 CONTINUE
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297 0041EE1 1849
298 0042021 1850
299 0042161 1851
300 00422A1 1852
301 00423E1 1853
302 0042561 1854
303 00426E1 1855
304 0042801 1856
305 0042F21 1857
306 0043341 1858
307 00437C1 1859
308 0043941 1860
309 00439C1 1861
310 0043EE1 1862
311 00440A1 1863
312 00442E1 1864
313 00446E1 1865
314 0044AE1 1866
315 0044FE1 1867
316 10 CONTINUE
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318 0041EE1 1869
319 0042021 1870
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322 00423E1 1873
323 0042561 1874
324 00426E1 1875
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326 0042F21 1877
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331 0043EE1 1882
332 00440A1 1883
333 00442E1 1884
334 00446E1 1885
335 0044AE1 1886
336 0044FE1 1887
337 10 CONTINUE
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339 0041EE1 1889
340 0042021 1890
341 0042161 1891
342 00422A1 1892
343 00423E1 1893
344 0042561 1894
345 00426E1 1895
346 0042801 1896
347 0042F21 1897
348 0043341 1898
349 00437C1 1899
350 0043941 1900
351 00439C1 1901
352 0043EE1 1902
353 00440A1 1903
354 00442E1 1904
355 00446E1 1905
356 0044AE1 1906
357 0044FE1 1907
358 10 CONTINUE
359 0041061 1908
360 0041EE1 1909
361 0042021 1910
362 0042161 1911
363 00422A1 1912
364 00423E1 1913
365 0042561 1914
366 00426E1 1915
367 0042801 1916
368 0042F21 1917
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370 00437C1 1919
371 0043941 1920
372 00439C1 1921
373 0043EE1 1922
374 00440A1 1923
375 00442E1 1924
376 00446E1 1925
377 0044AE1 1926
378 0044FE1 1927
379 10 CONTINUE
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381 0041EE1 1929
382 0042021 1930
383 0042161 1931
384 00422A1 1932
385 00423E1 1933
386 0042561 1934
387 00426E1 1935
388 0042801 1936
389 0042F21 1937
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391 00437C1 1939
392 0043941 1940
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396 00442E1 1944
397 00446E1 1945
398 0044AE1 1946
399 0044FE1 1947
400 10 CONTINUE
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409 0042801 1956
410 0042F21 1957
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418 00446E1 1965
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420 0044FE1 1967
421 10 CONTINUE
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423 0041EE1 1969
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426 00422A1 1972
427 00423E1 1973
428 0042561 1974
429 00426E1 1975
430 0042801 1976
431 0042F21 1977
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435 00439C1 1981
436 0043EE1 1982
437 00440A1 1983
438 00442E1 1984
439 00446E1 1985
440 0044AE1 1986
441 0044FE1 1987
442 10 CONTINUE
443 0041061 1988
444 0041EE1 1989
445 0042021 1990
446 0042161 1991
447 00422A1 1992
448 00423E1 1993
449 0042561 1994
450 00426E1 1995
451 0042801 1996
452 0042F21 1997
453 0043341 1998
454 00437C1 1999
455 0043941 2000
456 00439C1 2001
457 0043EE1 2002
458 00440A1 2003
459 00442E1 2004
460 00446E1 2005
461 0044AE1 2006
462 0044FE1 2007
463 10 CONTINUE
464 0041061 2008
465 0041EE1 2009
466 0042021 2010
467 0042161 2011
468 00422A1 2012
469 00423E1 2013
470 0042561 2014
471 00426E1 2015
472 0042801 2016
473 0042F21 2017
474 0043341 2018
475 00437C1 2019
476 0043941 2020
477 00439C1 2021
478 0043EE1 2022
479 00440A1 2023
480 00442E1 2024
481 00446E1 2025
482 0044AE1 2026
483 0044FE1 2027
484 10 CONTINUE
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486 0041EE1 2029
487 0042021 2030
488 0042161 2031
489 00422A1 2032
490 00423E1 2033
491 0042561 2034
492 00426E1 2035
493 0042801 2036
494 0042F21 2037
495 0043341 2038
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498 00439C1 2041
499 0043EE1 2042
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501 00442E1 2044
502 00446E1 2045
503 0044AE1 2046
504 0044FE1 2047
505 10 CONTINUE
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508 0042021 2050
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512 0042561 2054
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514 0042801 2056
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523 00446E1 2065
524 0044AE1 2066
525 0044FE1 2067
526 10 CONTINUE
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532 00423E1 2073
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535 0042801 2076
536 0042F21 2077
537 0043341 2078
538 00437C1 2079
539 0043941 2080
540 00439C1 2081
541 0043EE1 2082
542 00440A1 2083
543 00442E1 2084
544 00446E1 2085
545 0044AE1 2086
546 0044FE1 2087
547 10 CONTINUE
548 0041061 2088
549 0041EE1 2089
550 0042021 2090
551 0042161 2091
552 00422A1 2092
553 00423E1 2093
554 0042561 2094
555 00426E1 2095
556 0042801 2096
557 0042F21 2097
558 0043341 2098
559 00437C1 2099
560 0043941 2100
561 00439C1 2101
562 0043EE1 2102
563 00440A1 2103
564 00442E1 2104
565 00446E1 2105
566 0044AE1 2106
567 0044FE1 2107
568 10 CONTINUE
569 0041061 2108
570 0041EE1 2109
571 0042021 2110
572 0042161 2111
573 00422A1 2112
574 00423E1 2113
575 0042561 2114
576 00426E1 2115
577 0042801 2116
578 0042F21 2117
579 0043341 2118
580 00437C1 2119
581 0043941 2120
582 00439C1 2121
583 0043EE1 2122
584 00440A1 2123
585 00442E1 2124
586 00446E1 2125
587 0044AE1 2126
588 0044FE1 2127
589 10 CONTINUE
590 0041061 2128
591 0041EE1 2129
592 0042021 2130
593 0042161 2131
594 00422A1 2132
595 00423E1 2133
596 0042561 2134
597 00426E1 2135
598 0042801 2136
599 0042F21 2137
600 0043341 2138
601 00437C1 2139
602 0043941 2140
603 00439C1 2141
604 0043EE1 2142
605 00440A1 2143
606 00442E1 2144
607 00446E1 2145
608 0044AE1 2146
609 0044FE1 2147
610 10 CONTINUE
611 0041061 2148
612 0041EE1 2149
613 0042021 2150
614 0042161 2151
615 00422A1 2152
616 00423E1 2153
617 0042561 2154
618 00426E1 2155
619 0042801 2156
620 0042F21 2157
621 0043341 2158
622 00437C1 2159
623 0043941 2160
624 00439C1 2161
625 0043EE1 2162
626 00440A1 2163
627 00442E1 2164
628 00446E1 2165
629 0044AE1 2166
630 0044FE1 2167
631 10 CONTINUE
632 0041061 2168
633 0041EE1 2169
634 0042021 2170
635 0042161 2171
636 00422A1 2172
637 00423E1 2173
638 0042561 2174
639 00426E1 2175
640 0042801 2176
641 0042F21 2177
642 0043341 2178
643 00437C1 2179
644 0043941 2180
645 00439C1 2181
646 0043EE1 2182
647 00440A1 2183
648 00442E1 2184
649 00446E1 2185
650 0044AE1 2186
651 0044FE1 2187
652 00440A1 2188
653 00442E1 2189
654 00446E1 2190
655 0044AE1 2191
656 0044FE1 2192
657 00440A1 2193
658 00442E1 2194
659 00446E1 2195
660 0044AE1 2196
661 0044FE1 2197
662 00440A1 2198
663 00442E1 2199
664 00446E1 2200
665 0044AE1 2201
666 0044FE1 2202
667 00440A1 2203
668 00442E1 2204
669 00446E1 2205
670 0044AE1 2206
671 0044FE1 2207
672 00440A1 2208
673 00442E1 2209
674 00446E1 2210
675 0044AE1 2211
676 0044FE1 2212
677 00440A1 2213
678 00442E1 2214
679 00446E1 2215
680 0044AE1 2216
681 0044FE1 2217
682 10 CONTINUE
683 0041061 2218
684 0041EE1 2219
685 0042021 2220
686 0042161 2221
687 00422A1 2222
688 00423E1 2223
689 0042561 2224
690 00426E1 2225
691 0042801 2226
692 0042F21 2227
693 0043341 2228
694 00437C1 2229
695 0043941 2230
696 00439C1 2231
697 0043EE1 2232
698 00440A1 2233
699 00442E1 2234
700 0
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58 004512I          I=IS+LT-1I
59 00452AI          204  F(I,JS,KS)=(C(I)-A(I)*F(I+1,JS,KS))/D(I)
60 0045CAI          DO 205 I=IS,LT
61 0045E2I          PPBLK=F(I,JS,KS)
62 004614I          DO 205 J=JS+1,MT
63 004630I          DO 205 K=KS+1,NT
64 00464CI          F(I,J,K)=PPBLK
65 004675I          205  CONTINUE
66 0046C6I          RETURN
67 0046CEI          END

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NO ERRORS: F70 R05-01.0C SUBROUTINE LINERX 02/21/86 09:59:43 TABLE SPACE: 3 KB
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 199 WORDS
 SINGLE PRECISION FLATING PT SUPPORT REQUIRED FOR EXECUTION

ORIGINAL PAGE IS
OF POOR QUALITY

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1 0000001
2 0000041
3 0000041
4 1/VAR/UC(21,18,10),V(21,18,10),P(21,18,10),OK(21,18,10),
5 2,DE(21,18,10),ERR/ERR/ERR/ERR,ERR/ERR/ERR/ERR,
6 3,PF(21,18,10),W(21,18,10),TM(21,18,10)
7 1/PRC/P/ VISE(21,18,10),CEN(21,18,10),VISC/DEWIN/FLWIN
8 1/PRCCR/ CU(21,18,10),DV(21,18,10),CW(21,18,10)
9 1/TUR/ SIGK,SIGE,CMU1,C2,CMU1,CMU2,E,CCK,HINUM,SMNUM,ANV1(800),
10 2,YN(SOC),YN1(600),STIX(800),SINY(800),SINZ(800),ANW1(800),
11 3,YPLN(800),TAUN(800),ISC(800),JBC(800),KBC(800),ITY(800),
12 4,TALW(800),GEN(21,18,10),MC(21,18,10),IJLO(21,18,10),IITO
13 1/CCEF/ AP(21,18,10),SU(21,18,10),SP(21,18,10),SUK(21,18,10),
14 2,SPK(21,18,10),AE(21,18,10),AW(21,18,10),AN(21,18,10),
15 3,ASC(21,18,10),AT(21,18,10),AB(21,18,10),AP(21,18,10)
16 COMM/CN
17 1/TRAN/ X(21,18,10),Y(21,18,10),Z(21,18,10),TJO(21,18,10),
18 2,CX(21,18,10),CY(21,18,10),CZ(21,18,10),
19 3,EX(21,18,10),EY(21,18,10),EZ(21,18,10),
20 3,SX(21,18,10),SY(21,18,10),SZ(21,18,10)
21 1/LIMIT/ L/M/LT,MT/L1/L2,M1/M2/LD,MO/ISWU,ISWW/ISWP/ISWK/ISWE,
22 2,ALU,ALV,ALP,ALKALE,ALVIS,ALW,N/N1/N2,NO/ISWW,IG,NT,ALC,DTT
23 C----EVALUATE WALL SECUNDARY CONDITIONS USING WALL FUNCTIONS
24 DO 150 III=1,IITO
25 I=IAC(III)
26 J=JBC(III)
27 K=KBC(III)
28 GO TO (1,2,3,4,5,6), ITTY(III)
29 1 CONTINUE
30 C----NORTH
31 00009EI
32 000032I
33 0000C6I
34 0000F2I
35 00011EI
36 00014AI
37 1 CALL WALLFN(IE,YP,YP1,CDK,COE,CDEN,SINX(III),SINY(III),
38 2 SINZ(III),F(I,J+1,K),AN(I,J,K),SU(I,J,K),SP(I,J,K),
39 3 TAUN(III),YPLN(III),GEN(I,J,K),VISE(I,J,K),
40 4 U(I,J+1,K),V(I,J+1,K),W(I,J+1,K),U(I,J,K),V(I,J,K),
41 5 W(I,J,K),U(I,J-1,K),V(I,J-1,K),W(I,J-1,K),
42 5 -SPK(I,J,K),SPPK(I,J,K),ANV1(III),ANW1(III),TJO(I,J,K))
43 4 GO TO 150
44 2 CONTINUE
45 C----SOUTH
46 0005BAI
47 0005CEI
48 00C60EI
49 00063AI
50 000666I
51 1 CALL WALLFN(IE,YP,YP1,CCX,COE,CDEN,SINX(III),SINY(III),
52 2 SINZ(III),F(I,J-1,K),AN(I,J,K),SU(I,J,K),SP(I,J,K),
53 3 TAUN(III),YPLN(III),GEN(I,J,K),VISE(I,J,K),
54 4 U(I,J-1,K),V(I,J-1,K),W(I,J-1,K),U(I,J,K),V(I,J,K),
55 5 -SPK(I,J,K),SPPK(I,J,K),ANV1(III),ANW1(III),TJO(I,J,K))
56 4 GO TO 150
57 3 CONTINUE

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58 000406I 59 YP=YNC(III) 60 YP1=YN1(III) 61 CCK=CK(I,J,K) 62 CDE=DE(I,J,K) 63 CEN=DEN(I,J,K) 64 CALL WALLFN(IE,YP,YP1,CCK,CDE,CEN,SINX(III),SINY(III),SINX(III),SINY(III), 65 1 SINZ(III),F(I,J,K),AN(I,J,K),SU(I,J,K),SP(I,J,K), 66 2 TALN(I,J,K),YPLN(III),GEN(I,J,K),VISE(I,J,K), 67 3 UCI+1,J,K),YPLN(III+1,J,K),AN(I,J,K),U(I,J,K),V(I,J,K), 68 4 W(I,J,K),U(I,J,K),V(I,J,K),U(I,J,K),V(I,J,K), 69 5 -SPK(I,J,K),SPK(I,J,K),ANV1(III),ANV1(III),TJC(I,J,K)) 70 GO TO 150 71 0004F2I 72 CONTINUE 73 C-----WEST 74 YP=YNC(III) 75 YP1=YN1(III) 76 CCK=CK(I,J,K) 77 CDE=DE(I,J,K) 78 001072I 79 CALL WALLFN(IE,YP,YP1,CCK,CDE,CEN,SINX(III),SINY(III), 80 1 SINZ(III),F(I,J,K),AN(I,J,K),SU(I,J,K),SP(I,J,K), 81 2 TALN(III),YPLN(III),GEN(I,J,K),VISE(I,J,K), 82 3 UCI+1,J,K),VCI-1,J,K),W(I-1,J,K),U(I,J,K),V(I,J,K), 83 4 W(I,J,K),U(I,J,K),V(I,J,K),W(I+1,J,K),W(I,J,K), 84 5 -SPK(I,J,K),SPK(I,J,K),ANV1(III),ANV1(III),TJC(I,J,K)) 85 GO TO 150 86 CONTINUE 87 C-----TOP 88 YP=YNC(III) 89 YP1=YN1(III) 90 COK=OK(I,J,K) 91 COE=DE(I,J,K) 92 CEN=DEN(I,J,K) 93 CALL WALLFN(IE,YP,YP1,CCK,CDE,CEN,SINX(III),SINY(III), 94 1 SINZ(III),F(I,J,K+1),AN(I,J,K),SU(I,J,K),SP(I,J,K), 95 2 TALN(III),YPLN(III),GEN(I,J,K),VISE(I,J,K), 96 3 UCI+1,J,K+1),VCI+1,J,K+1),W(I,J,K+1),U(I,J,K),V(I,J,K), 97 4 W(I,J,K+1),U(I,J,K+1),V(I,J,K+1),W(I,J,K+1), 98 5 -SPK(I,J,K),SPK(I,J,K),ANV1(III),ANV1(III),TJC(I,J,K)) 99 GO TO 150 100 CONTINUE 101 001A1E1 102 001A32I 103 001A45I 104 001A72I 105 001A8E1 106 001AC4I 107 1 CALL WALLFN(IE,YP,YP1,CCK,CDE,CEN,SINX(III),SINY(III), 108 2 SINZ(III),F(I,J,K-1),AN(I,J,K),SU(I,J,K),SP(I,J,K), 109 3 TALN(III),YPLN(III),GEN(I,J,K),VISE(I,J,K), 110 4 UCI+1,J,K-1),VCI+1,J,K-1),W(I,J,K-1),U(I,J,K+1), 111 5 -SPK(I,J,K),SPK(I,J,K),ANV1(III),ANV1(III),TJC(I,J,K)) 112 GO TO 23I 113 001F49I 114 001F46I

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NO ERRORS: F7D RUS-01.3C SUBROUTINE BOUNC 02/21/36 1C:0C:46 TABLE SPACE: 7 x3
STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 135 WORKS
SINGLE PRECISION ELATING PT SUPPORT REQUIRED FOR EXECUTION

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1 00000001      SUBROUTINE WALLFNCIE,YP/YP1,CK1,DE1,DE11,SINX1,SINY1,
2      SINZ1,F1AN1,AN1,SP1,
3      TALN1,YPLN1,GEN1,VISE1,
4      U1,V1,W1,UC,VQ,
5      W2,U2,V2,W2,
6      SUK1,SPK1,ANW2,ANW1,T01)
7      COMMON
8      1/PRCP/ VISE(21,18,10) CEN(21,18,10),VIS(21,18,10),DENIN,FLWIN,
9      1/TUR/ SIGK,SIGE,CMU,C1,C2,CMU1,CMU2,E,CK,MIN,SMNU,ANV1(800),
10     2,YN(300),YA1(300),SINX(600),SINY(600),SINZ(600),ANW1(800),
11     3,YPLN(600),TAUN(600),TSC(800),UBC(800),KSC(800),ITY(300),
12     4,TALN(800),GEN(21,18,10),MC(21,18,10),TJ0(21,18,10),IT0
13      ----- ALL FUNCTIONS USING LOGARITHMIC WALL LAW
14      GC TC (1,2,3,4,5,6,7), IE
15      1  CONTINUE
16      C-----U
17      000003EI      SQRK=SQRT(CK1)
18      00005AI      YPLN1=CEN1*SQRTK*CN1*YP/VIISC
19      00007EI      IF(YPLN1.LE. 11.5) GC TO 111
20      000096I      TMULT=DE1*CN1*SCRTK*CK/ALOG(1*YPLN1) -
21      000050I      111  TMULT=VIISC/YP
22      000066I      112  TAUN=-TMULT
23      0000CF8I      PTA=TJC1*SINX1*TMULT/YP1
24      00001CAI      SP1=SP1-PTA
25      000123I      SU1=SU1+PTA+F1
26      00013AI      P1=SQRT(1.-SINX1**2)
27      000152I      ANV2=AN1*SCRT(1.-SINY1**2)
28      00016AI      ANW2=AN1*SCRT(1.-SINZ1**2)
29      0001C8I      AN1=P1*AN1
30      000208I      RETURN
31      00021AI      2  CONTINUE
32      000220I      C-----V
33      000220I      TMULT=-TALN1
34      000232I      PTA=TJC1*SINY1*TMULT/YP1
35      000232I      SP1=SP1-PTA
36      000250I      SU1=SU1+PTA*=1
37      000262I      AN1=ANV2
38      00027AI      RETURN
39      000286I      3  CONTINUE
40      00028CI      C-----W
41      00028CI      TMULT=-TAUN1
42      00029EI      PTA=TJC1*SINZ1*TMULT/YP1
43      00029EI      SP1=SP1-PTA
44      00029CI      SU1=SU1+PTA*=F1
45      0002CEI      AN1=ANW2
46      0002E6I      RETURN
47      0002F2I      4  CONTINUE
48      0002F8I      C-----TM
49      0002F8I      RETURN
50      0002F8I      5  CONTINUE
51      0002FEI      C-----K
52      0002FEI      SQRK=SQRT(CK1)
53      0002FEI      IF(YPLN1.LE. 11.63) GC TO 511
54      00031AI      DITM=DE1*CMU2*SQRTK*ALOG(1*YPLN1)/(CK*YP)
55      000332I      GC TO 512
56      000332I      DITM=SEN1*CMU2*SQRTK*YPLN1/YP
57      000388I      511

```

```

58 0003ACI 512 CONTINUE
59 0003ACI
60 00038E1 DDU=U0-U1
61 0003001 DDU=V0-V1
62 0003E2I P1=DDU**2+DDV**2+DCW**2
63 00042AI DDU=U2-U1
64 00043CI DDU=V2-V1
65 00044E1 DDW=W2-W1
66 000460I P2=DDU**2+DDV**2+DCW**2
67 0004A8I YP2=2*YF1-YP
68 0004COI GENEGEN1=VISE1*P2/Y02/YP2
69 0004E9I IF(GENR=LE, C-C) GENR=0.0
70 000504I SU1=TJC1*(SU1*DC1+(P1*TAN1**2/VISE1+GENR))
71 00053E1 SP1=TJC1*(SPR1-CITM)
72 00056E1 AN1=C-C
73 000562I RETURN
74 000568I 5 CONTINUE
75 C-----E
76 000568I TERM=CMU2/(CK+YP)
77 000582I SU1=H1LM*TERM*DC1**1.5
78 000582I SP1=-H1NM
79 0005C4I RETURN
80 0005CAI 7 CONTINUE
81 C-----F
82 0005CAI RETURN
33 0005D0I END
NO ERRORS: F7D 005-01.0C SUBROUTINE WALLFN 02/21/86 1C:01:12 TABLE SPACE: 5 KB
STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 122 WORDS
SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

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1 00000001
2 00000041
3 1/ VAR/U(21,18,10),V(21,18,10),P(21,18,10),DK(21,18,10),
4 2/ DE(21,18,10),ERRU,ERRV,ERRM,ERRK,ERRW,
5 3/ PF(21,18,10),W(21,18,10),TM(21,18,10),
6 1/ PRCP/ VISE(21,18,10),DEN(21,18,10),VISC,DENIN,FLOWIN
7 1/ PCCR/ DU(21,18,10),DV(21,18,10),DW(21,18,10)
8 1/ CCEF/ AP(21,18,10),SU(21,18,10),SP(21,18,10),SUK(21,18,10),
9 2/ SPK(21,18,10),AE(21,18,10),AN(21,18,10),
10 3/ AS(21,18,10),AT(21,18,10),AP(21,18,10),AP0(21,18,10),
11 4/ COMMON
12 1/ TRAN/ X(21,18,10),Y(21,18,10),Z(21,18,10),TJ0(21,18,10),
13 2/ CX(21,18,10),CY(21,18,10),CZ(21,18,10),
14 3/ EX(21,18,10),EY(21,18,10),EZ(21,18,10),
15 3/ SX(21,16,10),SY(21,18,10),SZ(21,18,10),
16 1/ UNSTOY/U0(21,18,10),VC(21,18,10),WG(21,18,10),DK0(21,18,10),
17 2/ CEO(21,18,10),DEN0(21,18,10),DW0(21,18,10)
18 1/ LIMIT/ LM,LIT,LT,L1,M1,M2,LO,MO,ISU,ISWV,ISWP,ISWK,ISWE,
19 2/ ALU,ALV,ALPA,LALV,IS,ALW,N1,N2,NO,ISWH,IG,NT,ALC,CTT
20 C-----SYMMETRIC, CYCLIC AND EXIT CONDITIONS AND LINK MODIFICATIONS
21 GO TO (1,2,3), IC
22 1
23 C-----CONTINUE
24 1
25 C-----BOTTOM
26 K=1
27 DO 10 I=1,L
28 0000041
29 0000041
30 0000041
31 0000041
32 0000041
33 0000041
34 0000041
35 0000041
36 0000041
37 0000041
38 0000041
39 0000041
40 0000041
41 0000041
42 0000041
43 0000041
44 0000041
45 0000041
46 0000041
47 0000041
48 0000041
49 0000041
50 0000041
51 0000041
52 0000041
53 0000041
54 0000041
55 0000041
56 0000041
57 0000041
COMMON
1/ TRAN/ X(21,18,10),Y(21,18,10),Z(21,18,10),TJ0(21,18,10),
2/ CX(21,18,10),CY(21,18,10),CZ(21,18,10),
3/ EX(21,18,10),EY(21,18,10),EZ(21,18,10),
3/ SX(21,16,10),SY(21,18,10),SZ(21,18,10),
1/ UNSTOY/U0(21,18,10),VC(21,18,10),WG(21,18,10),DK0(21,18,10),
2/ CEO(21,18,10),DEN0(21,18,10),DW0(21,18,10)
1/ LIMIT/ LM,LIT,LT,L1,M1,M2,LO,MO,ISU,ISWV,ISWP,ISWK,ISWE,
2/ ALU,ALV,ALPA,LALV,IS,ALW,N1,N2,NO,ISWH,IG,NT,ALC,CTT
C-----SYMMETRIC, CYCLIC AND EXIT CONDITIONS AND LINK MODIFICATIONS
GO TO (1,2,3), IC
1
C-----CONTINUE
1
C-----BOTTOM
K=1
DO 10 I=1,L
DO 10 J=2,LT
U(I,J,K)=U(I,J,K+1)
V(I,J,K)=V(I,J,K+1)
W(I,J,K)=W(I,J,K+1)
TM(I,J,K)=TM(I,J,K+1)
DK(I,J,K)=DK(I,J,K+1)
DE(I,J,K)=DE(I,J,K+1)
10 CONTINUE
C-----SALT OUT (BASED ON INFLCW MASS FLOW RATE)
10
1/ INIT
FLOW=0.0
ARCEN=C*0
DO 50 J=2,JT
DO 50 K=2,KT
UC=(V(I,J,K)+V(I,J-1,K)+V(I,J,K-1)+V(I,J-1,K-1))*0.25
CEN=(CEN(I,J,K)+CEN(I,J-1,K)+CEN(I,J,K-1)+CEN(I,J-1,K-1))*0.25
P1=(X(I,J,K)+X(I,J-1,K)-X(I,J-1,K-1))*X(I,J-1,K-1)*0.5
P2=(Y(I,J,K)+Y(I,J-1,K)-Y(I,J-1,K-1))*Y(I,J-1,K-1)*0.5
P3=(Z(I,J,K)+Z(I,J-1,K)-Z(I,J-1,K-1))*Z(I,J-1,K-1)*0.5
Q1=(X(I,J,K)+X(I,J-1,K)-X(I,J-1,K-1))*X(I,J-1,K-1)*0.5
Q2=(Y(I,J,K)+Y(I,J-1,K)-Y(I,J-1,K-1))*Y(I,J-1,K-1)*0.5
Q3=(Z(I,J,K)+Z(I,J-1,K)-Z(I,J-1,K-1))*Z(I,J-1,K-1)*0.5
AREA=SQRT(P1+P2+P3*P3)*SQRT(Q1*Q1+Q2*Q2+Q3*Q3)
FLOW=FLOW+DENC*AREA*UC
ARCEN=ARCEN+DENC*AREA
CONTINUE
50
UINC=(FLOW-FLOWIN)/ARCEN
DO 60 J=2,JT
DO 60 K=2,KT
U(I,J,K)=U(I,J,K)
V(I,J,K)=V(I,J,K)
W(I,J,K)=W(I,J,K)
1833 28
1834 29
1835 30
1836 31
1837 32
1838 33
1839 34
1840 35

```

```

58 000A44I
59 000A94I
60 000AE4I      60  CONTINUE
61 000B14I      RETURN
62 C-----LINK CCEFF. MODIFICATIONS
63 000B14I      2  CONTINUE
64 C-----EAST OUT
65 000B14I      I=IT
66 000B26I      DO 200 J=2, JT
67 000B34I      DO 200 K=2, KT
68 000B4EI      AE(I,J,K)=0.0
69 000B7AI      200  CONTINUE
70 000B8AI      RETURN
71 000B90I      C-----UPDATE UNSTEADY COEFF.
72 000B90I      3  CONTINUE
73 000B30I      IF(CCTT .NE. 0.0) GO TO 301
74 000BC8I      DO 300 I=1, IT
75 000B20I      DO 300 J=1, JT
76 000BF8I      DO 300 K=1, KT
77 000C10I      300  APO(I,J,K)=0.0
78 000C84I      RETRN
79 000C84I      301  CONTINUE
80 000C84I      DO 310 I=1, IT
81 -000CA2I      DO 310 J=1, JT
82 000CB4I      DO 310 K=1, KT
83 000C02I      APO(I,J,K)=DENO(I,J,K)/DTT
84 000D24I      U0(I,J,K)=U1(I,J,K)
85 000D70I      V0(I,J,K)=V1(I,J,K)
86 000D98I      W0(I,J,K)=W1(I,J,K)
87 000E08I      TMD(I,J,K)=TM(I,J,K)
88 000E54I      DKO(I,J,K)=DK(I,J,K)
89 000EA0I      DEC(I,J,K)=DE(I,J,K)
90 000EEC1      310  CONTINUE
91 000F34I      RETURN
92 000F34I      END

```

NO ERRORS: F7D R05-01.0C SUBROUTINE SYMBOL 02/21/86 1C:01:59 TABLE SPACE: 7 KB
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 131 WORDS
 SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

```

1 0000001  SUBROUTINE WALVAL(PW,IS,IT,JS,JT,KS,KT,F)
2 0000041  DIMENSION F(21,19,10)
3 0000041  COMMON
4 1/TURY SIGK,SIGE,CMU,C1,C2,CMU1,CMU2,E,CK,MINUM,SHNUM,ANV1(800),
5 2 YN(800),YN1(300),SINX(800),SINY(800),SINZ(800),AN41(800),
6 3 YOLN(800),TAUN(800),I3C(800),I5C(800),K3C(800),IITY(800),
7 4 TAUN(800),GEN(21,19,10),WC(21,18,10),IJLOC(21,18,10),IJTC
8 C-----ASSIGN WALL VALUES
9 0000041  DO 10 J=JS,JT
10 0000281  DO 10 K=KS,KT
11 0000401  F(IIS-1,J,K)=PW*F(IIS,J,K)
12 0000A21  10 F(IIT-1,J,K)=PW*F(IIT,J,K)
13 0001341  DO 20 I=IS-1,IT+1
14 0001521  DO 20 K=KS,KT
15 00016A1  FCI(J,JS-1,K)=PW*F(I,JS,K)
16 0001CC1  20 FCI(J,IT+1,K)=PW*F(I,IT,K)
17 00025E1  DO 30 I=IS-1,IT+1
18 00027C1  DO 30 J=JS-1,JT+1
19 00029A1  FCI(J,KS-1)=PW*F(I,J,KS)
20 0002FA1  30 FCI(J,KT+1)=PW*F(I,KT)
21 00038A1  DO 40 ITT=1,IITC
22 00039E1  F=I3C(ITT)
23 000321  J=JEC(ITT)
24 0003C61  K=KEC(ITT)
25 00030A1  GO TO (1,2,3,4,5,6), IITY(ITT)
26 0004161  1 F(I,J+1,K)=PW*F(I,J,K)
27 0004781  2 GO TO 4C
28 00047E1  2 FCI(J-1,K)=PW*F(I,J,K)
29 0004501  2 GO TO 4C
30 0004E61  3 FCI(I,J,K)=PW*F(I,J,K)
31 0005481  3 GO TO 4C
32 00054E1  4 F(I-1,J,K)=PW*F(I,J,K)
33 0005301  5 GO TO 4C
34 00059E1  5 F(I,J,K+1)=PW*F(I,J,K)
35 0006161  5 GO TO 4C
36 00061C1  6 F(I,J,K-1)=PW*F(I,J,K)
37 00067C1  40 CONTINUE
38 0006941  RETURN
39 00069A1  END

```

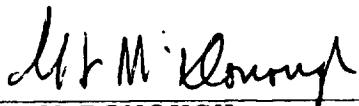
NO ERRORS: F7D ROS-01.0C SUBROUTINE WALVAL 02/21/86 1C:02:24 TABLE SPACE: 3 K9
 STATEMENT BUFFER: 20 LINES/1321 BYTES STACK SPACE: 154 WORDS
 SINGLE PRECISION FLOATING PT SUPPORT REQUIRED FOR EXECUTION

APPROVAL

A COMPUTER CODE FOR THREE-DIMENSIONAL INCOMPRESSIBLE FLOWS
USING NONORTHOGONAL BODY-FITTED COORDINATE SYSTEMS

By Y. S. Chen

The information in this report has been reviewed for technical content. Review of any information concerning Department of Defense or nuclear energy activities or programs has been made by the MSFC Security Classification Officer. This report, in its entirety, has been determined to be unclassified.



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